

Nonlinear Finite Element Methods Material nonlinearities

Vinh Phu Nguyen

Sources of nonlinearities

Geometrical nonlinearities

large displacement/rotation (structural instability)

finite deformation: large strain (metal forming)

Material nonlinearities

plasticity

cohesive zone models

damage

visco-plasticity

$$\sigma = f(\epsilon, \alpha)$$

Boundary nonlinearities

contact problems

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References

Nonlinear finite elements for continua and structures, T. Belytschko, W.K. Liu and B. Moran, Wiley, 2000.

Nonlinear finite element methods, P. Wriggers, Springer, 2008.

Non-linear finite element analysis of solids and structures, R. de Borst, M.A. Crisfield, J.J.C. Remmers, C.V. Verhoosel, Wiley, 2012.

Nonlinear continuum mechanics for finite element analysis, J. Bonet and R.D. Wood, Cambridge, 1997.

Material models

Thermodynamics+experiments

Rate independent

(Time independent)

Elastic

Elasto-plastic

Elasto-damage

Plastic-damage

Rate dependent

(Time dependent)

strain rate

Viscoelastic

Viscoplastic

Visco-elasto-damage

Equilibrium equation

$$\int_{\Omega} \delta \boldsymbol{\epsilon}^T \boldsymbol{\sigma} d\Omega = \int_{\Gamma_t} \delta \mathbf{u}^T \bar{\mathbf{t}} d\Gamma_t$$

Linear problems

$$\int_{\Omega} \delta \boldsymbol{\epsilon}^T \mathbf{D}^e \boldsymbol{\epsilon} d\Omega = \int_{\Gamma_t} \delta \mathbf{a}^T \bar{\mathbf{t}} d\Gamma_t$$

$$\int_{\Omega} \delta \mathbf{a}^T \mathbf{B}^T \mathbf{D}^e \mathbf{B} \mathbf{a} d\Omega = \int_{\Gamma_t} \delta \mathbf{a}^T \mathbf{N}^T \bar{\mathbf{t}} d\Gamma_t$$

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Equilibrium equation

Discrete equilibrium equation

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \mathbf{f}_{\text{ext}}$$

with

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma} d\Omega$$

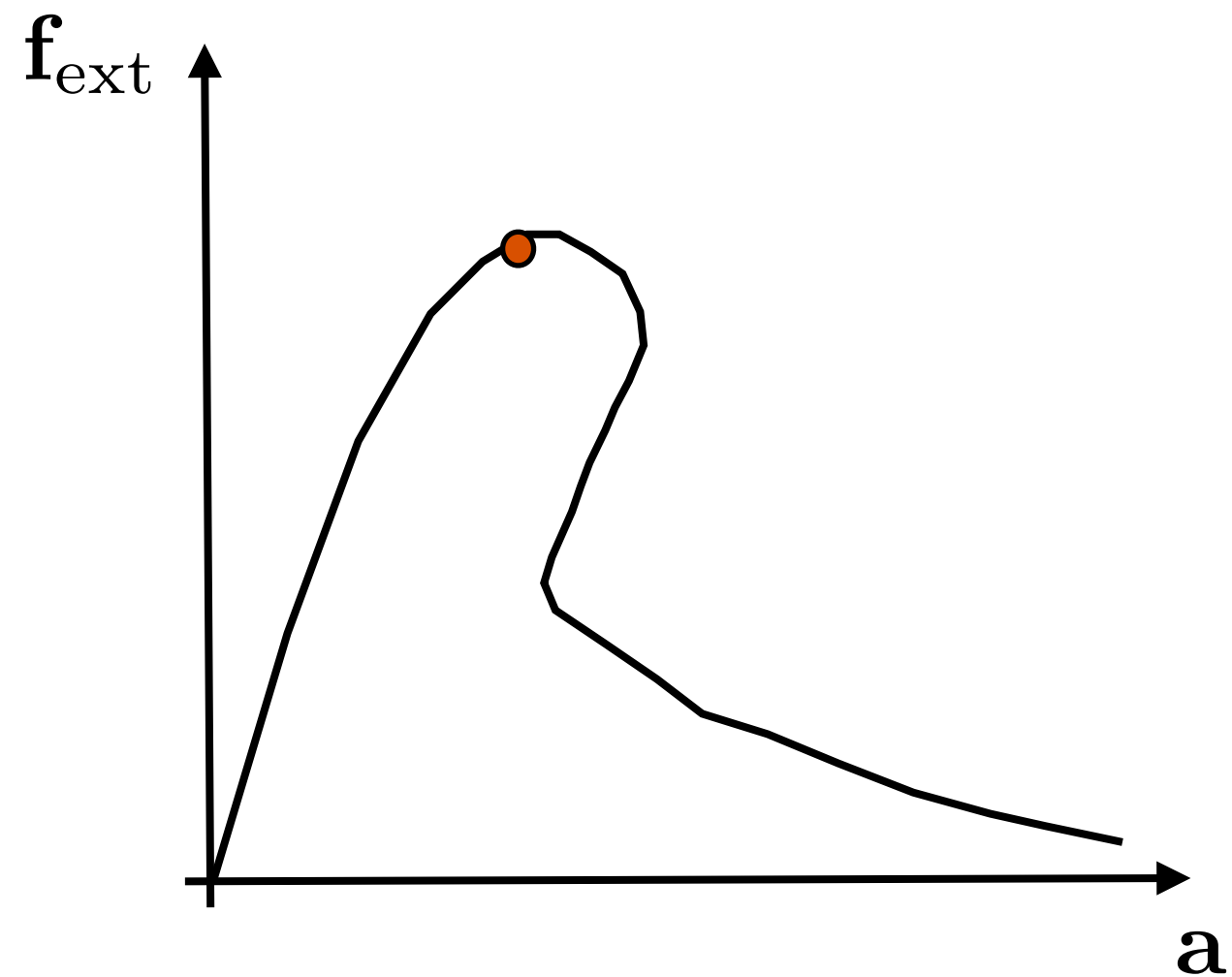
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$$\boldsymbol{\epsilon} = \mathbf{B}\mathbf{a}$$

$$\boldsymbol{\sigma} = f(\boldsymbol{\epsilon}, \alpha)$$

history variables

equilibrium path/curve



Argyris (1964)

Equilibrium equation

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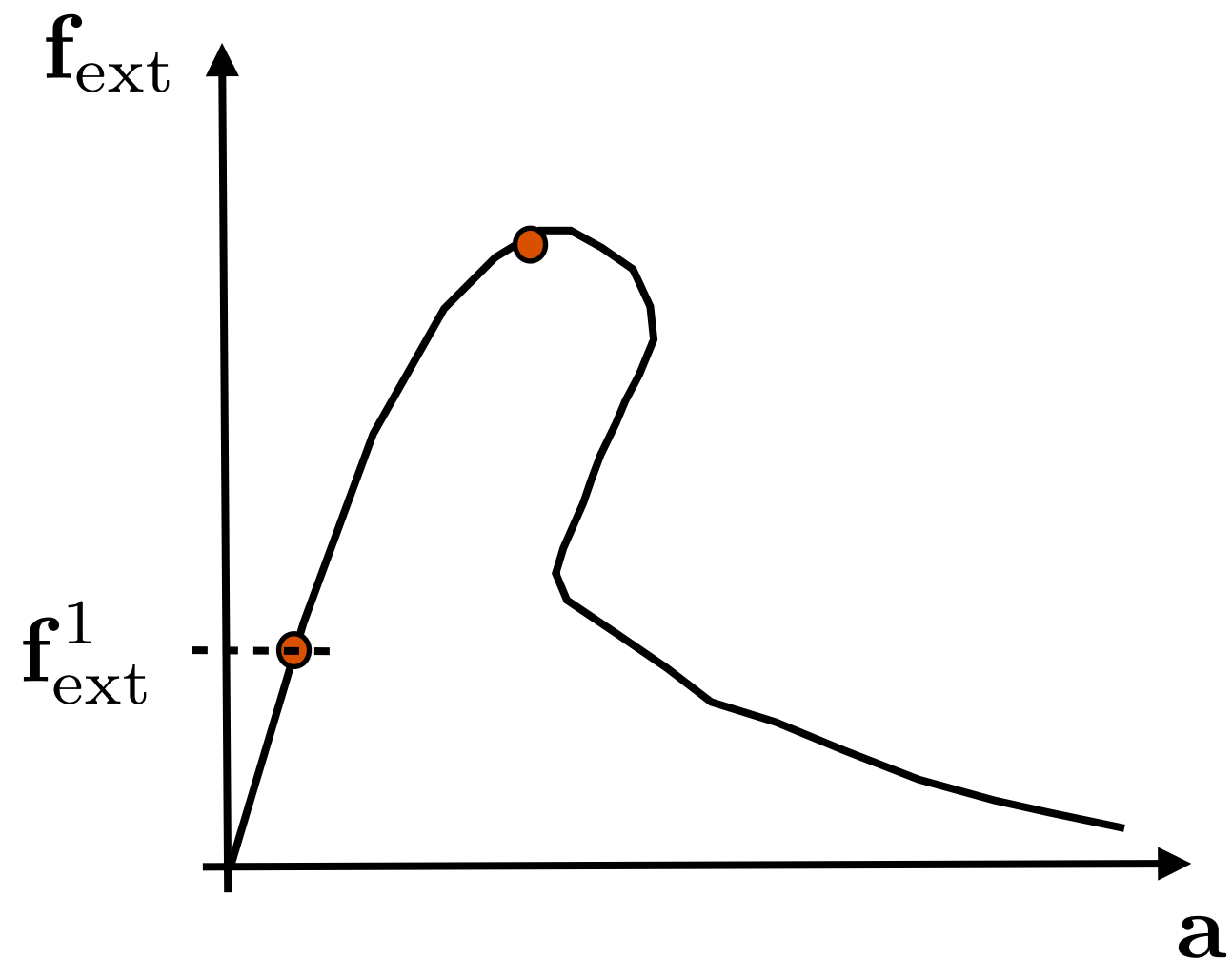
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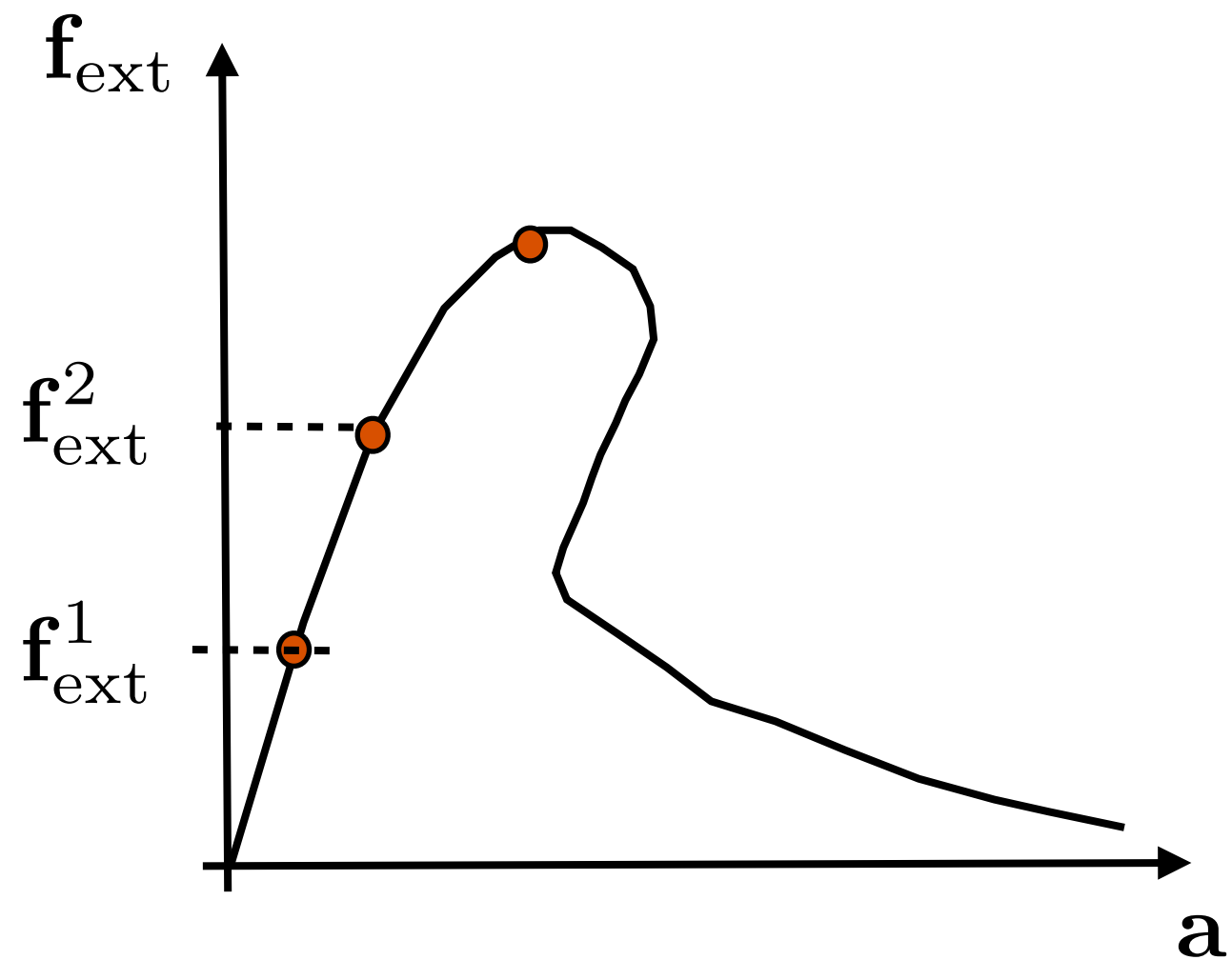
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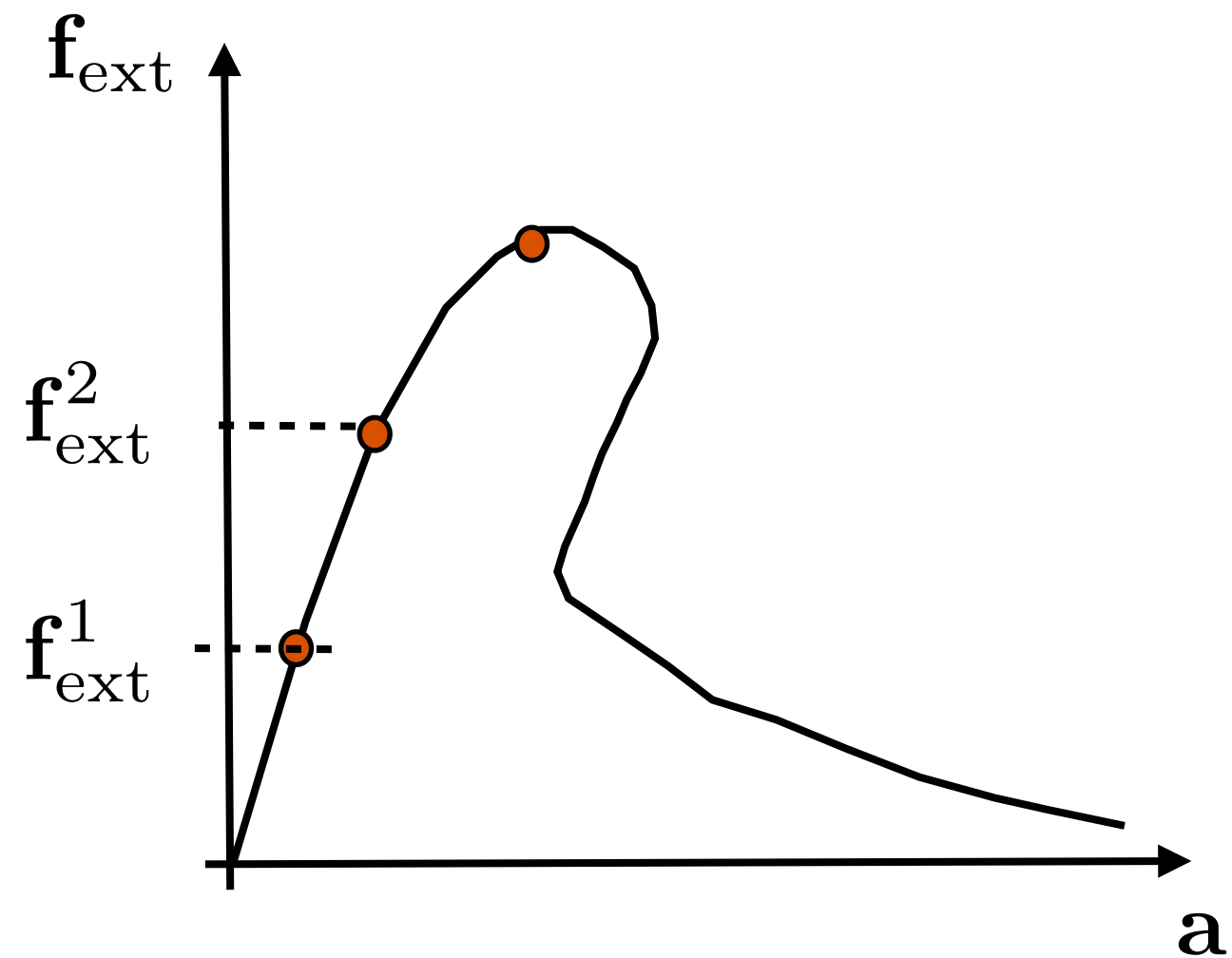
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Incremental

Argyris (1964)

Equilibrium equation

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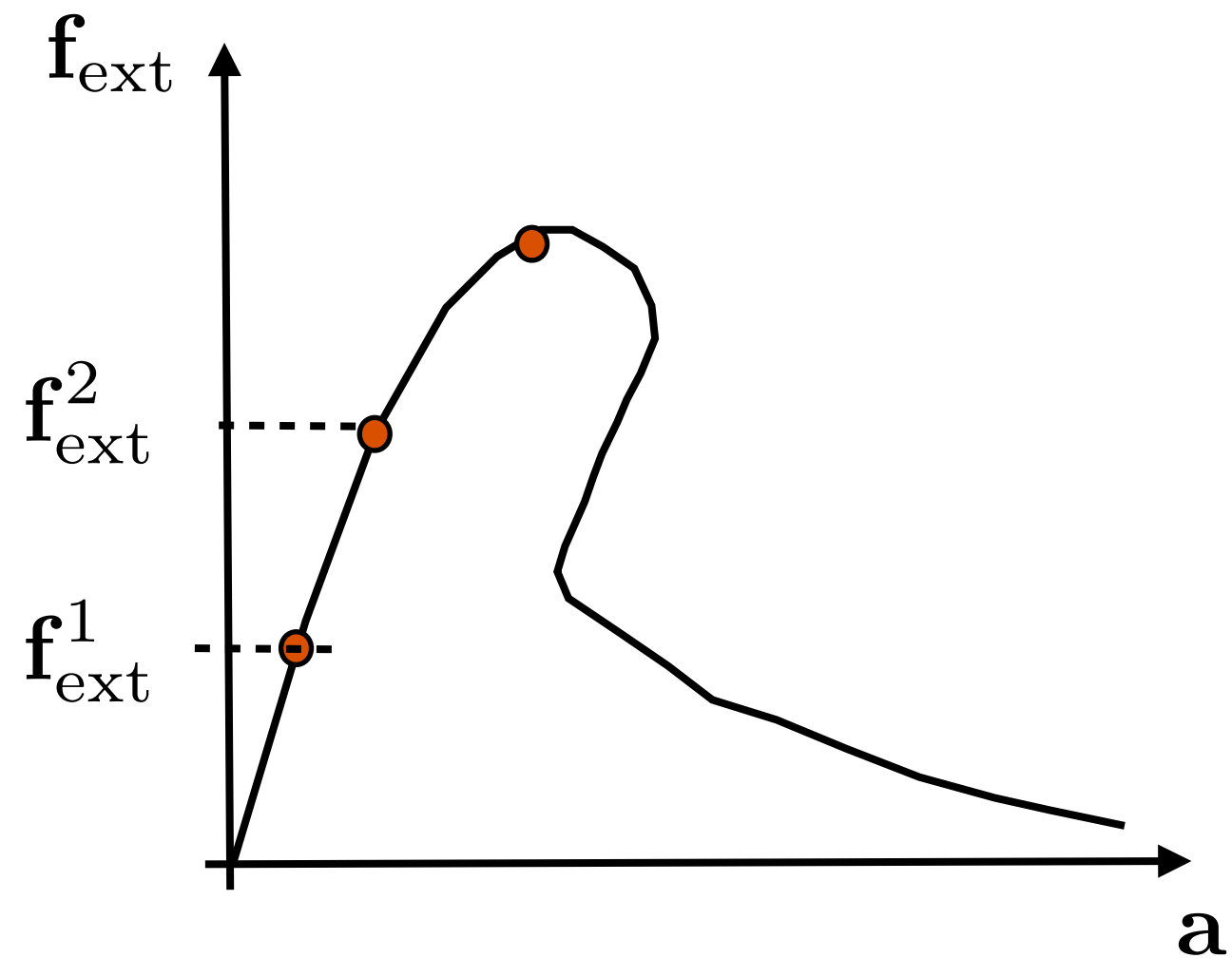
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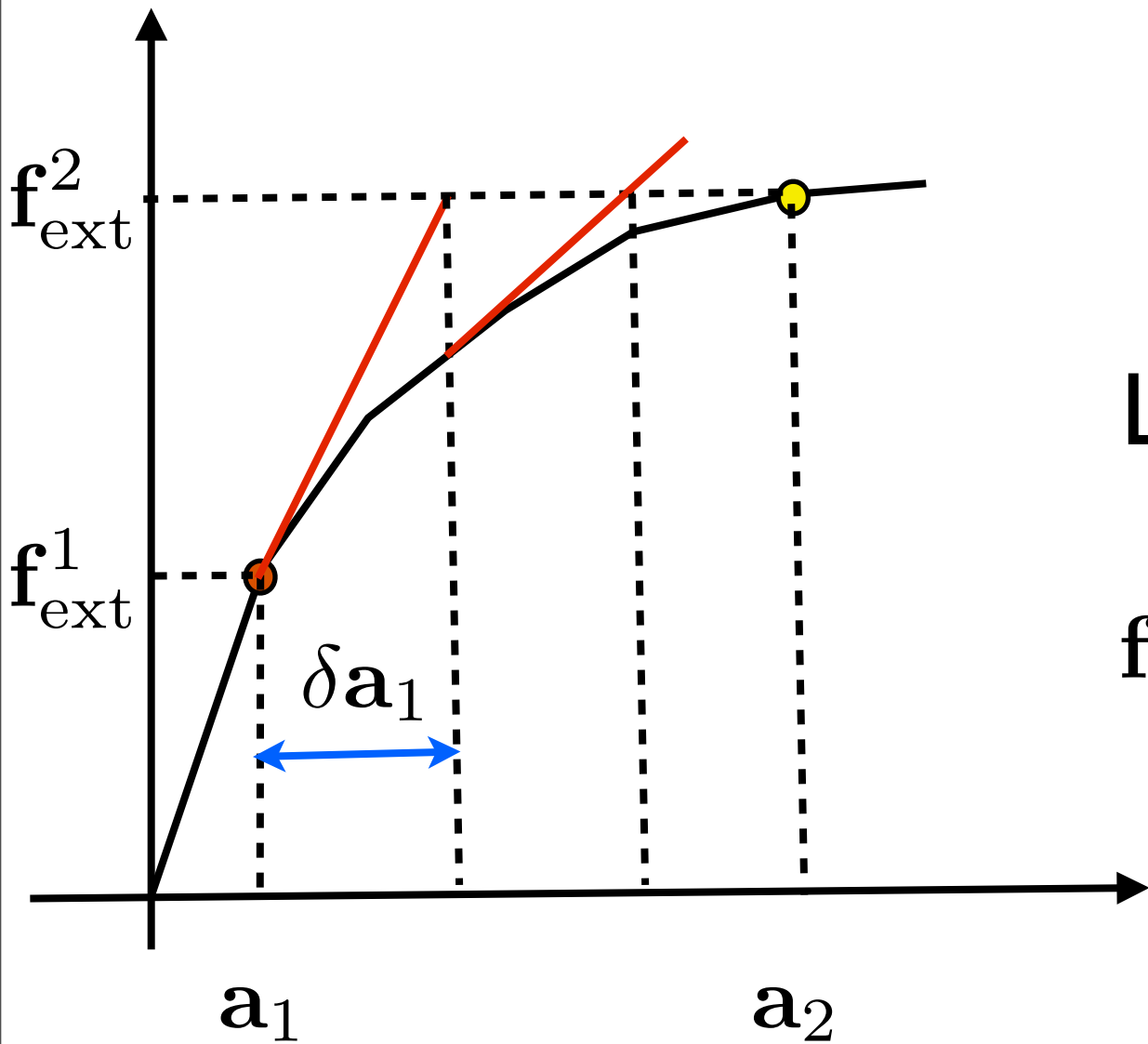


Incremental iterative

Argyris (1964)

Newton-Raphson methods

Full NR scheme



$$\mathbf{f}_{\text{int}}(\mathbf{a}_2) - \mathbf{f}_{\text{ext}}^2 = \mathbf{0}$$

Linearization around \mathbf{a}_1

$$\mathbf{f}_{\text{int}}^n(\mathbf{a}_1) - \mathbf{f}_{\text{ext}}^2 + \frac{\partial \mathbf{f}_{\text{int}}^n}{\partial \mathbf{a}} \Big|_{\mathbf{a}_1} \delta \mathbf{a} = \mathbf{0}$$

$$\mathbf{K}(\mathbf{a}_1) \delta \mathbf{a}_1 = \mathbf{f}_{\text{ext}}^n - \mathbf{f}_{\text{int}}^n(\mathbf{a}_1)$$

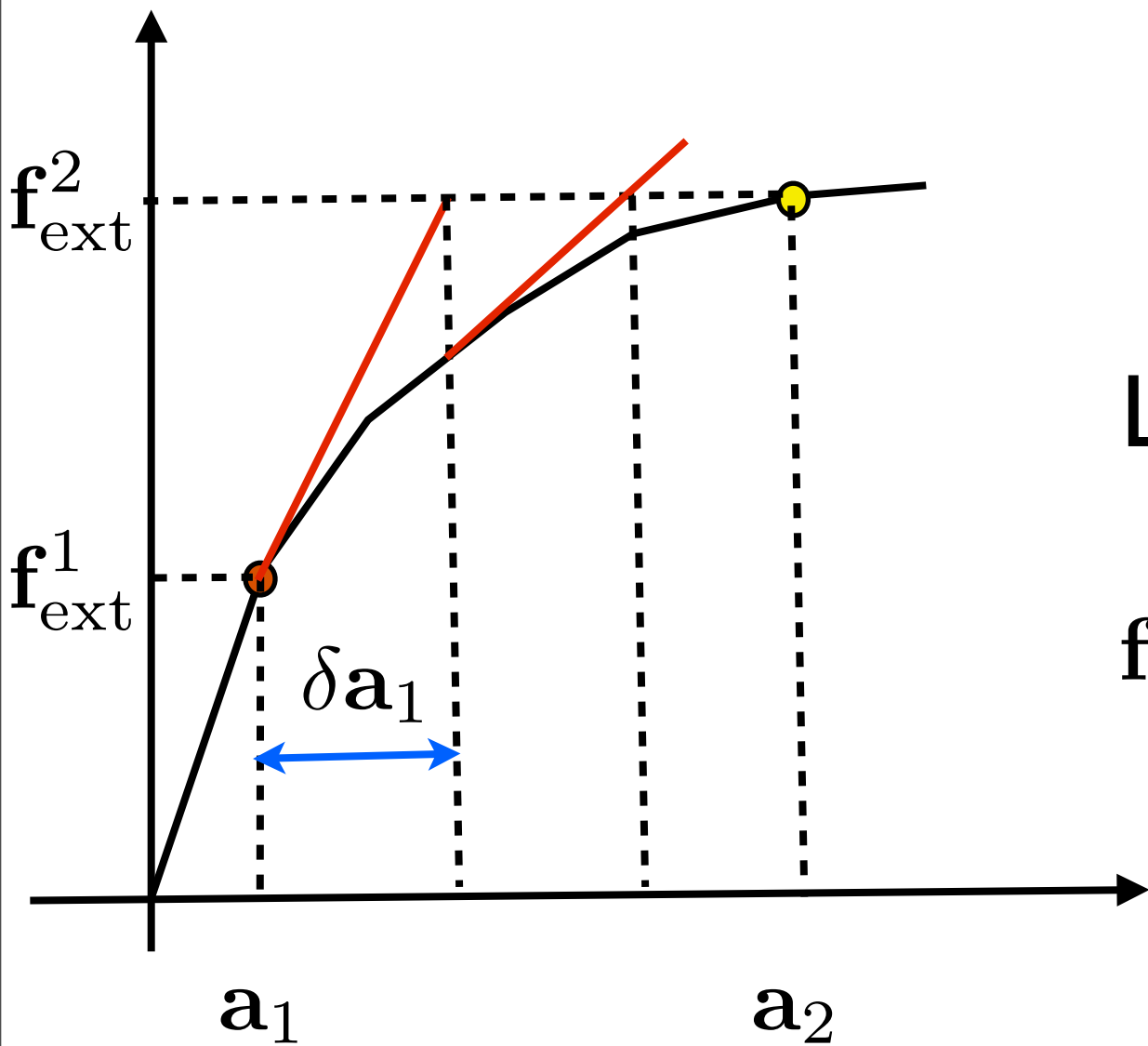
$$\mathbf{a}_1 \leftarrow \delta \mathbf{a}_1$$

Tangent stiffness matrix

$$\mathbf{K} = \frac{\partial \mathbf{f}_{\text{int}}}{\partial \mathbf{a}}$$

Newton-Raphson methods

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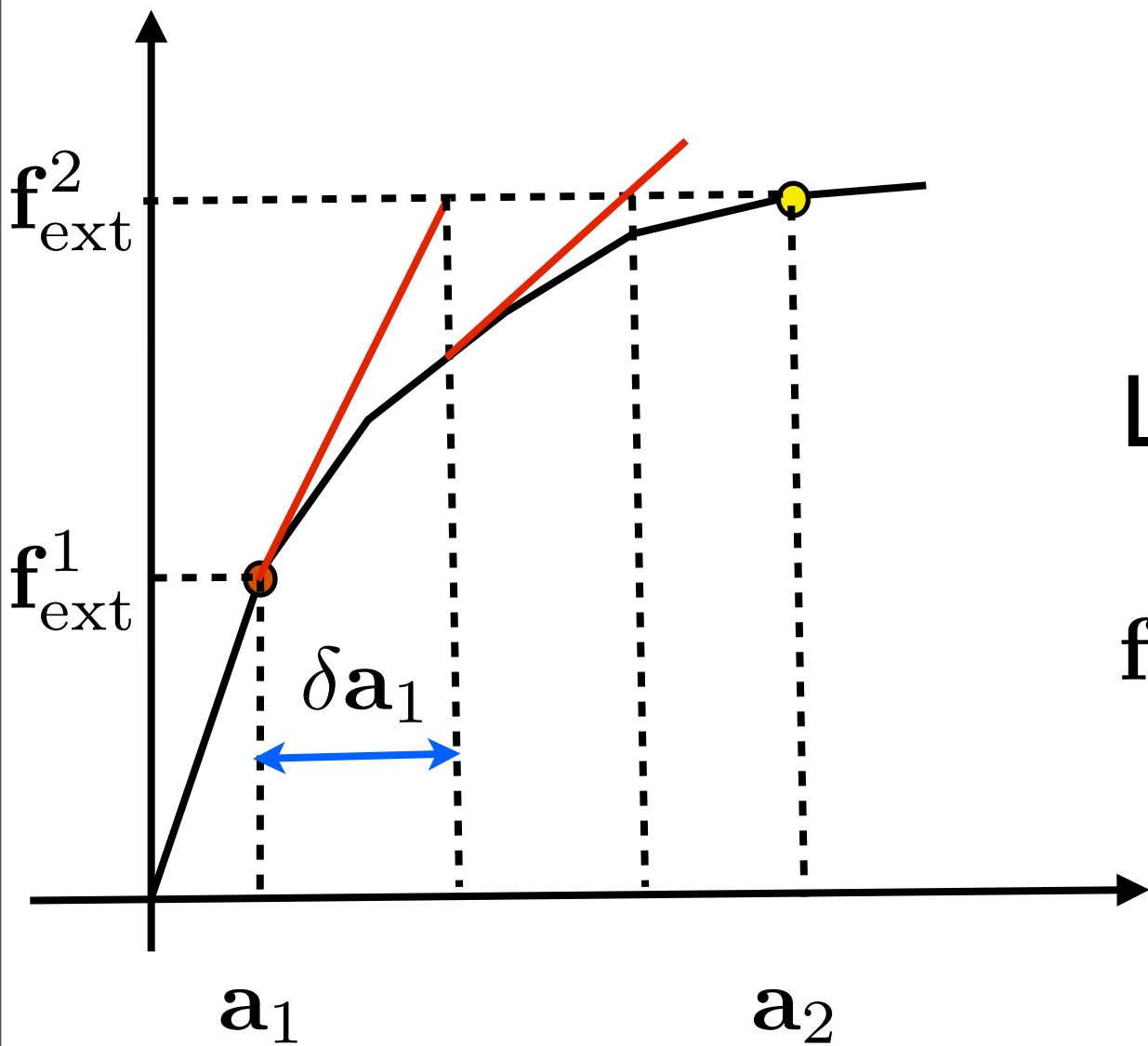
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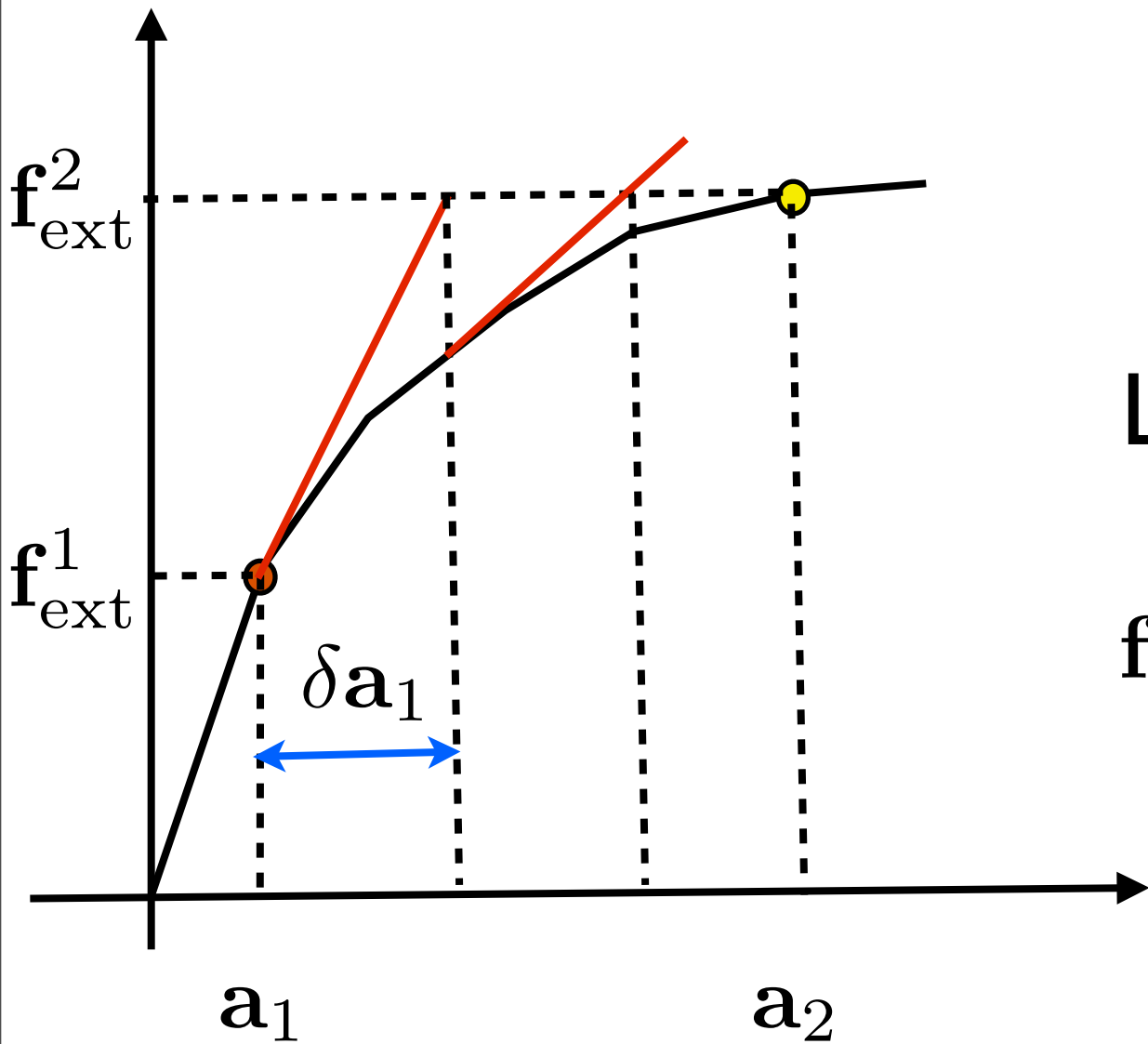
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Tangent stiffness matrix

$$\mathbf{K} = \frac{\partial \mathbf{f}_{\text{int}}}{\partial \mathbf{a}}$$

Newton-Raphson methods

Full NR scheme



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Tangent stiffness matrix

$$\mathbf{K} = \frac{\partial \mathbf{f}_{\text{int}}}{\partial \mathbf{a}}$$

residual/out of balance vector

Tangent stiffness matrix

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma} d\Omega$$

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma}[\boldsymbol{\epsilon}(\mathbf{a})] d\Omega$$

$$\mathbf{K} = \int_{\Omega} \mathbf{B}^T \frac{\partial \boldsymbol{\sigma}[\boldsymbol{\epsilon}(\mathbf{a})]}{\partial \mathbf{a}} d\Omega$$

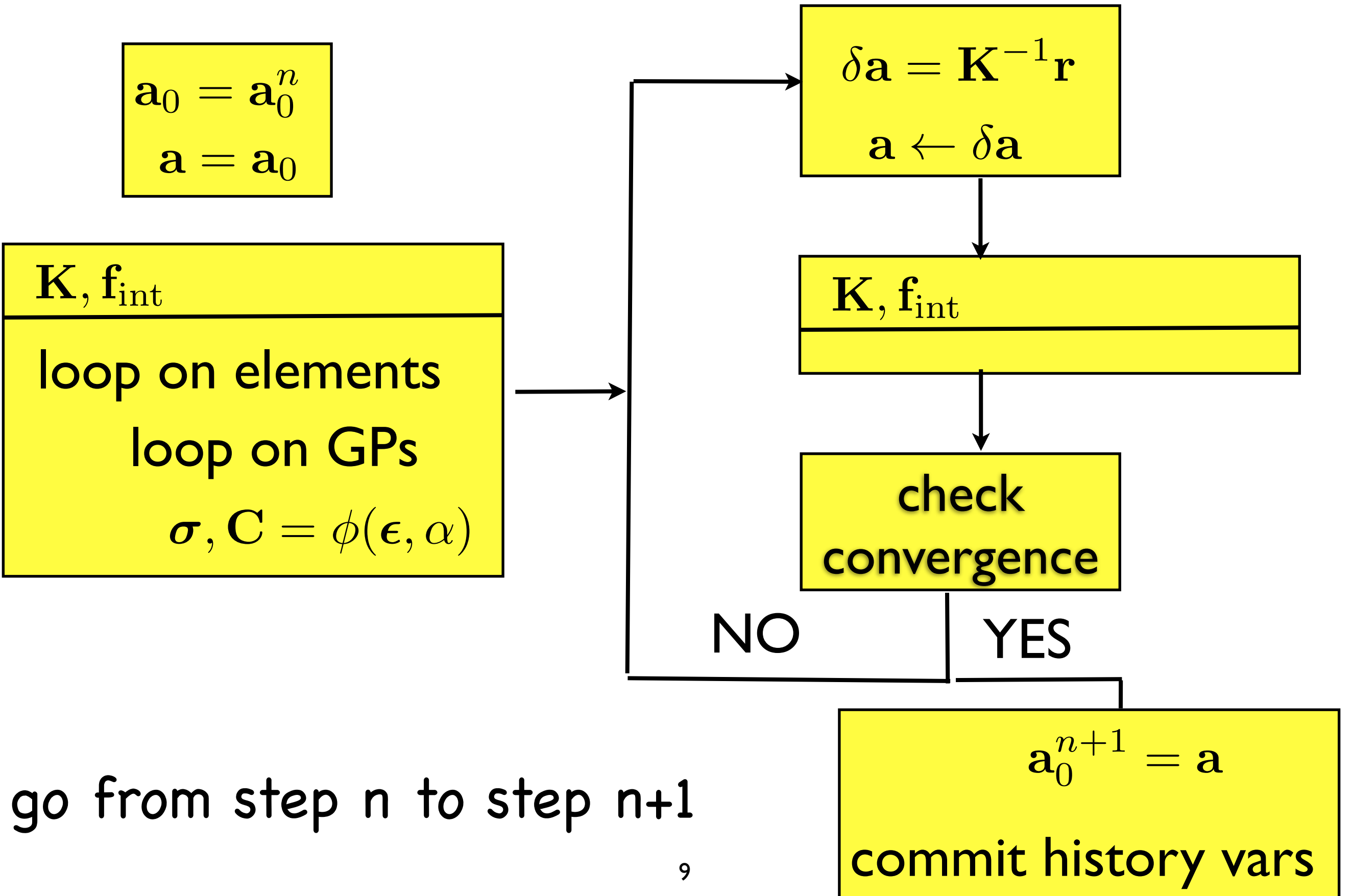
$$\mathbf{K} = \frac{\partial \mathbf{f}_{\text{int}}^n}{\partial \mathbf{a}}$$

$$\mathbf{d}\boldsymbol{\sigma} = \mathbf{C} d\boldsymbol{\epsilon}$$

$$\mathbf{K} = \int_{\Omega} \mathbf{B}^T \frac{\partial \boldsymbol{\sigma}}{\partial \boldsymbol{\epsilon}} \mathbf{B} d\Omega$$

$$\mathbf{K} = \int_{\Omega} \mathbf{B}^T \mathbf{C} \mathbf{B} d\Omega$$

Flowchart



Flowchart

$$\mathbf{a}_0 = \mathbf{a}_0^n$$
$$\mathbf{a} = \mathbf{a}_0$$

$\mathbf{K}, \mathbf{f}_{\text{int}}$

loop on elements
loop on GPs
 $\boldsymbol{\sigma}, \mathbf{C} = \phi(\boldsymbol{\epsilon}, \alpha)$

(XFEM: cracks)

$$\delta \mathbf{a} = \mathbf{K}^{-1} \mathbf{r}$$
$$\mathbf{a} \leftarrow \delta \mathbf{a}$$

$$\mathbf{K}, \mathbf{f}_{\text{int}}$$

check
convergence

NO

YES

go from step n to step n+1

$$\mathbf{a}_0^{n+1} = \mathbf{a}$$

commit history vars

At integration level

For one Gauss point, do

1. compute displacement \mathbf{a}
2. compute strains $\boldsymbol{\epsilon}$, get history α
3. compute stresses/material tangent
4. compute internal force vector
5. compute tangent stiffness

$$\mathbf{a} = \mathbf{N}\mathbf{a}_e$$

$$\boldsymbol{\epsilon} = \mathbf{B}\mathbf{a}$$

$$\boldsymbol{\sigma} = f(\boldsymbol{\epsilon}, \alpha) \quad \mathbf{C}$$

$$\mathbf{f}_{\text{int}} = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma} d\Omega$$

$$\mathbf{K} \leftarrow \int_{\Omega} \mathbf{B}^T \mathbf{C} \mathbf{B} d\Omega$$

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At integration level

For one Gauss point, do

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stress update

$$\mathbf{a} = \mathbf{N}\mathbf{a}_e$$

$$\boldsymbol{\epsilon} = \mathbf{B}\mathbf{a}$$

$$\boldsymbol{\sigma} = f(\boldsymbol{\epsilon}, \alpha) \quad \mathbf{C}$$

$$\mathbf{f}_{\text{int}} = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma} d\Omega$$

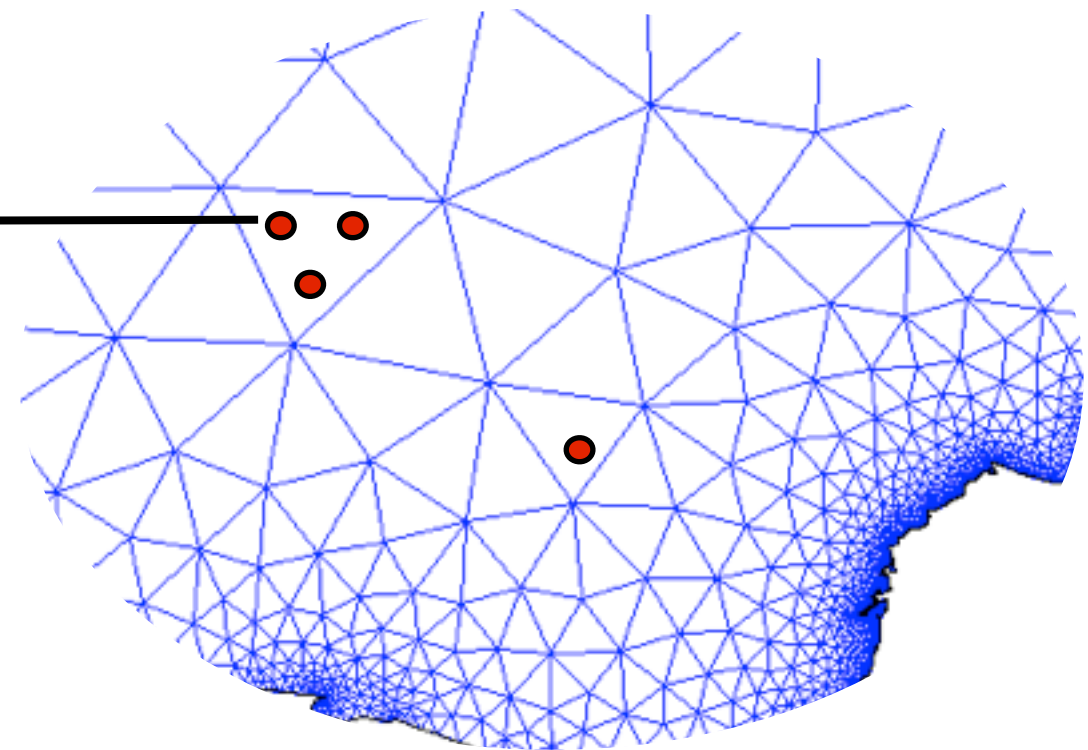
$$\mathbf{K} \leftarrow \int_{\Omega} \mathbf{B}^T \mathbf{C} \mathbf{B} d\Omega$$

Storage of history vars

History variables

FE mesh

$$\alpha = \begin{bmatrix} \alpha_1 \\ \alpha_2 \\ \alpha_3 \\ \vdots \\ \alpha_{n-1} \\ \alpha_n \end{bmatrix}$$



α_0 : history vars at previous converged load step

n: number of integration points of the mesh

Isotropic damage model

$$\sigma = (1 - \omega) \mathbf{D} \epsilon$$

$$\kappa = \max \epsilon_{\text{eq}}$$

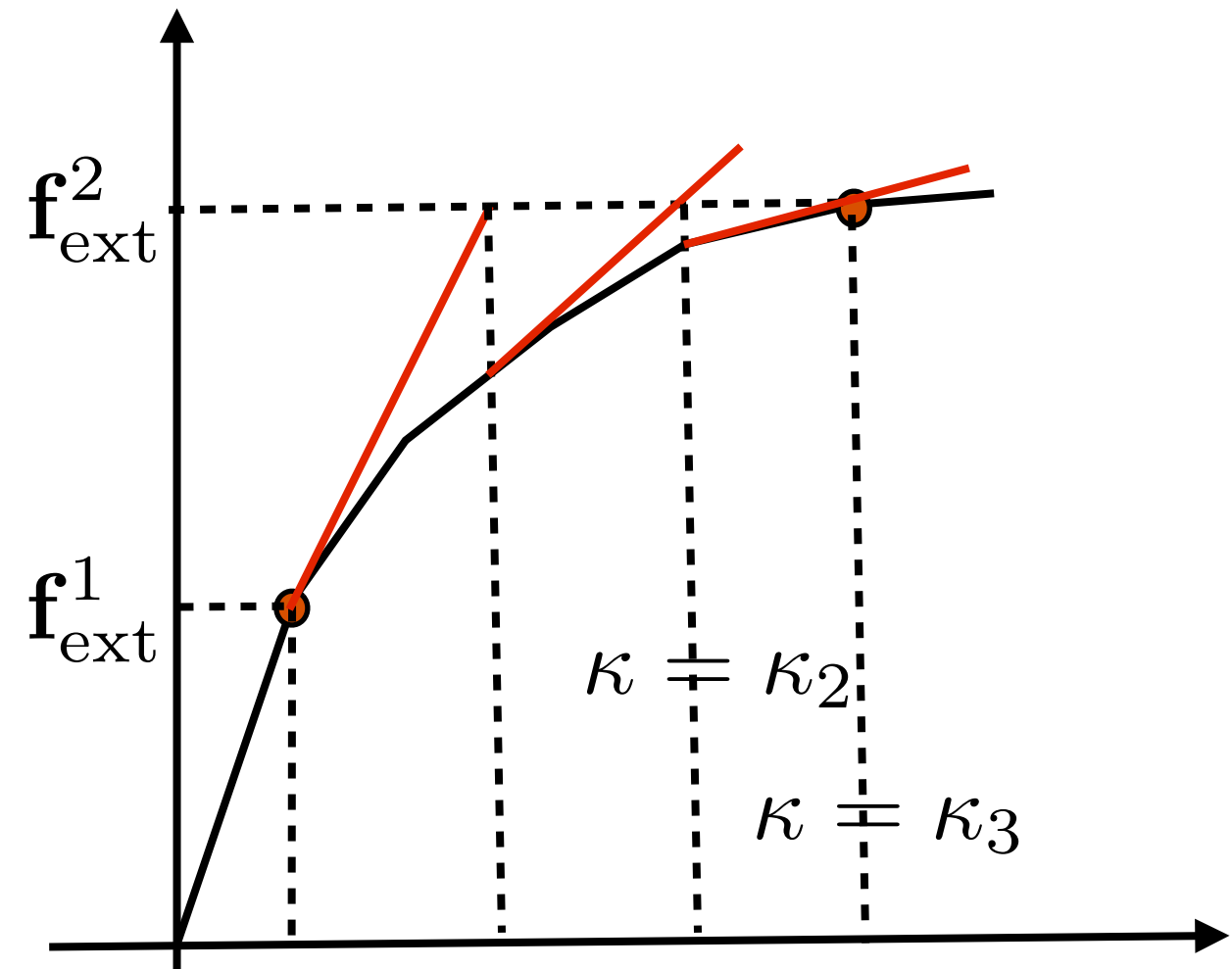
$$\omega = f(\kappa)$$

$$\epsilon_{\text{eq}} = g(\epsilon)$$

$$\dot{\sigma} = (1 - \omega) \mathbf{D} \dot{\epsilon} - \mathbf{D} \epsilon \dot{\omega}$$

$$\dot{\sigma} = (1 - \omega) \mathbf{D} \dot{\epsilon} - \mathbf{D} \epsilon \frac{\partial f}{\partial \epsilon_{\text{eq}}} \frac{\partial \epsilon_{\text{eq}}}{\partial \epsilon} \dot{\epsilon}$$

$$\dot{\sigma} = \left[(1 - \omega) \mathbf{D} - \mathbf{D} \epsilon \frac{\partial f}{\partial \epsilon_{\text{eq}}} \frac{\partial \epsilon_{\text{eq}}}{\partial \epsilon} \right] \dot{\epsilon}$$



κ
 κ_0

$\kappa = \kappa_1$

$\kappa_0 = \kappa$

commit history variables

C

consistent material tangent

$$\omega = 1 - \frac{\kappa}{\kappa_I} [1 - \alpha + \alpha \exp^{-\beta(\kappa - \kappa_I)}], \quad \kappa \geq \kappa_I$$

$$\epsilon_{\text{eq}} = \sqrt{\langle \epsilon_1 \rangle^2 + \langle \epsilon_2 \rangle^2 + \langle \epsilon_3 \rangle^2} \quad \langle x \rangle = 0.5(x + |x|)$$

$$\frac{\partial \epsilon_{\text{eq}}}{\partial \epsilon_{ij}} = \frac{1}{\epsilon_{\text{eq}}} \left(\langle \epsilon_1 \rangle \frac{\partial \epsilon_1}{\partial \epsilon_{ij}} + \langle \epsilon_2 \rangle \frac{\partial \epsilon_2}{\partial \epsilon_{ij}} + \langle \epsilon_3 \rangle \frac{\partial \epsilon_3}{\partial \epsilon_{ij}} \right)$$

$$\langle \epsilon_k \rangle \frac{\partial \langle \epsilon_k \rangle}{\partial \epsilon_{ij}} = \begin{cases} \epsilon_k \frac{\partial \epsilon_k}{\partial \epsilon_{ij}} & \text{if } \epsilon_k > 0 \\ 0 & \text{if } \epsilon_k \leq 0 \end{cases}$$

$$I_1 = \epsilon_{xx} + \epsilon_{yy} + \epsilon_{zz}$$

$$\epsilon^3 - I_1 \epsilon^2 + I_2 \epsilon - I_3 = 0$$

$$I_2 = \epsilon_{xx}\epsilon_{yy} + \epsilon_{yy}\epsilon_{zz} + \epsilon_{zz}\epsilon_{xx} - \epsilon_{xy}^2 - \epsilon_{yz}^2 - \epsilon_{zx}^2$$

$$I_3 = \epsilon_{xx}\epsilon_{yy}\epsilon_{zz} + 2\epsilon_{xy}\epsilon_{yz}\epsilon_{zx} - \epsilon_{xx}\epsilon_{yz}^2 - \epsilon_{yy}\epsilon_{zx}^2 - \epsilon_{zz}\epsilon_{xy}^2$$

$$\frac{\partial \epsilon_k}{\partial \epsilon_{ij}} = \frac{\partial \epsilon_k}{\partial I_1} \frac{\partial I_1}{\partial \epsilon_{ij}} + \frac{\partial \epsilon_k}{\partial I_2} \frac{\partial I_2}{\partial \epsilon_{ij}} + \frac{\partial \epsilon_k}{\partial I_3} \frac{\partial I_3}{\partial \epsilon_{ij}}$$

Isotropic damage: stress update

PyFEM

```
def getStress( self, kinematics ):

    kappa = self.getHistoryParameter('kappa')

    eps , detadstrain = self.getEquivStrain( kinematics.strain )

    if eps > kappa:
        progDam = True
        kappa    = eps
    else:
        progDam = False

    self.setHistoryParameter( 'kappa', kappa )

    omega , domegadkappa = self.getDamage( kappa )

    effStress = dot( self.De , kinematics.strain )

    stress     = ( 1. - omega ) * effStress
    tang       = ( 1. - omega ) * self.De

    if progDam:
        tang += -domegadkappa * outer( effStress , detadstrain )

    return stress , tang
```

One-dimensional elasto-plasticity

```
def getStress( self, deformation ):
```

```
# retrieve history variables
```

```
epsp0 = self.getHistoryParameter('plasticStr')  
alpha0 = self.getHistoryParameter('hardening')
```

two history variables/GP

```
sigmaTrial = self.E * ( deformation.strain - epsp0 )  
yieldFunc = abs ( sigmaTrial ) - ( self.sigmaY + self.K * alpha0 )
```

```
if yieldFunc <= 0.:  
    sigma = sigmaTrial  
    tang = self.E
```

isotropic hardening

```
    epsp = epsp0  
    alpha = alpha0
```

```
else:
```

```
    deltaGamma = yieldFunc / ( self.E + self.K )  
    sigma = ( 1. - deltaGamma * self.E / abs (sigmaTrial) ) * sigmaTrial  
    epsp = epsp0 + deltaGamma * sign ( sigmaTrial )  
    alpha = alpha0 + deltaGamma  
    tang = self.E * self.K / ( self.E + self.K )
```

```
self.setHistoryParameter( 'plasticStr', epsp )  
self.setHistoryParameter( 'hardening', alpha )
```

```
return sigma, tang
```

Convergence rate

It has been mathematically proved, full NR converges quadratically closed to the solution.

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In reality ...

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```
Evaluating the external force...
AttributeError: issim
module `extraModules.solver.nonLin' : residual scale factor = 3.9758e+01
module `extraModules.solver.nonLin' : iter = 1, scaled residual = 8.9753e-02
module `extraModules.solver.nonLin' : iter = 2, scaled residual = 1.9729e-02
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The Newton-Raphson solver converged in 4 iterations
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```

quadratic

Convergence criteria

Residual

$$\frac{\|\mathbf{r}\|}{\|\mathbf{f}_{\text{ext}}\|} \leq \epsilon$$

Energy

$$\frac{\mathbf{f}_{\text{int}}^T \delta \mathbf{a}}{\mathbf{f}_{\text{int},1}^T \delta \mathbf{a}_1} \leq \epsilon$$

Displacement

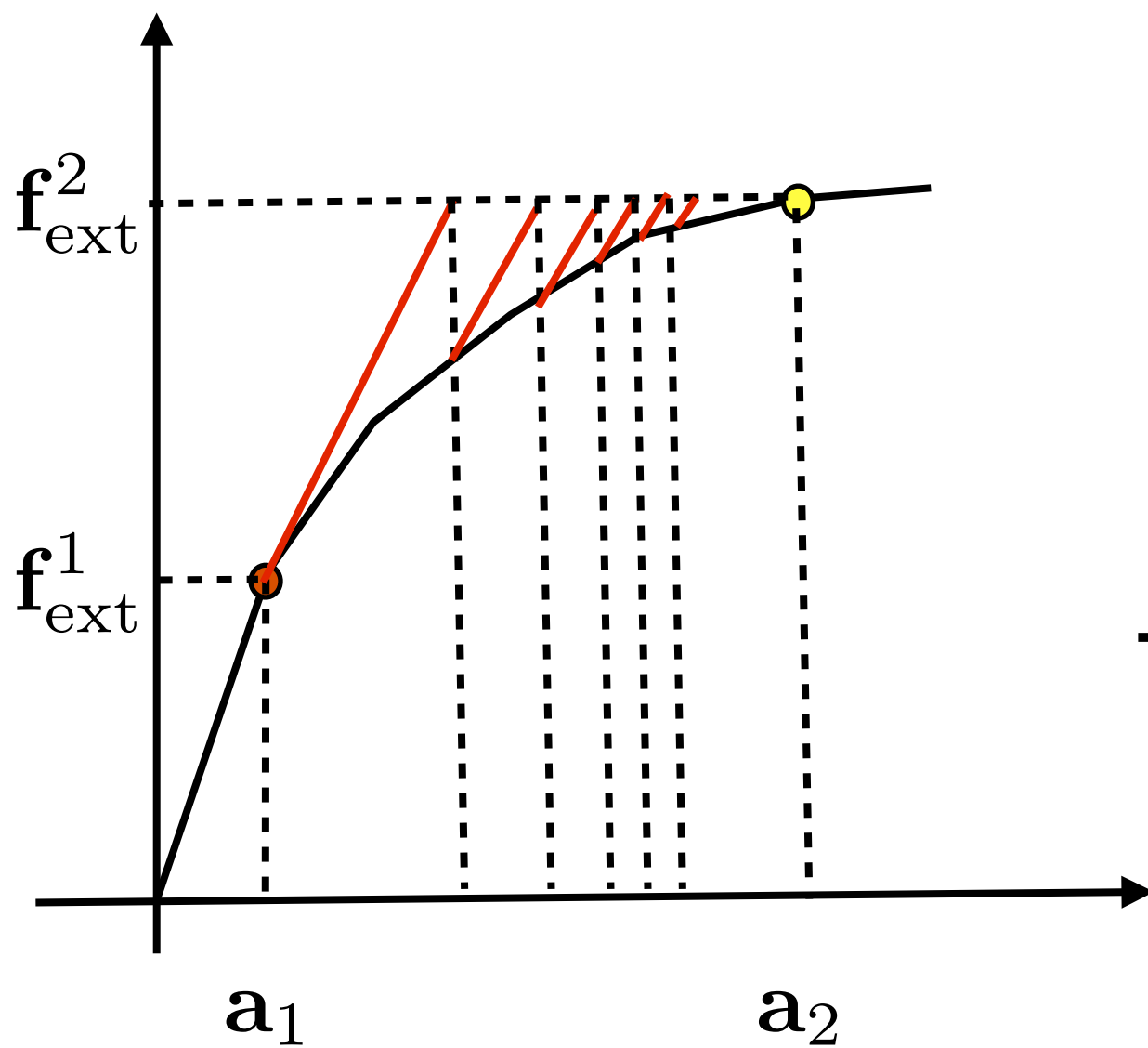
$$\frac{\|\delta \mathbf{a}\|}{\max \|\delta \mathbf{a}_i\|} \leq \epsilon$$

relative norms

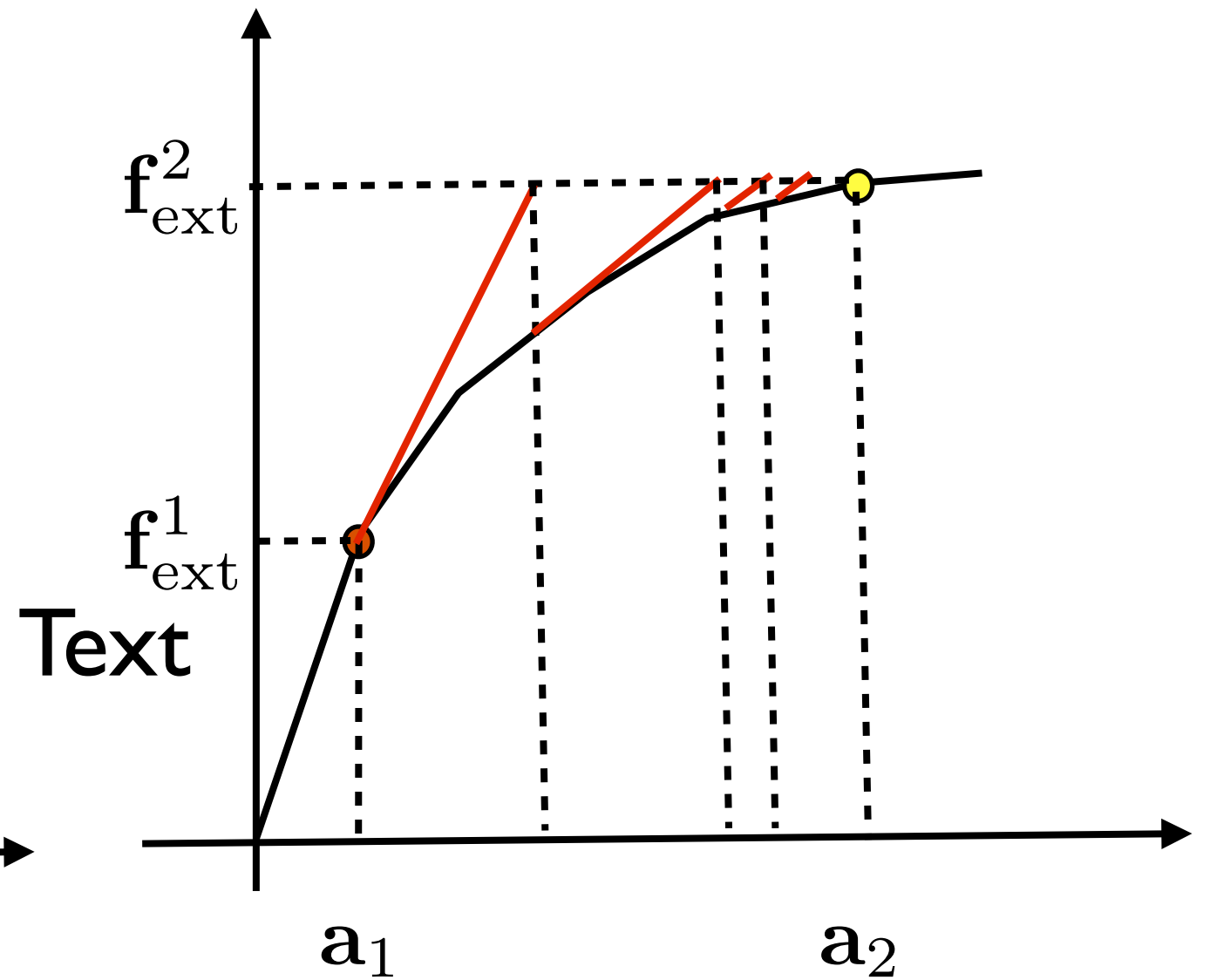
tolerance is problem dependent

$$\|\mathbf{a}\| = \sqrt{\sum_{i=1}^n a_i^2}$$

Modified NR methods



Initial stiffness method



Modified Newton-Raphson

more iterations per step

without consistent material tangent

Direct displacement control

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \mathbf{0}$$

$$a_5 = \lambda \bar{a}, \quad \lambda = 0, 1, 2, \dots$$

For each load step:

First iteration:

$$\mathbf{K} \delta \mathbf{a}_{(1)} = -\mathbf{f}_{\text{int}}$$

$$\delta a_5 = \bar{a}$$

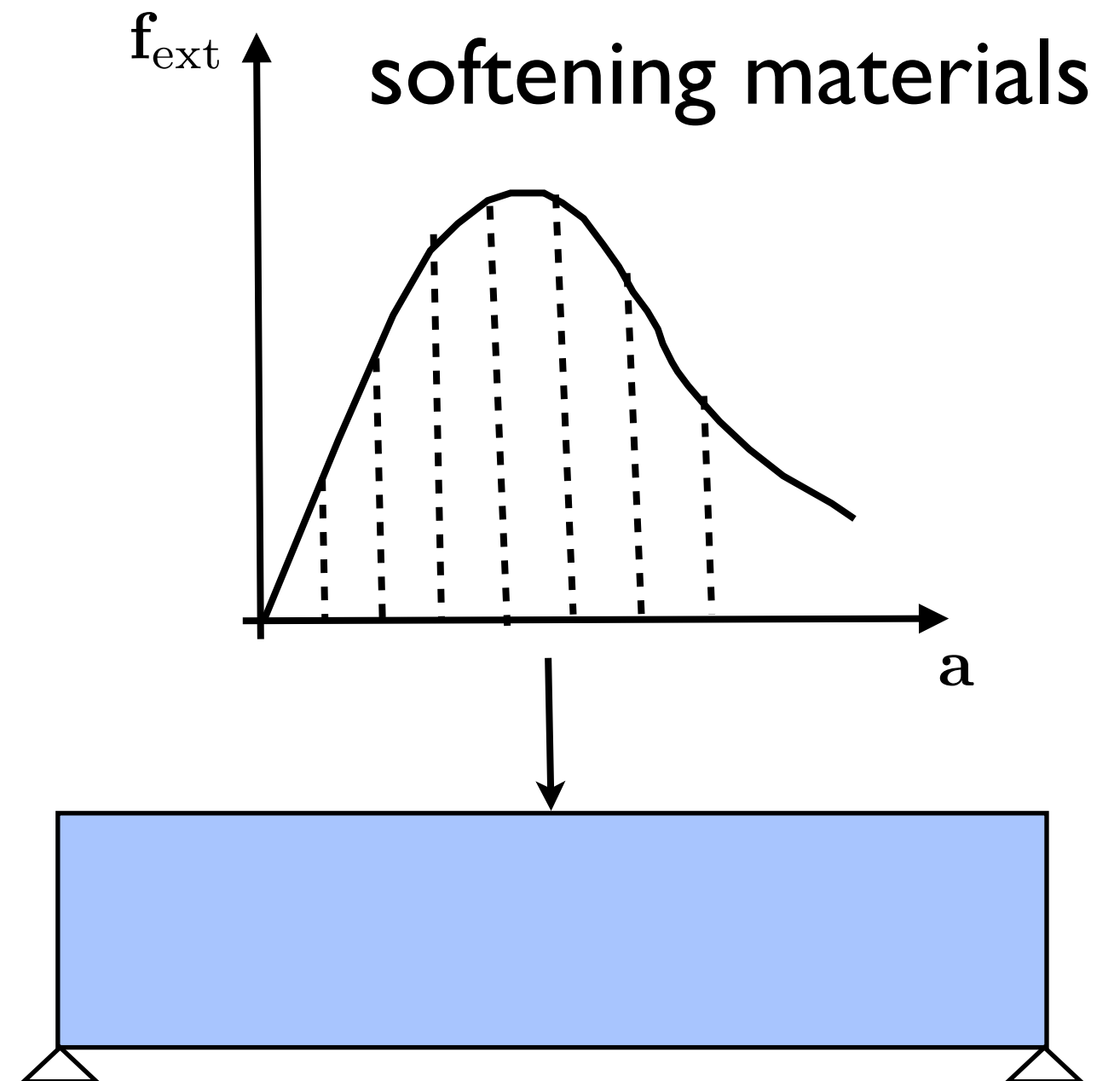
From second iteration:

$$\mathbf{K} \delta \mathbf{a}_{(i)} = -\mathbf{f}_{\text{int}}$$

$$\delta a_5 = 0$$

Argyris (1965)

Batoz, Dhatt (1979)



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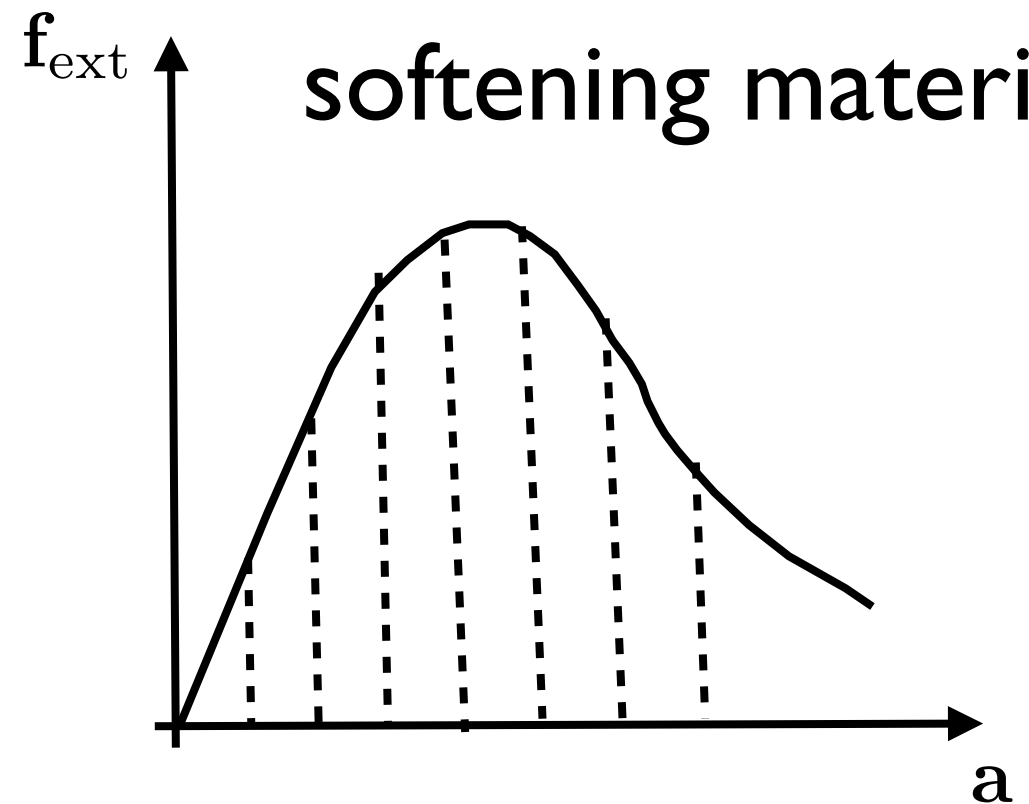
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f_{ext} softening materials



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For each load step:

First iteration:

$$\mathbf{K} \delta \mathbf{a}_{(1)} = -\mathbf{f}_{\text{int}}$$

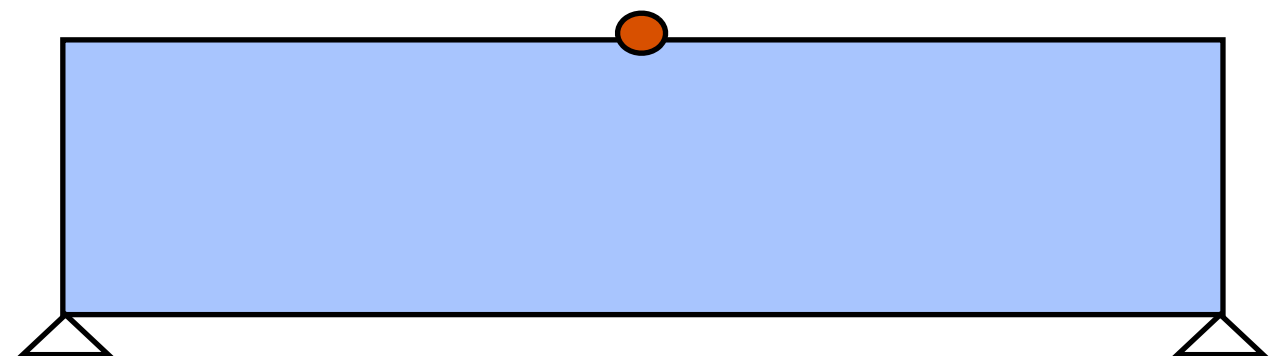
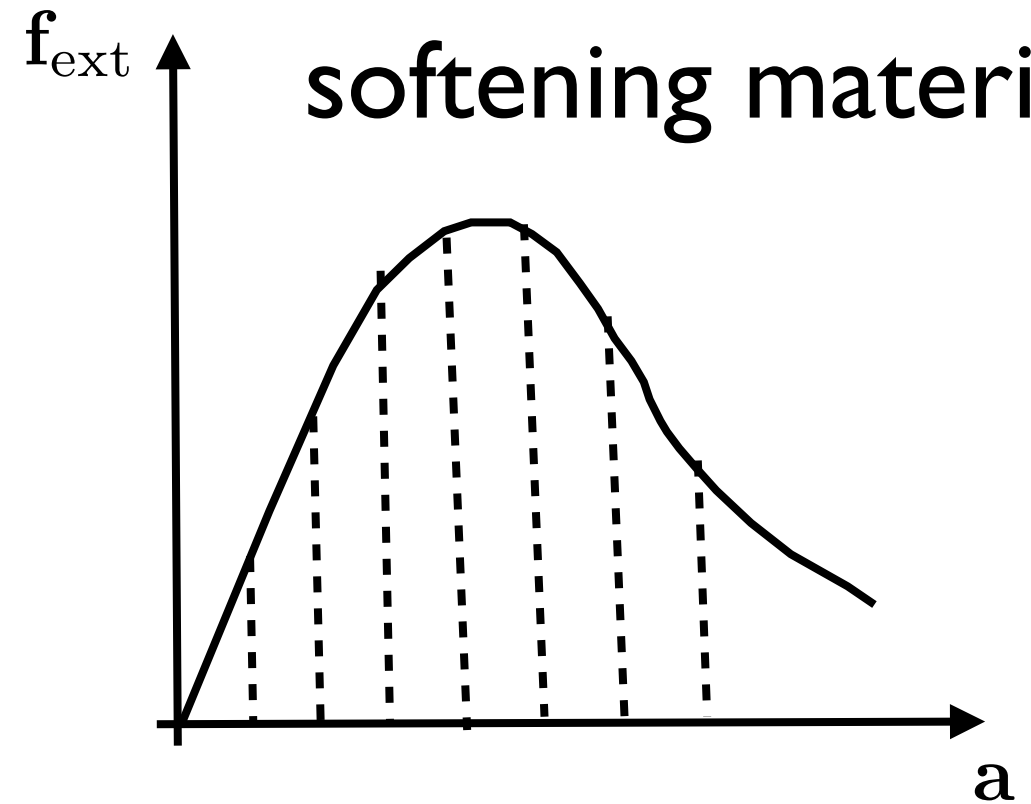
$$\delta a_5 = \bar{a}$$

From second iteration:

$$\mathbf{K} \delta \mathbf{a}_{(i)} = -\mathbf{f}_{\text{int}}$$

$$\delta a_5 = 0$$

f_{ext} softening materials



Direct displacement control

$$\mathbf{f}_{\text{int}}(\mathbf{a}) = \mathbf{0}$$

$$a_5 = \lambda \bar{a}, \quad \lambda = 0, 1, 2, \dots$$

For each load step:

First iteration:

$$\mathbf{K} \delta \mathbf{a}_{(1)} = -\mathbf{f}_{\text{int}}$$

$$\delta a_5 = \bar{a}$$

From second iteration:

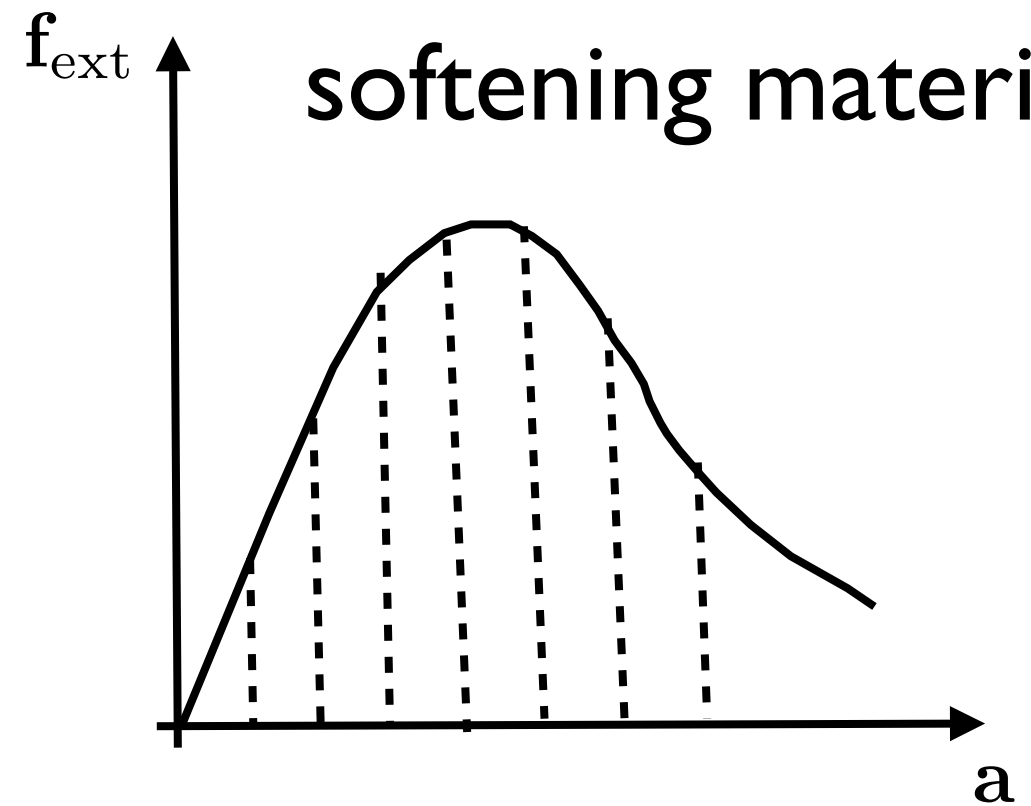
$$\mathbf{K} \delta \mathbf{a}_{(i)} = -\mathbf{f}_{\text{int}}$$

$$\delta a_5 = 0$$

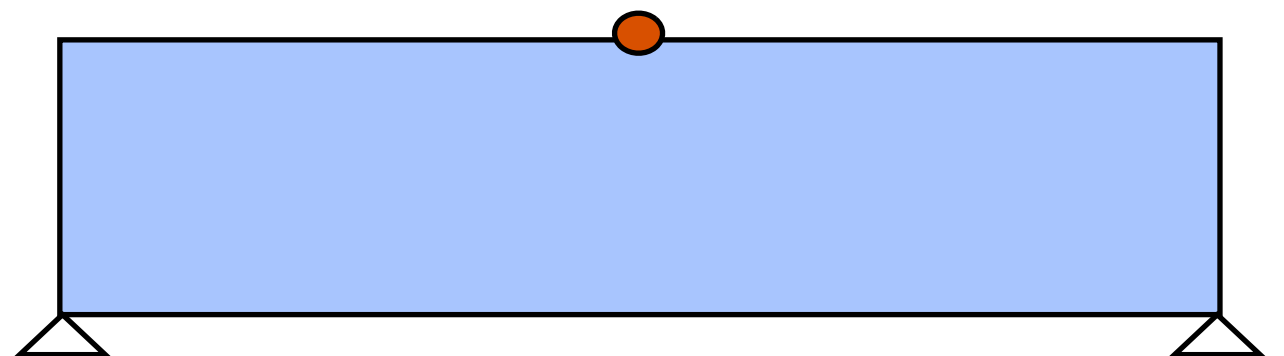
Argyris (1965)

Batoz, Dhatt (1979)

f_{ext}
softening materials



$$a_5 = \lambda \bar{a}, \quad \lambda = 0, 1, 2, \dots$$



Divergence issues

```
module@extraModules.solver.nonLin' : residual scale factor = 4.2697e+01
module@extraModules.solver.nonLin' : iter = 1, scaled residual = 1.1246e-01
module@extraModules.solver.nonLin' : iter = 2, scaled residual = 5.5655e-02
module@extraModules.solver.nonLin' : iter = 3, scaled residual = 3.1334e-02
module@extraModules.solver.nonLin' : iter = 4, scaled residual = 3.9274e-02
module@extraModules.solver.nonLin' : iter = 5, scaled residual = 5.4948e-02
module@extraModules.solver.nonLin' : iter = 6, scaled residual = 3.3929e-02
module@extraModules.solver.nonLin' : iter = 7, scaled residual = 4.6461e-02
module@extraModules.solver.nonLin' : iter = 8, scaled residual = 2.7252e-02
module@extraModules.solver.nonLin' : iter = 9, scaled residual = 4.7695e-02
module@extraModules.solver.nonLin' : iter = 10, scaled residual = 4.1914e-02
module@extraModules.solver.nonLin' : iter = 11, scaled residual = 1.1236e-01
module@extraModules.solver.nonLin' : iter = 12, scaled residual = 6.0917e-02
module@extraModules.solver.nonLin' : iter = 13, scaled residual = 5.6249e-02
module@extraModules.solver.nonLin' : iter = 14, scaled residual = 3.4289e-02
module@extraModules.solver.nonLin' : iter = 15, scaled residual = 4.4957e-02
module@extraModules.solver.nonLin' : iter = 16, scaled residual = 2.7413e-02
```

the system is failed structurally or numerically???

Elasto-plastic models

Linear viscoelasticity