

Computational modeling of material failure

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in collaboration with

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Outline

- Computational models for fracture
 - Continuum mechanics: LEFM, Cohesive zone models
 - Peridynamics
 - Continuous/discontinuous description of failure (Damage models, XFEM, interface elements)
- Multiscale modeling of fracture
 - Hierarchical, semi-concurrent and concurrent methods
 - Computational homogenization models for fracture
- Image-based modeling
 - Conforming mesh methods
 - Level Set/XFEM, Finite Cell Method (non-conforming)
 - Voxel based methods

Continuum mechanics theories

Cauchy, Euler, Lagrange...

S. Silling 2000

Peridynamics

Peridynamics is a formulation of continuum mechanics that is oriented toward deformations with discontinuities, especially fractures.

Integral equation

$$\rho(\mathbf{x})\ddot{\mathbf{u}}(\mathbf{x}, t) = \int_{H_{\mathbf{x}}} \mathbf{f}(\mathbf{u}(\mathbf{x}', t) - \mathbf{u}(\mathbf{x}, t), \mathbf{x}' - \mathbf{x}) dV_{\mathbf{x}'} - \mathbf{b}(\mathbf{x}, t)$$

$$\sigma_{ij,j} + \rho b_i = \rho \ddot{u}_i$$

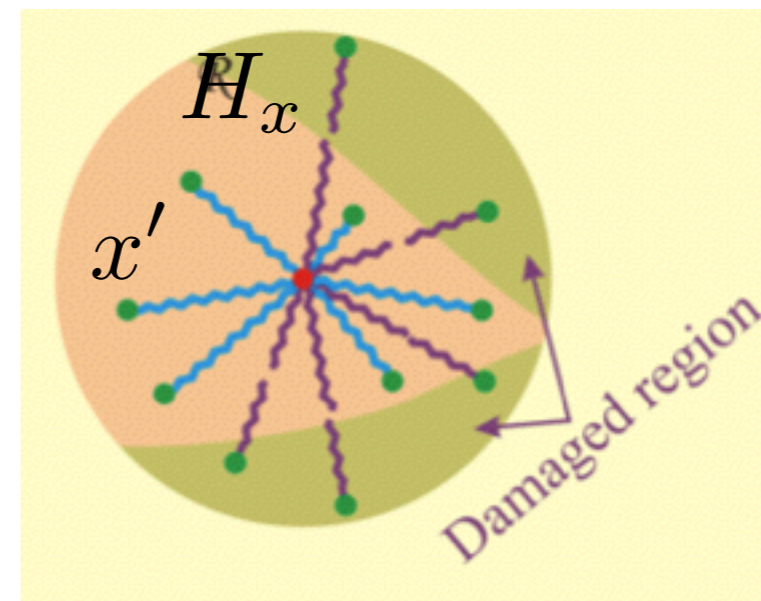
$$\sigma_{ij} = C_{ijkl} \epsilon_{kl}$$

$$\epsilon_{ij} = \frac{1}{2} (u_{i,j} + u_{j,i})$$

+

Fracture Mechanics
Damage Mechanics

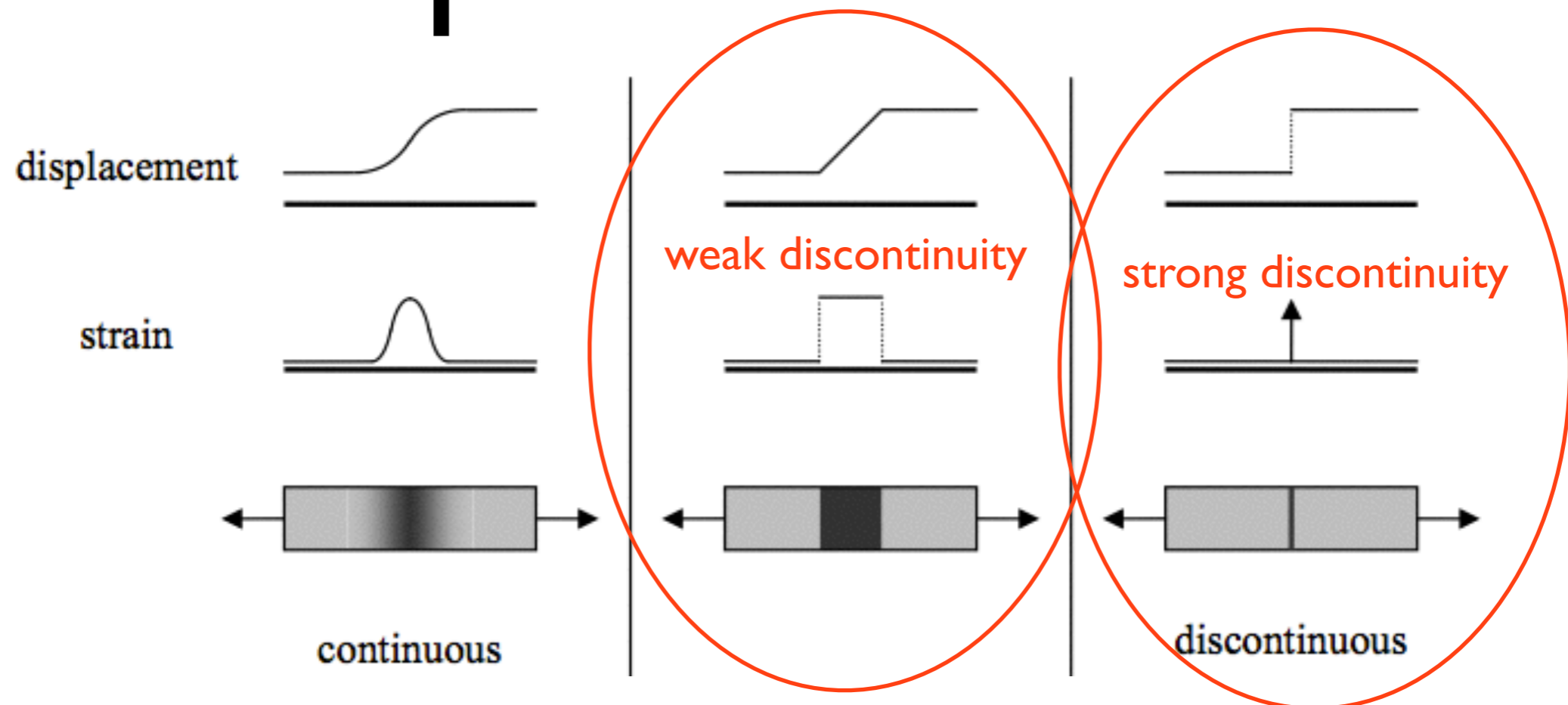
PDE



spatial derivatives of displacements:
do not exist at discontinuities (cracks)

No spatial derivatives of displacements

Continuous/discontinuous description of fracture



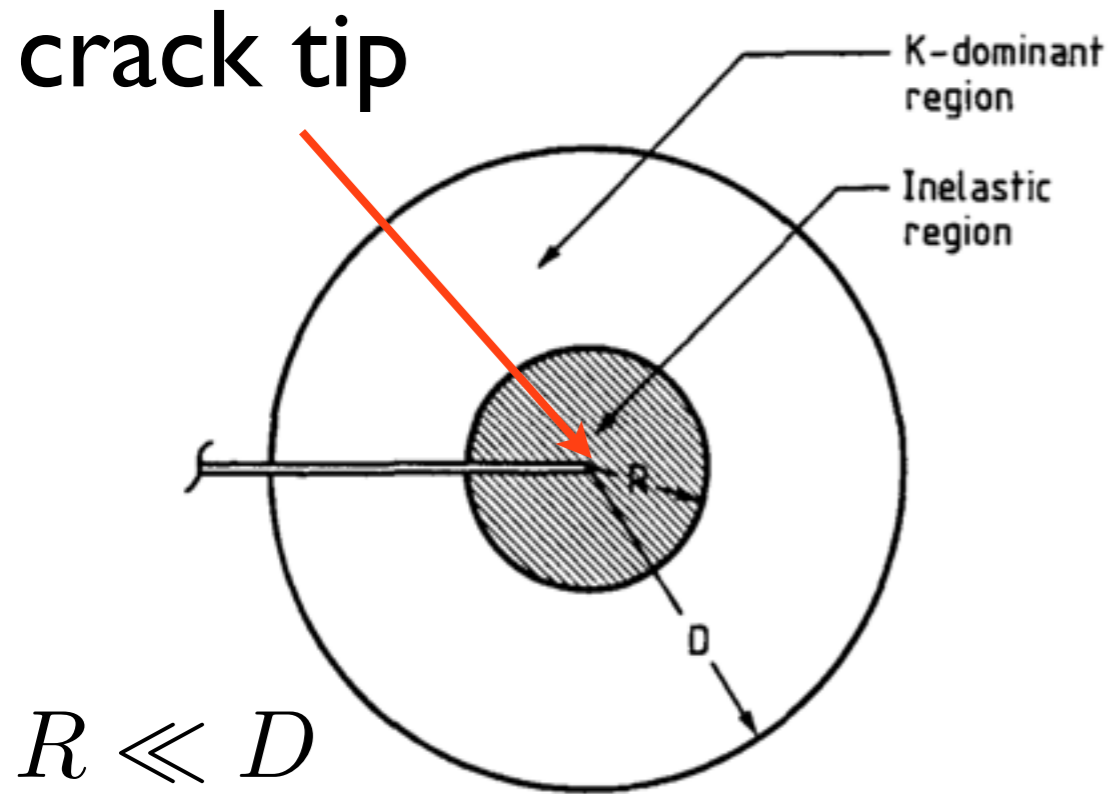
Damage Mechanics

Isotropic damage models
Softening plasticity models
Damage-plastic models

Fracture Mechanics

LEFM, EPFM, CZM

Fracture mechanics models



Elastic Plastic Fracture Mechanics (EPFM):

- ductile materials
- an existing crack is required

Linear Elastic Fracture Mechanics (LEFM):

- brittle materials
- ductile materials under Small Scale Yielding (SSY) condition
- an existing crack is required

Cohesive Zone Models (CZMs):

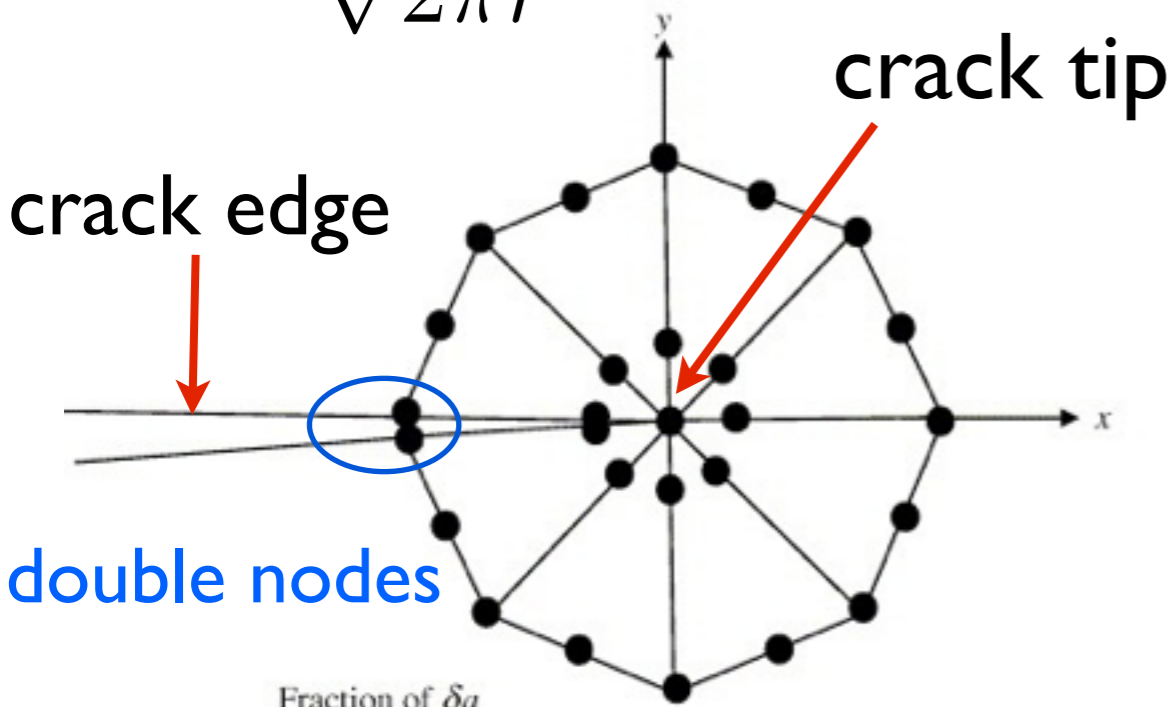
- quasi-brittle materials (concrete)
- ductile materials
- no existing crack is needed

Linear Elastic Fracture Mechanics (LEFM)

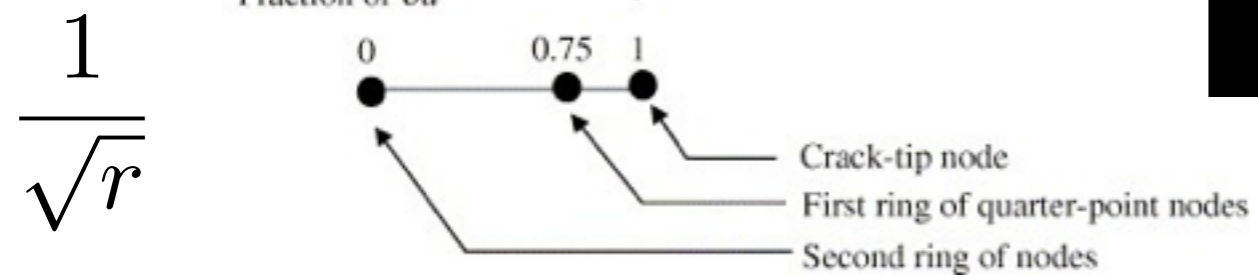
Remeshing is a key point.

$$\sigma_{ij} = \frac{K}{\sqrt{2\pi r}} f_{ij}(\theta) + \text{H.O.T}$$

SIF



double nodes



crack must locate on element edges

Barsoum element [1970s]



Very useful for fatigue life estimation $\frac{da}{dN} = C(\Delta K)^m$

Cohesive Zone Models (CZMs)

Barrenblatt 1962

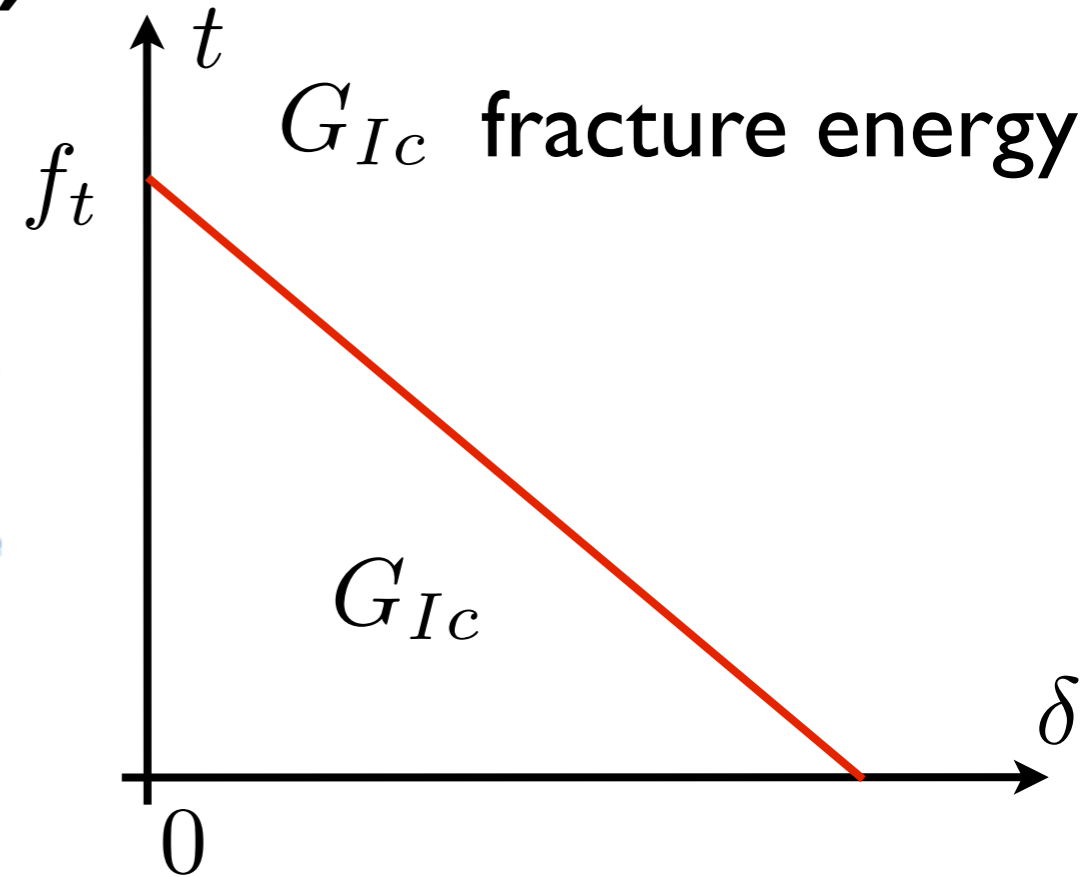
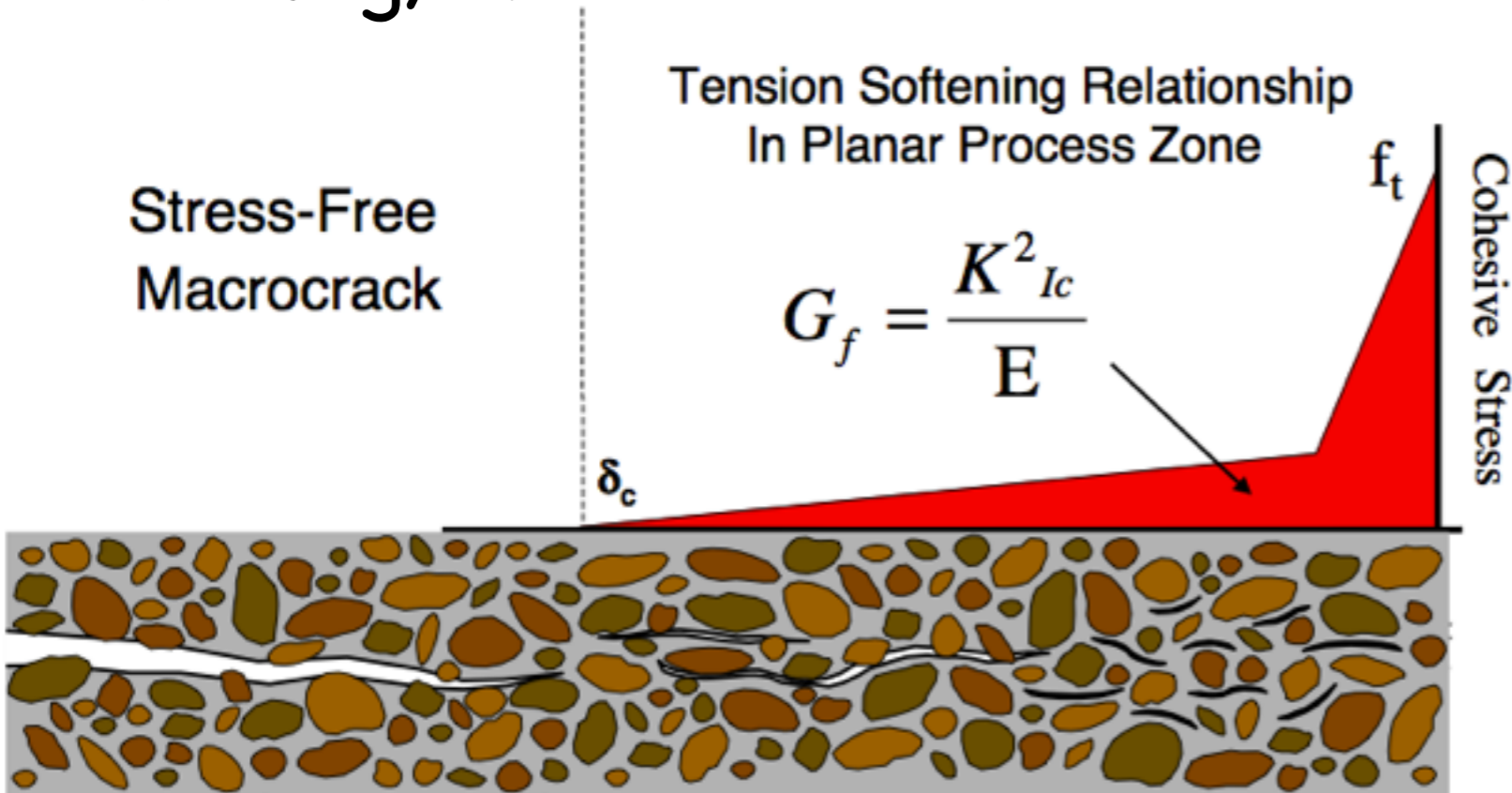
Dugdale 1960

Hilleborg, 1976

(CZMs)

f_t tensile strength

G_{Ic} fracture energy



[Extrinsic] Cohesive law

[Initially rigid] TSL

(Traction Separation Law)

Constitutive equations

$$\dot{\sigma} = \mathbf{D}\dot{\epsilon} \longrightarrow \text{deformation}$$

$$\dot{t}^c = \mathbf{T}[\dot{u}] \longrightarrow \text{separation}$$

crack initiation

$$\sigma_1^{\max} \geq f_t$$

crack direction

criterion

Cohesive crack model

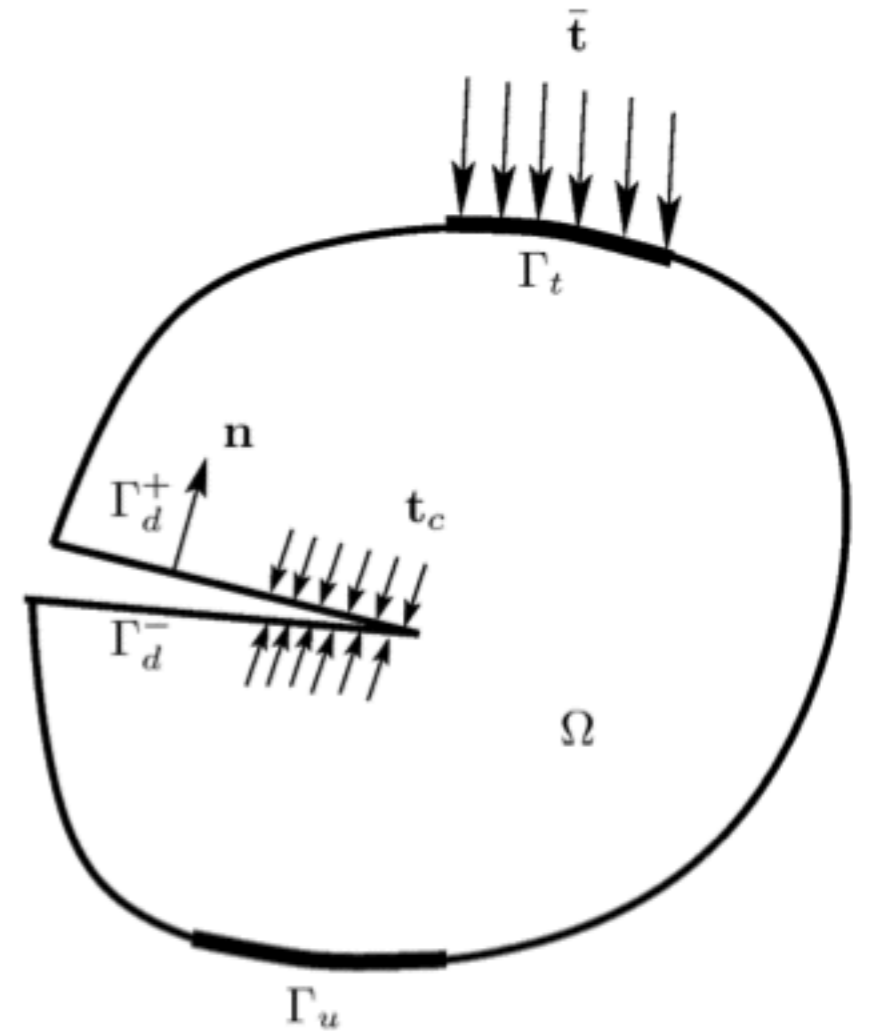
Governing equations (strong form)

$$\nabla \cdot \boldsymbol{\sigma} + \rho \mathbf{b} - \rho \ddot{\mathbf{u}} = 0 \quad \mathbf{x} \in \Omega$$

$$\mathbf{n} \cdot \boldsymbol{\sigma} = \bar{\mathbf{t}} \quad \mathbf{x} \in \Gamma_t$$

$$\mathbf{u} = \bar{\mathbf{u}} \quad \mathbf{x} \in \Gamma_u$$

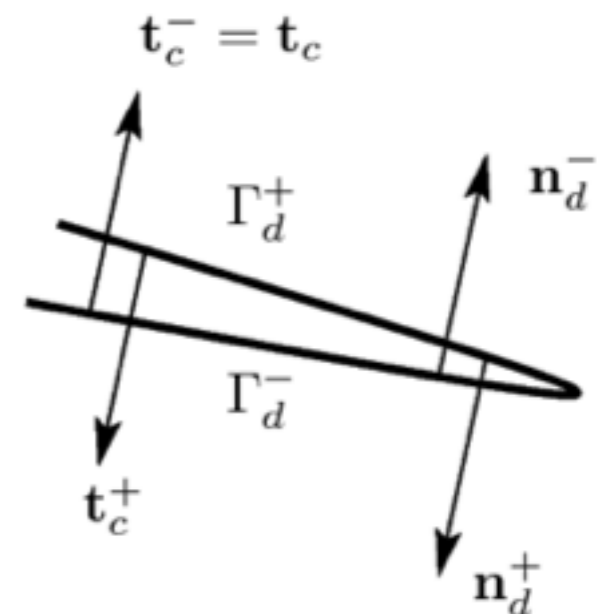
$$\mathbf{n}_d^+ \cdot \boldsymbol{\sigma} = \mathbf{t}_c^+; \quad \mathbf{n}_d^- \cdot \boldsymbol{\sigma} = \mathbf{t}_c^-; \quad \mathbf{t}_c^+ = -\mathbf{t}_c = -\mathbf{t}_c^- \quad \mathbf{x} \in \Gamma_d$$



Constitutive equations

$$\dot{\boldsymbol{\sigma}} = \mathbf{D} \dot{\boldsymbol{\epsilon}} \quad \longrightarrow \quad \text{deformation}$$

$$\dot{\mathbf{t}}^c = \mathbf{T}[[\dot{\mathbf{u}}]] \quad \longrightarrow \quad \text{separation}$$



Cohesive crack model

Weak form

$$\delta W^{\text{kin}} = \delta W^{\text{ext}} - \delta W^{\text{int}} - \delta W^{\text{coh}}$$

new term

where

$$\delta W^{\text{kin}} = \int_{\Omega} \delta \mathbf{u} \cdot \rho \ddot{\mathbf{u}} d\Omega$$

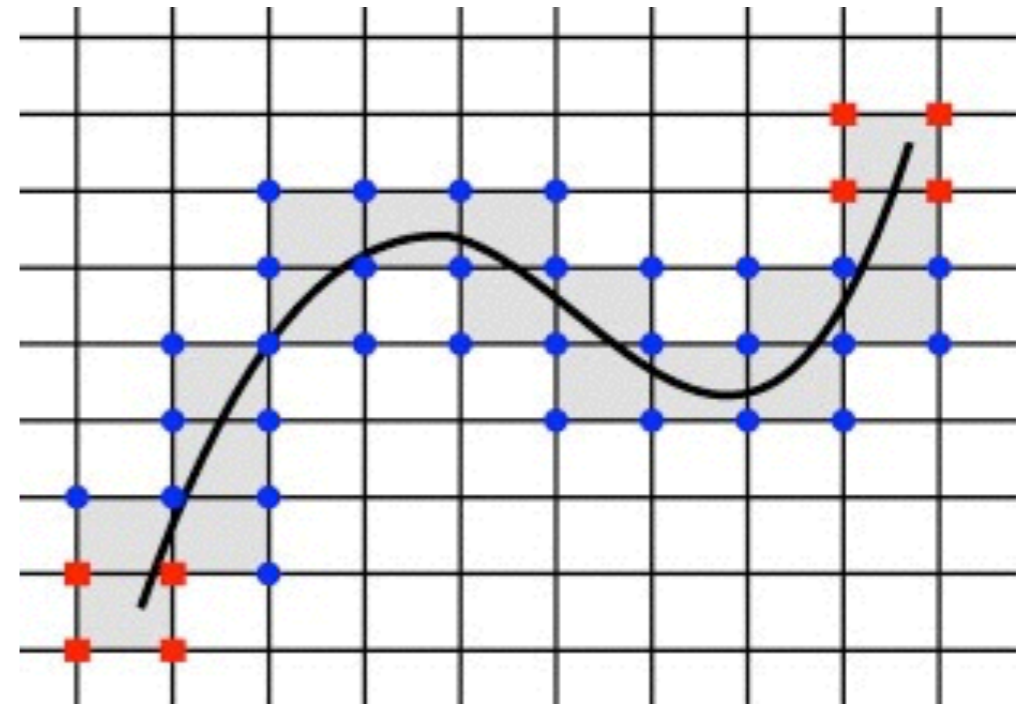
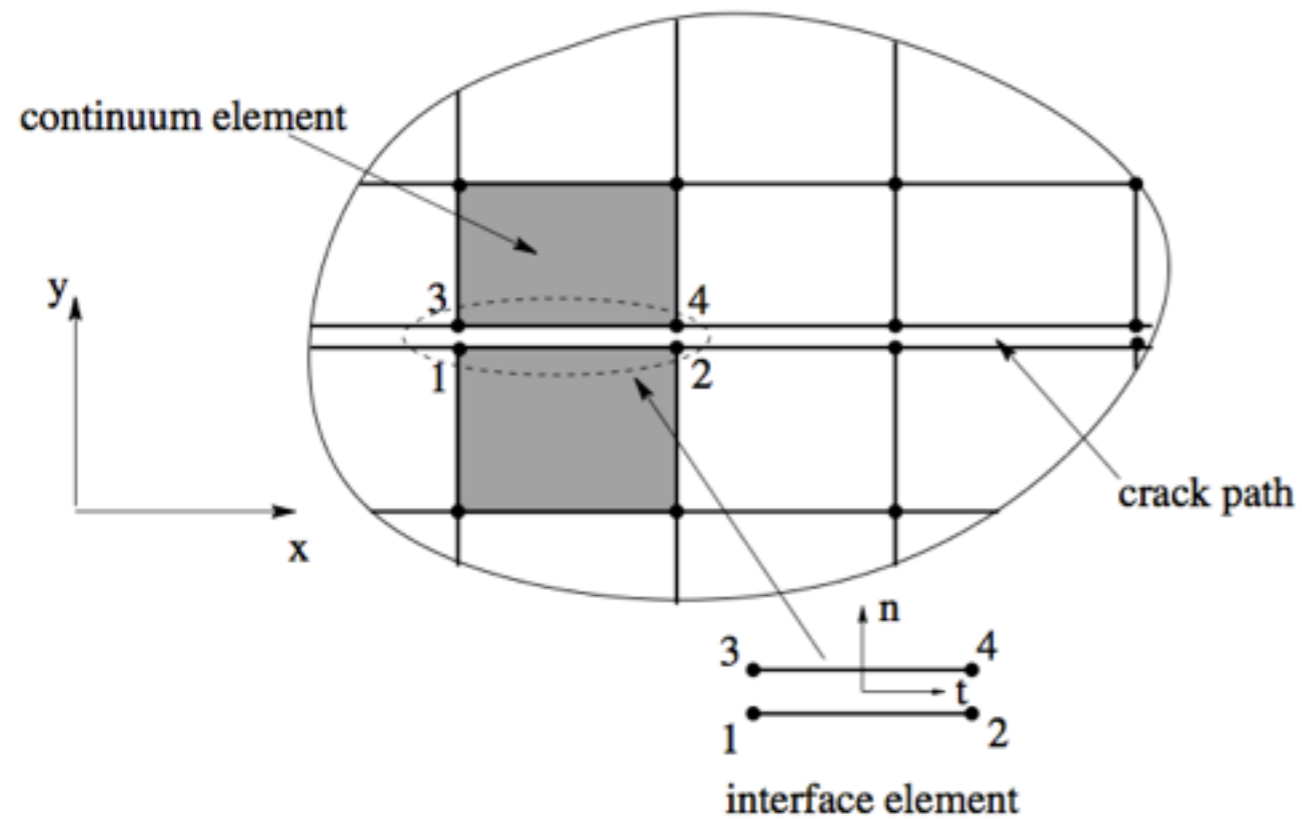
$$\delta W^{\text{int}} = \int_{\Omega} \nabla^s \delta \mathbf{u} : \boldsymbol{\sigma} d\Omega$$

$$\delta W^{\text{ext}} = \int_{\Omega} \delta \mathbf{u} \cdot \rho \mathbf{b} d\Omega + \int_{\Gamma_t} \delta \mathbf{u} \cdot \bar{\mathbf{t}} d\Gamma_t$$

$$\delta W^{\text{coh}} = \int_{\Gamma_d} \delta [[\mathbf{u}]] \cdot \mathbf{t}^c d\Gamma_d$$

different techniques

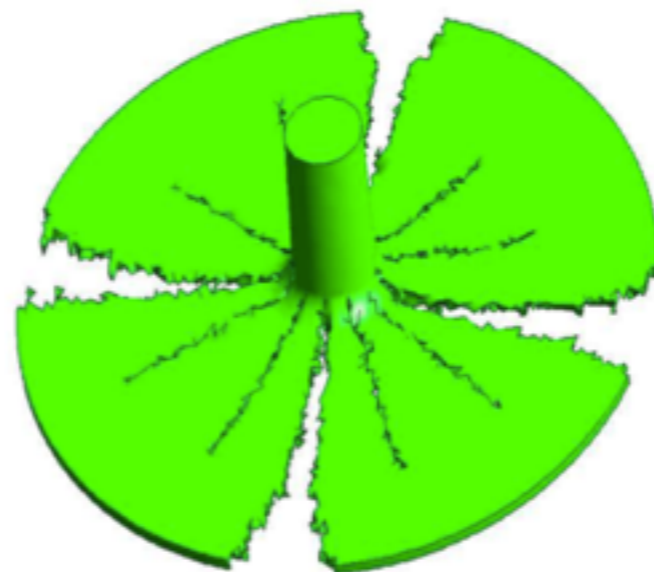
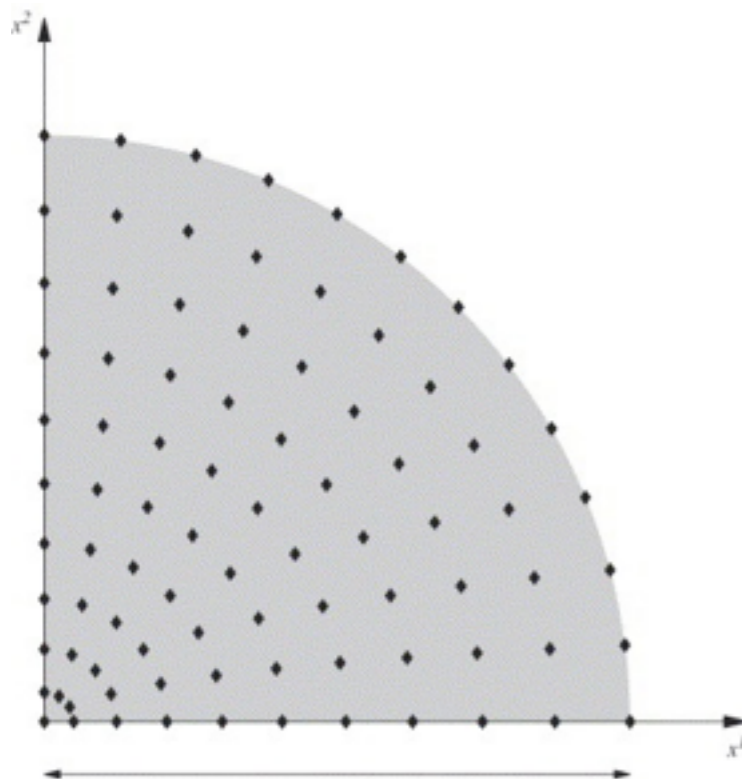
Crack discretization techniques



Zero-thickness interface elements, 1968

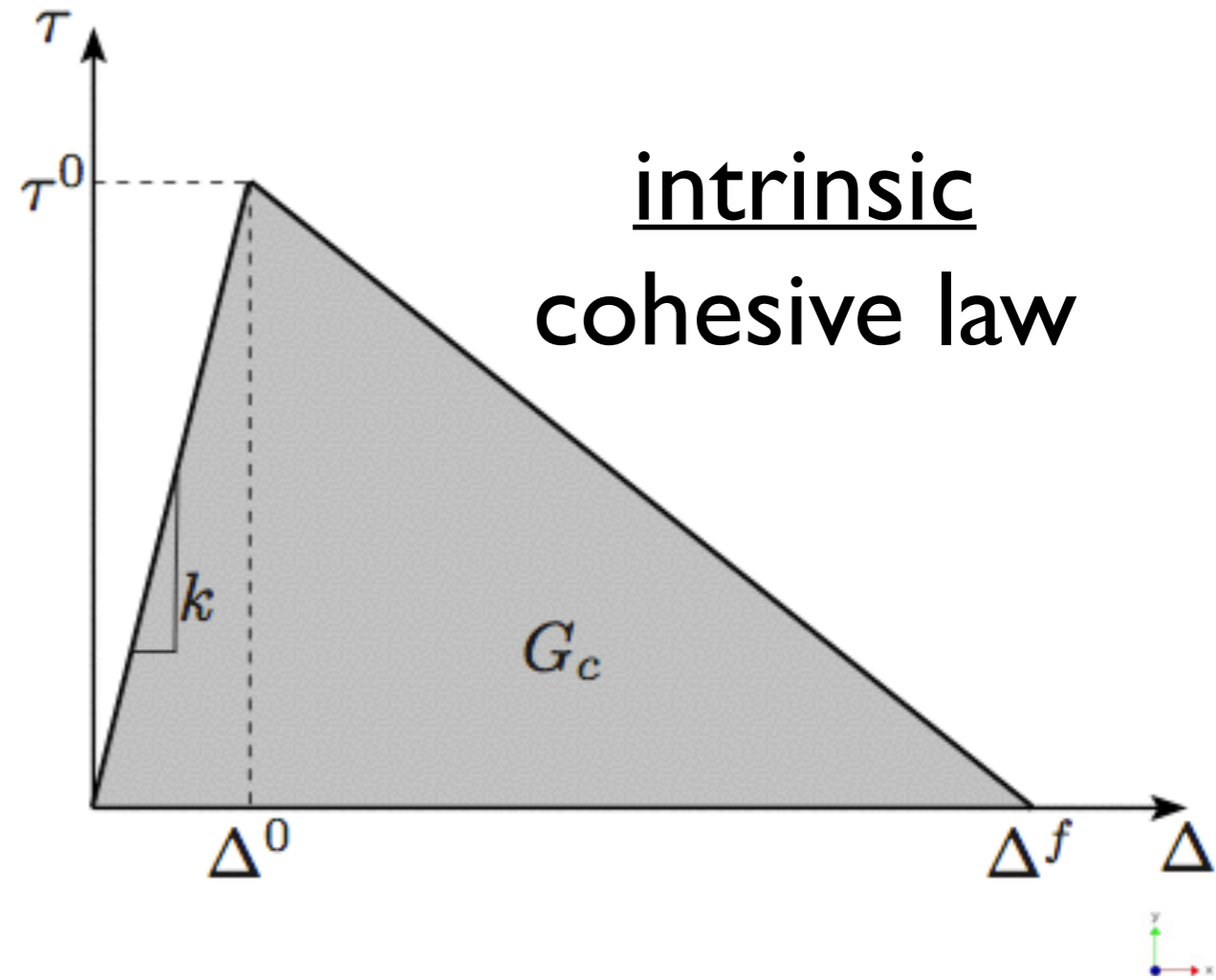
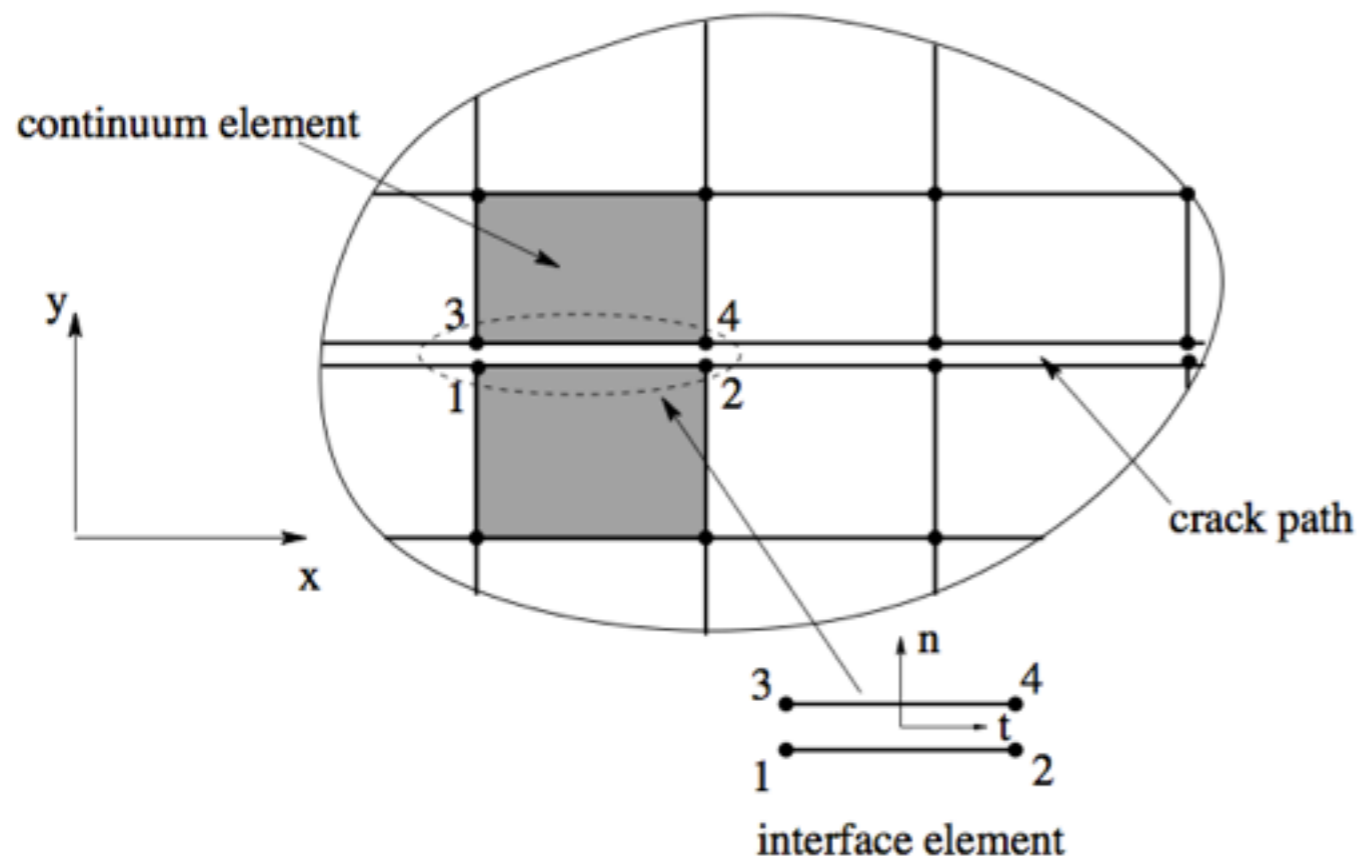
PUM FEM, 1999

Embedded strong discontinuity, 1987



Meshless/Meshfree methods, 1994

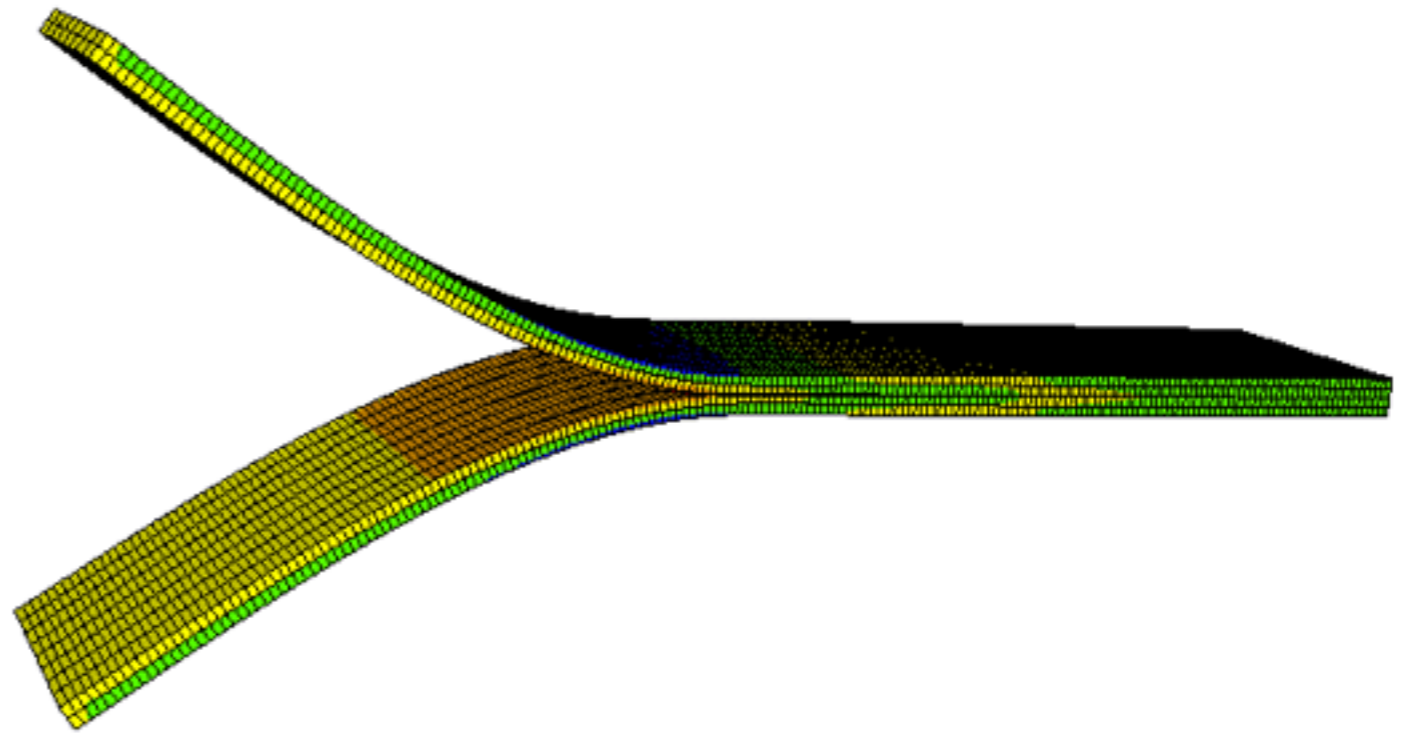
Interface elements



$$\mathbf{u}^+, \mathbf{u}^- \longrightarrow \llbracket \mathbf{u} \rrbracket \longrightarrow \Delta \longrightarrow \tau$$

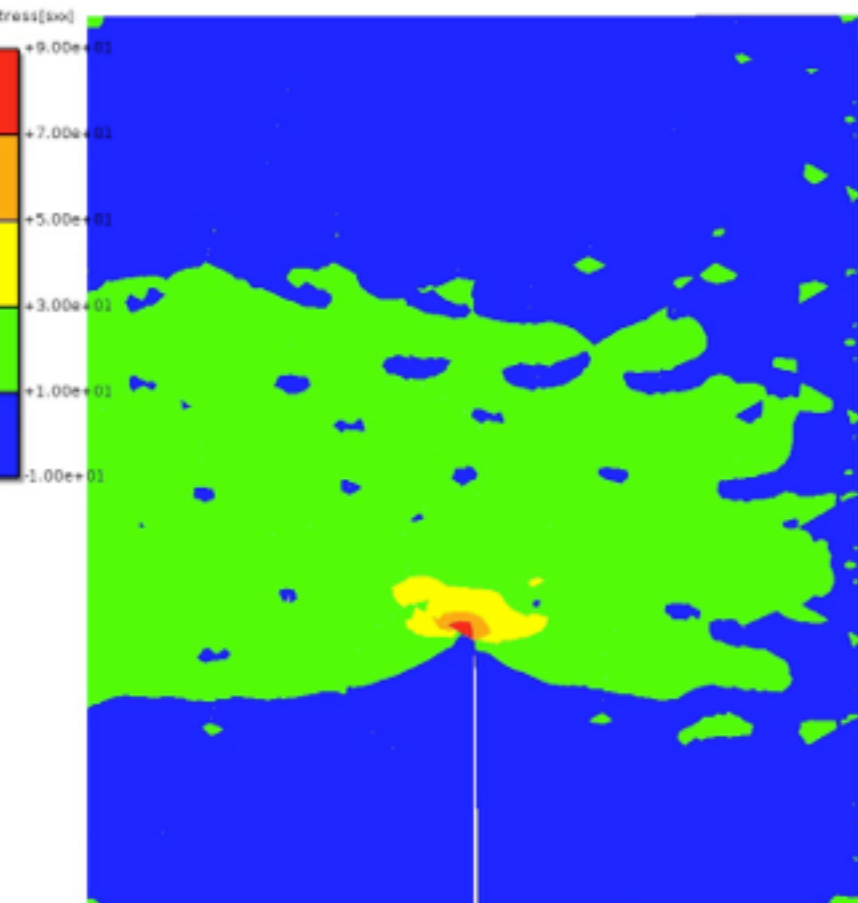
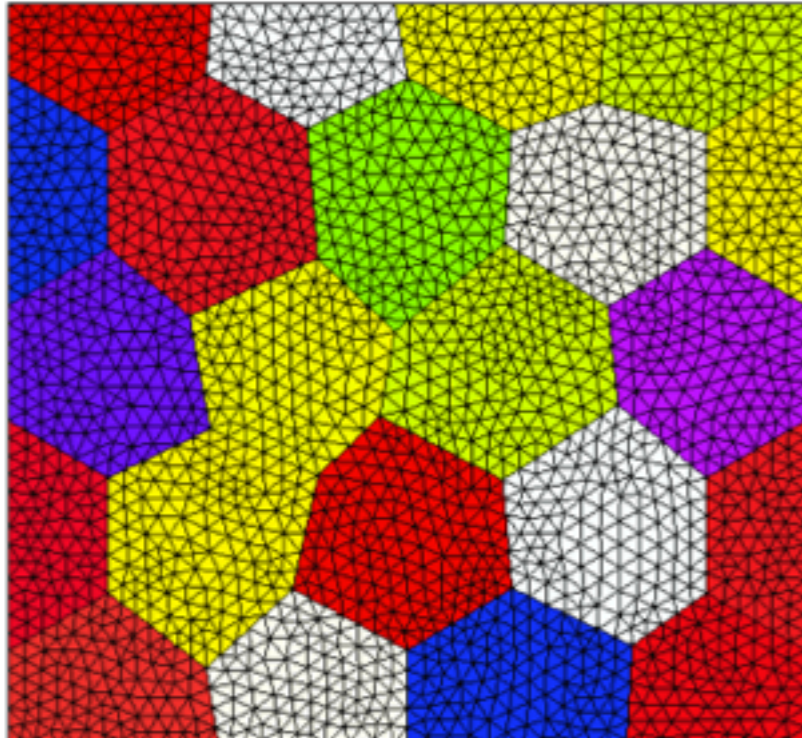
- (+) easy to implement 2D/3D
- (+) available in ABAQUS, LD-DYNA

- preprocessing: GMSH
- solver: jem/jive (C++)



Interface elements

inter-granular fracture of polycrystalline material



failure of a fiber reinforced composite

Interface elements with discontinuous Galerkin

Partition of Unity Methods

Melenk and
Babuska 1996

Approximation of the displacement field

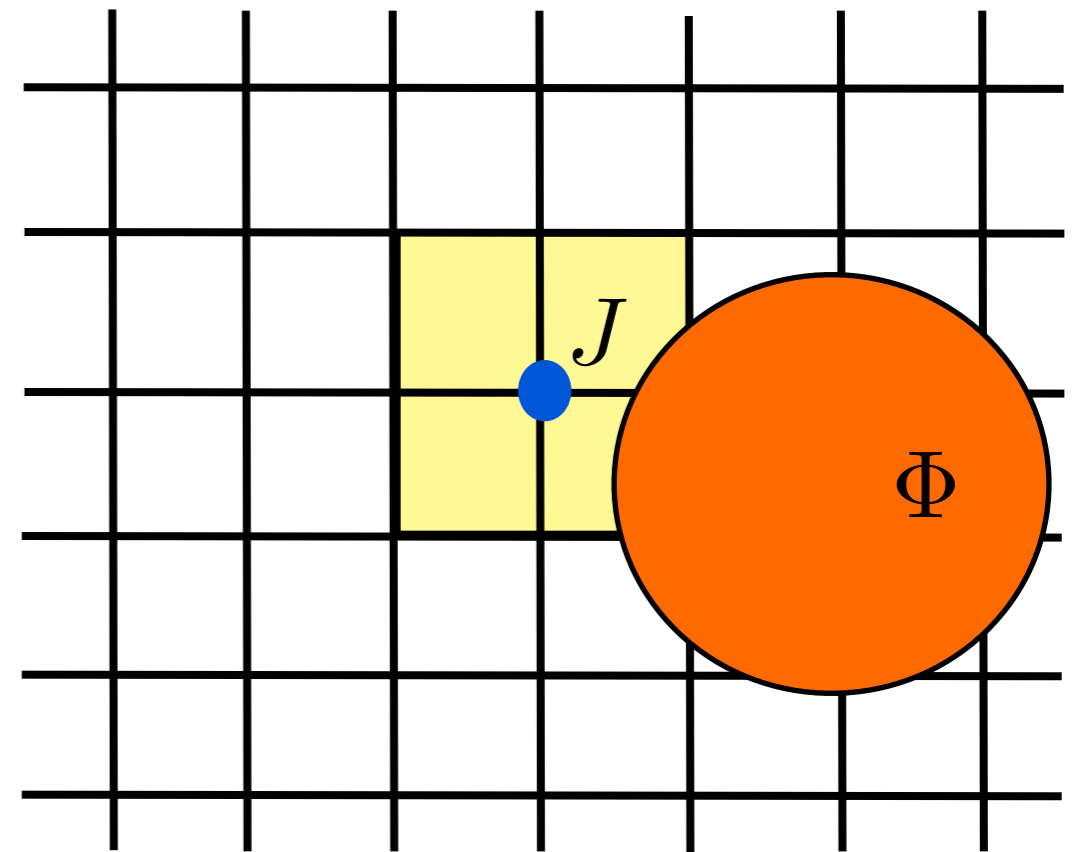
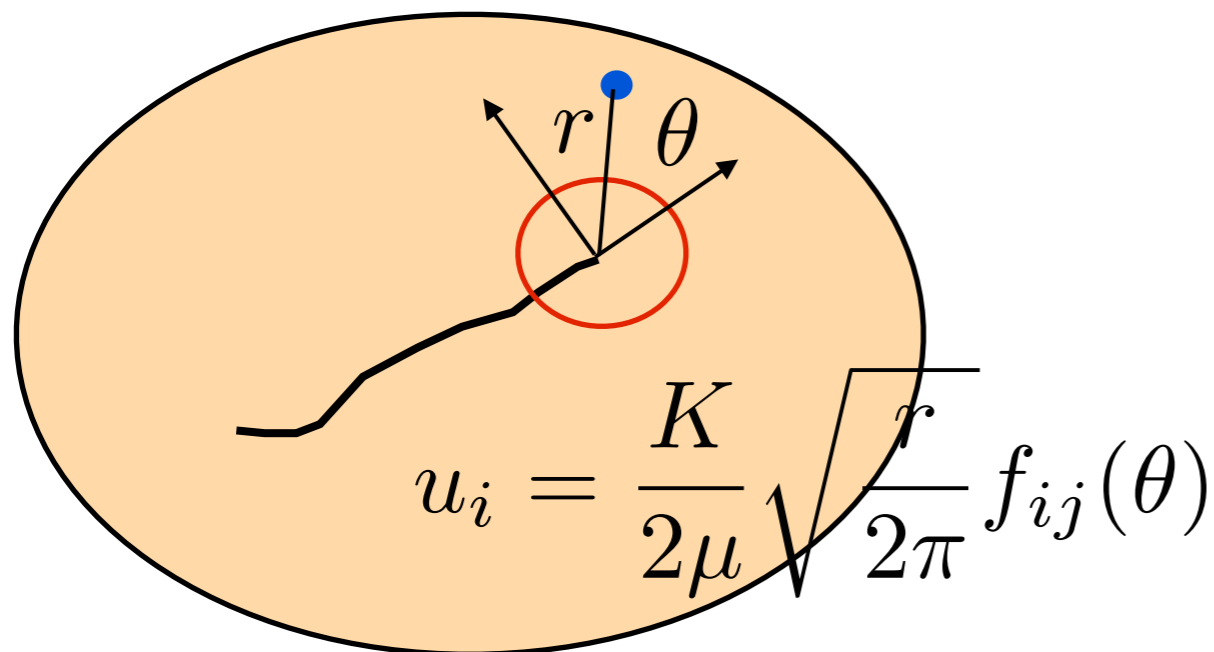
Sum of shape functions is equal to one

$$\sum_J N_J(\mathbf{x}) = 1 \quad (\text{PUM})$$

$$\sum_J N_J(\mathbf{x}) \Phi(\mathbf{x}) = \Phi(\mathbf{x})$$

$$\mathbf{u}^h(\mathbf{x}) = \sum_{I \in \mathcal{S}} N_I(\mathbf{x}) \mathbf{u}_I$$

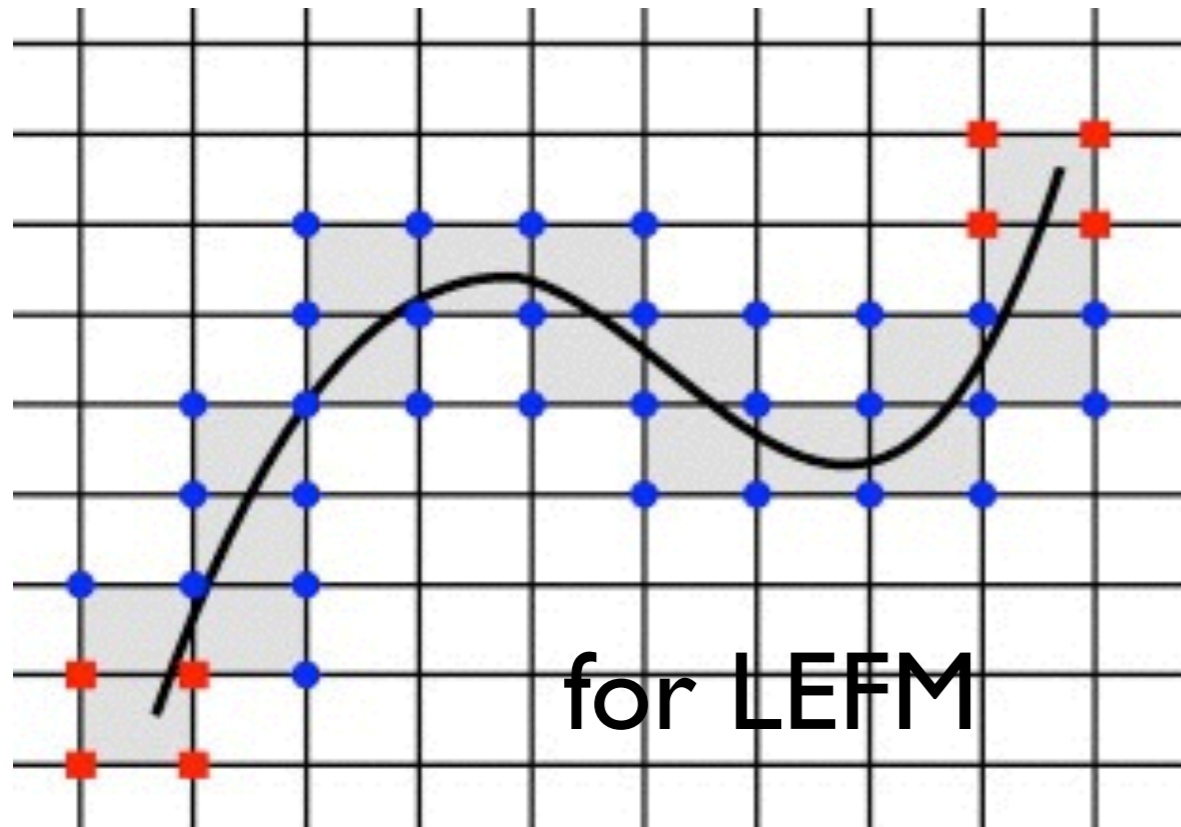
$$+ \sum_{J \in \mathcal{S}^c} N_J(\mathbf{x}) \Phi(\mathbf{x}) \mathbf{a}_J$$



Extended FEM (XFEM)

Belytschko et al., 1999

nothing but an instance of PUM for crack problems



Enrichment functions

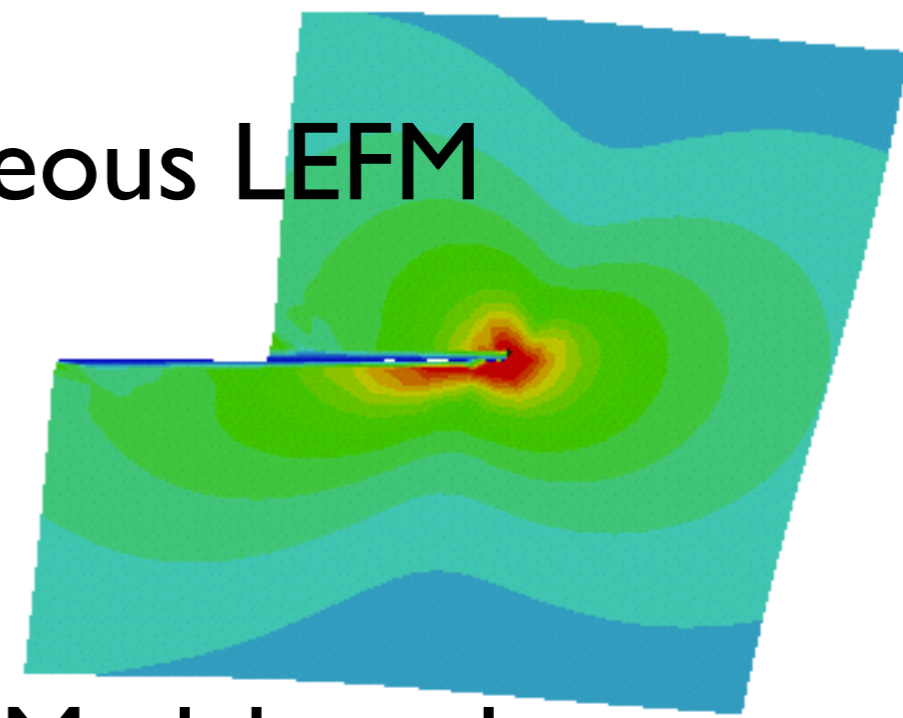
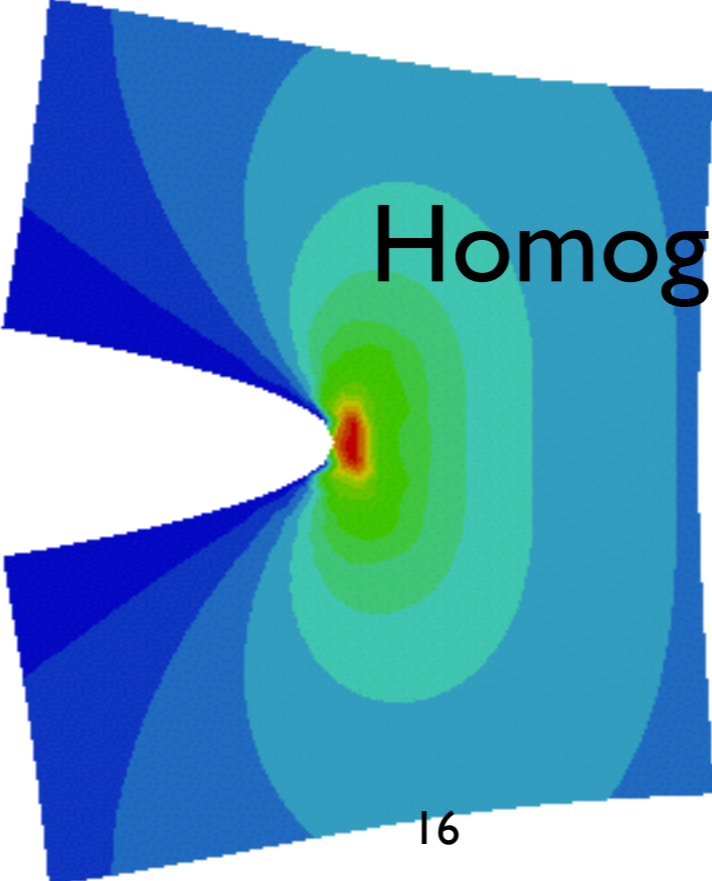
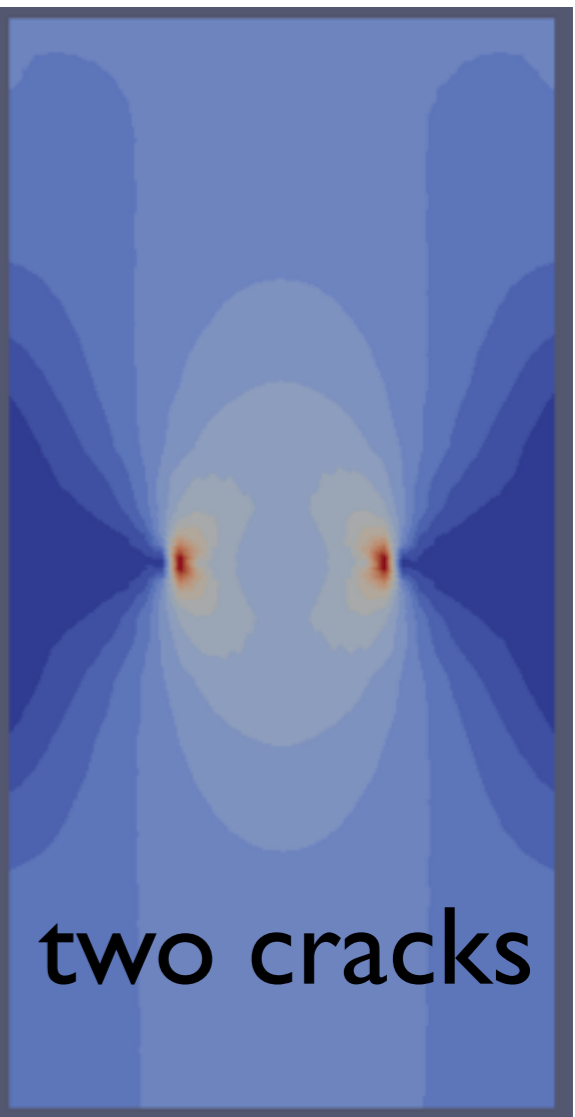
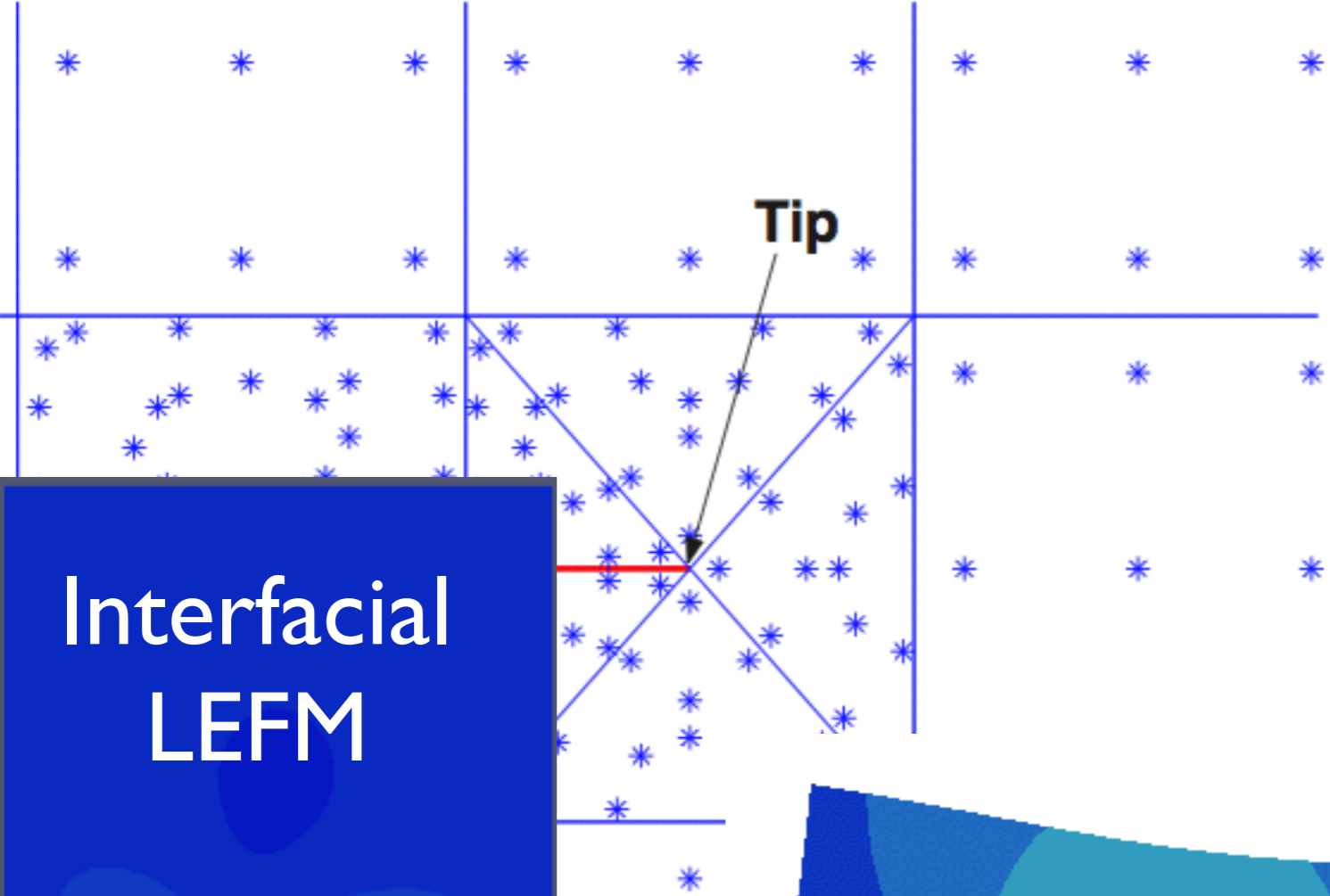
$$H(\mathbf{x}) = \begin{cases} +1 & \text{if } (\mathbf{x} - \mathbf{x}^*) \cdot \mathbf{n} \geq 0 \\ -1 & \text{otherwise} \end{cases}$$

$$[B_\alpha] = \left[\sqrt{r} \sin \frac{\theta}{2}, \sqrt{r} \cos \frac{\theta}{2}, \sqrt{r} \sin \frac{\theta}{2} \sin \theta, \sqrt{r} \cos \frac{\theta}{2} \sin \theta \right]$$

$$\mathbf{u}^h(\mathbf{x}) = \sum_{I \in \mathcal{S}} N_I(\mathbf{x}) \mathbf{u}_I + \sum_{J \in \mathcal{S}^c} N_J(\mathbf{x}) H(\mathbf{x}) \mathbf{a}_J + \sum_{K \in \mathcal{S}^t} N_K(\mathbf{x}) \left(\sum_{\alpha=1}^4 B_\alpha \mathbf{b}_K^\alpha \right)$$

\mathcal{S}^c ●
 \mathcal{S}^t ■

Sub-triangulation for numerical integration

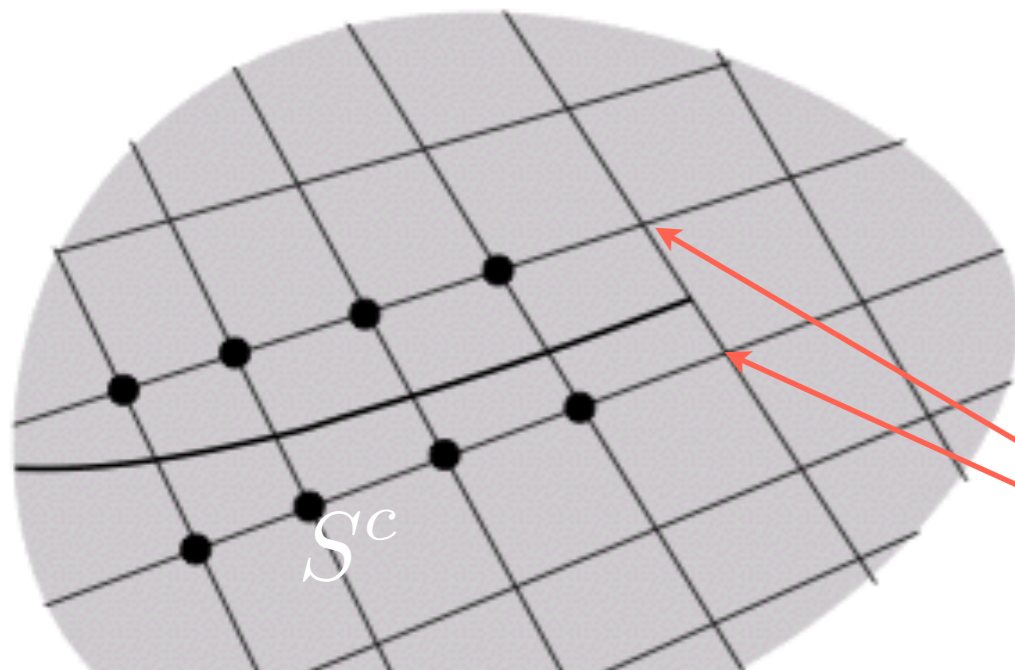


XFEM for cohesive cracks

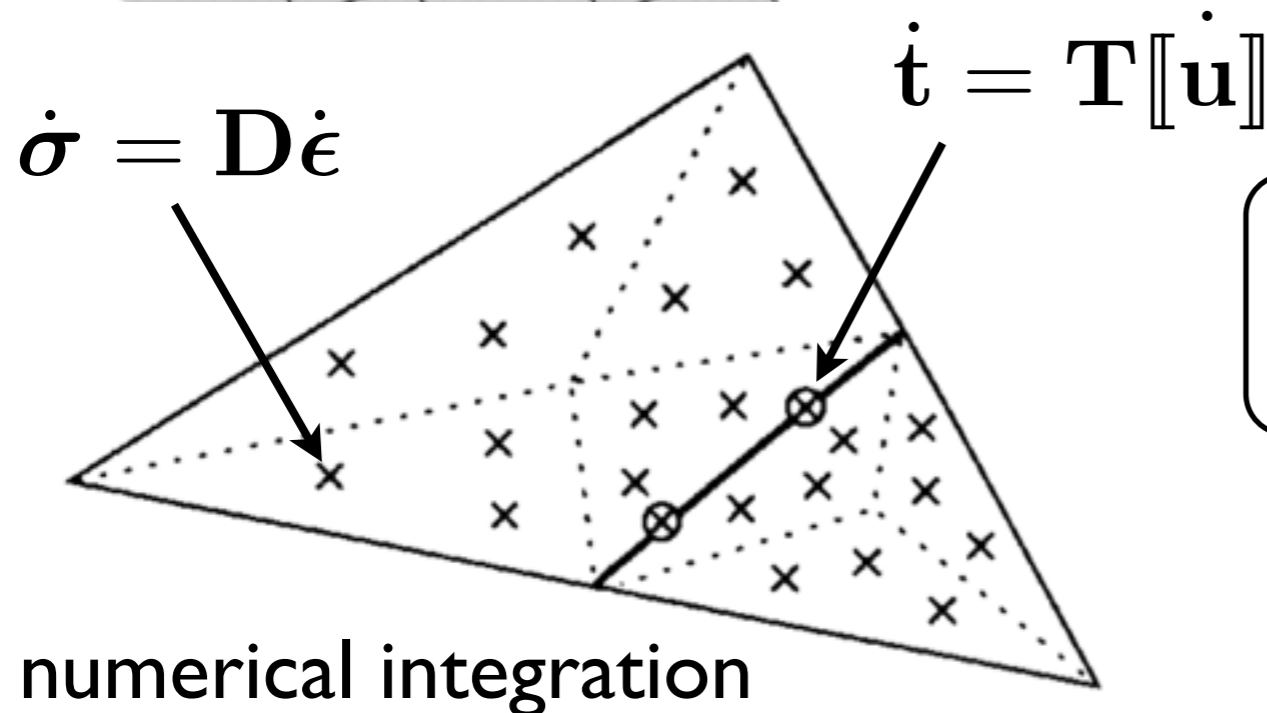
$$\mathbf{u}^h(\mathbf{x}) = \sum_{I \in \mathcal{S}} N_I(\mathbf{x}) \mathbf{u}_I + \sum_{J \in \mathcal{S}^c} N_J(\mathbf{x}) H(\mathbf{x}) \mathbf{a}_J$$

Wells, Sluys, 2001

No good crack tip solution is known, no tip enrichment!!!

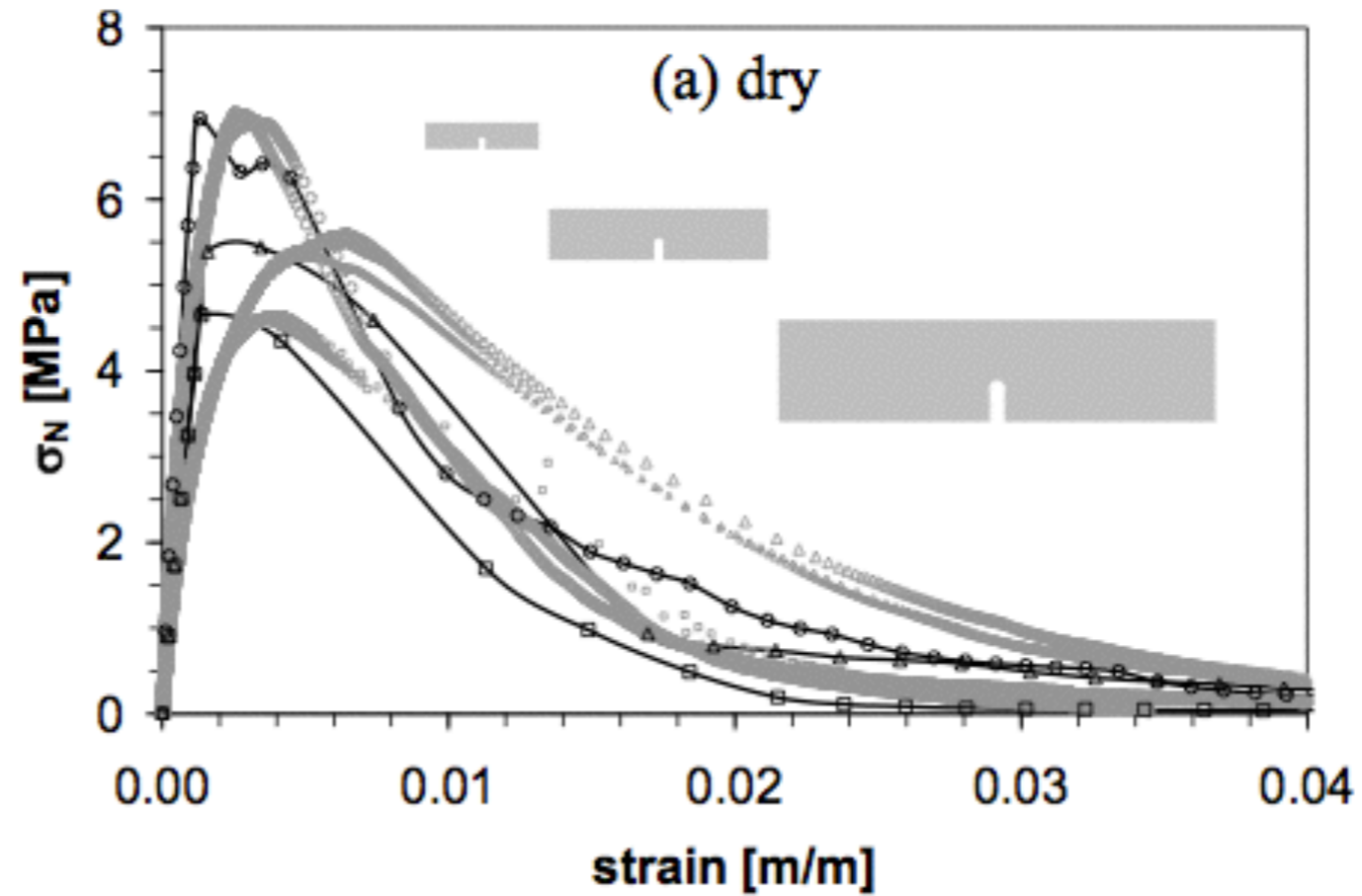
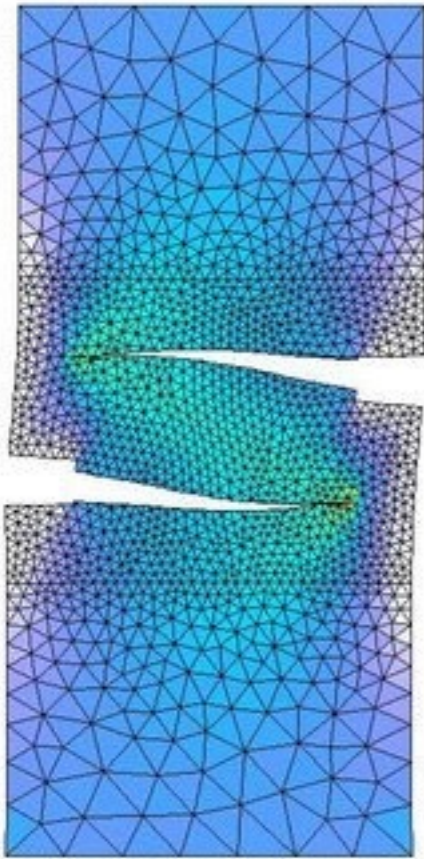


not enriched to ensure zero crack tip opening!!!

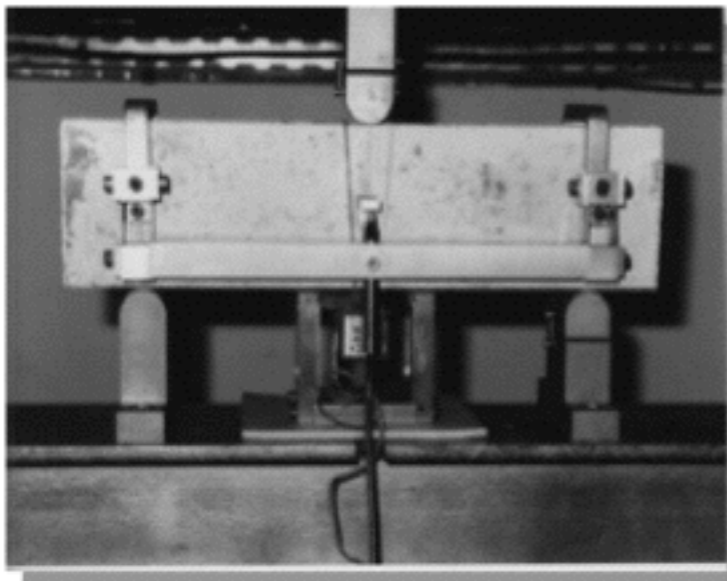


$$H(\mathbf{x}) = \begin{cases} +1 & \text{if } (\mathbf{x} - \mathbf{x}^*) \cdot \mathbf{n} \geq 0 \\ -1 & \text{otherwise} \end{cases}$$

XFEM/Cohesive zones



Size effect

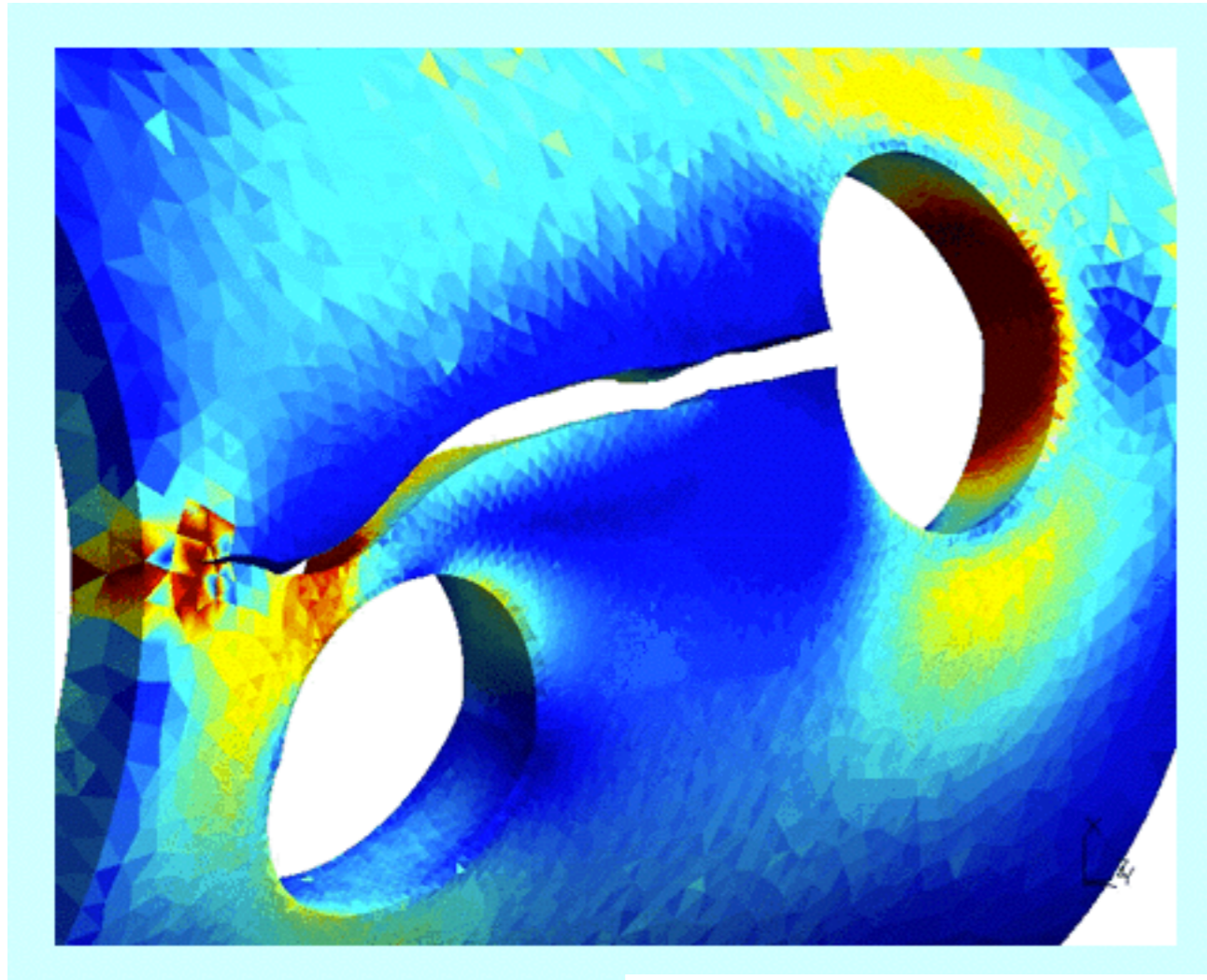


Usual lab tests (10 cm)

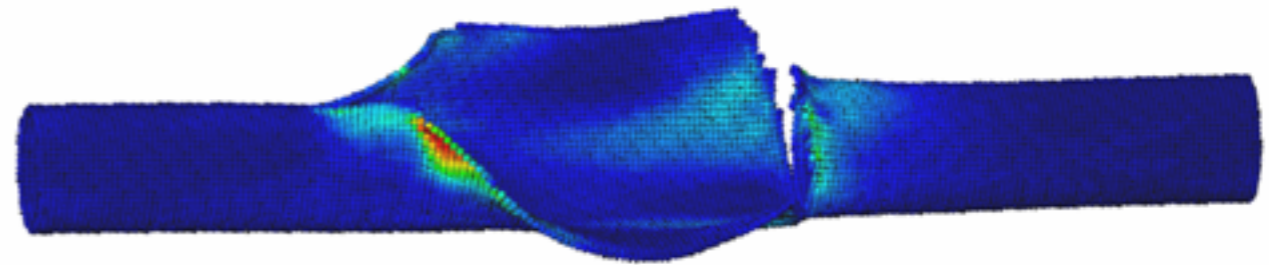


“Usual” structures (10m)

... convincing examples

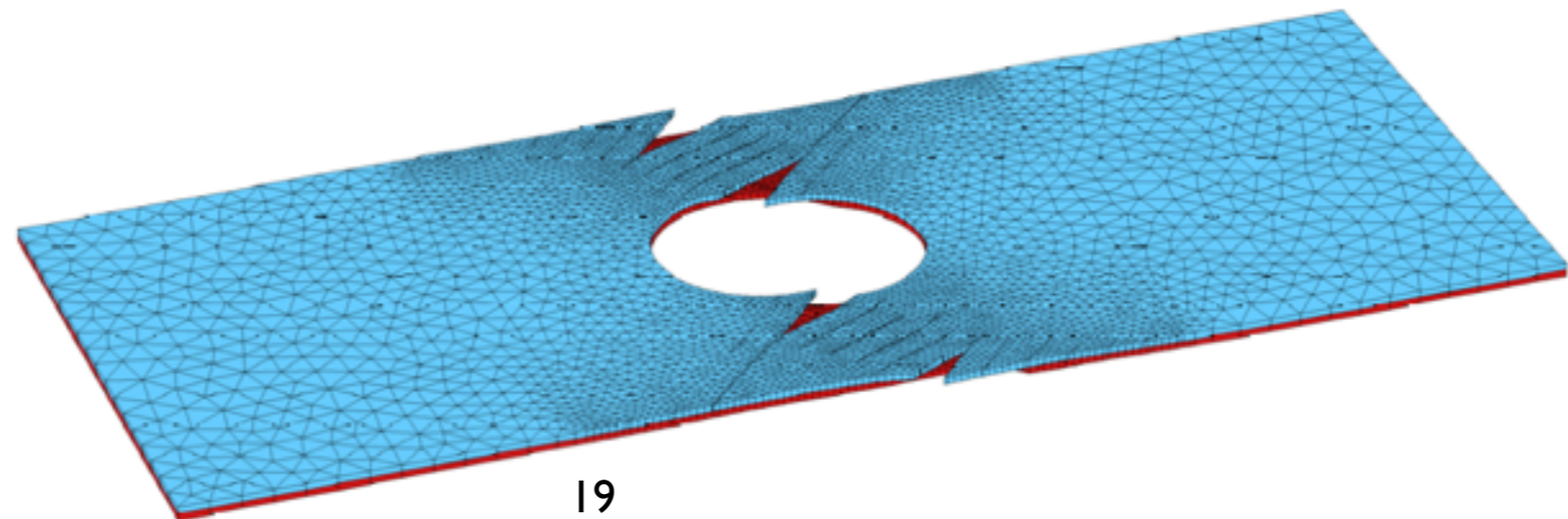


Northwestern Univ.



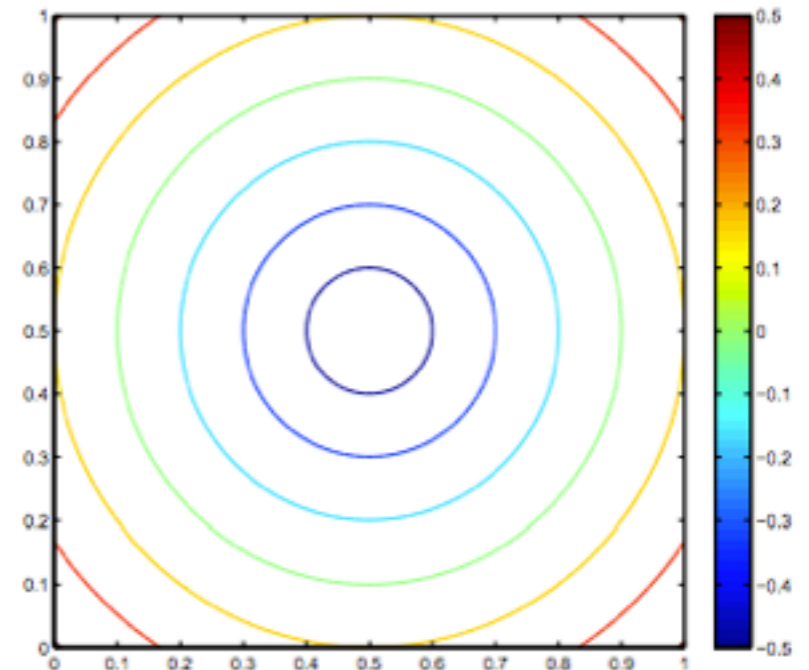
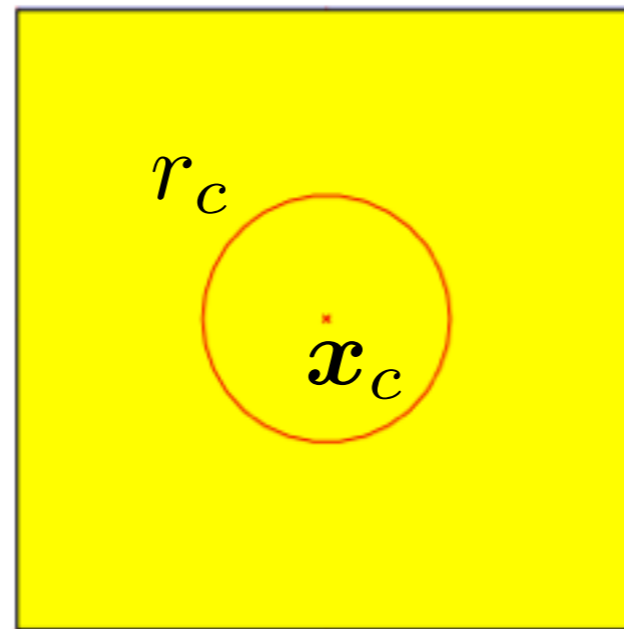
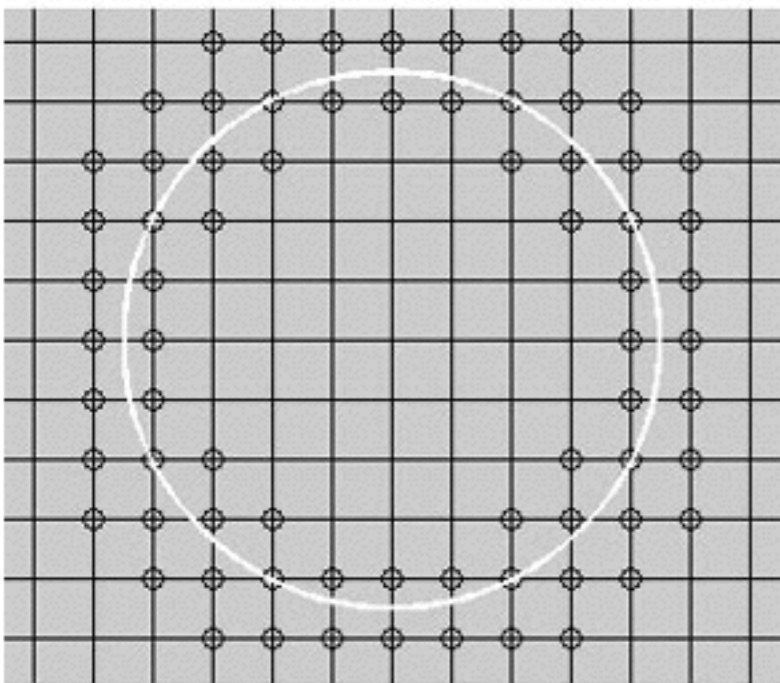
F.P. van der Mer, TU Delft

M. Duflot



XFEM for material interfaces

Sukumar et al. 2002



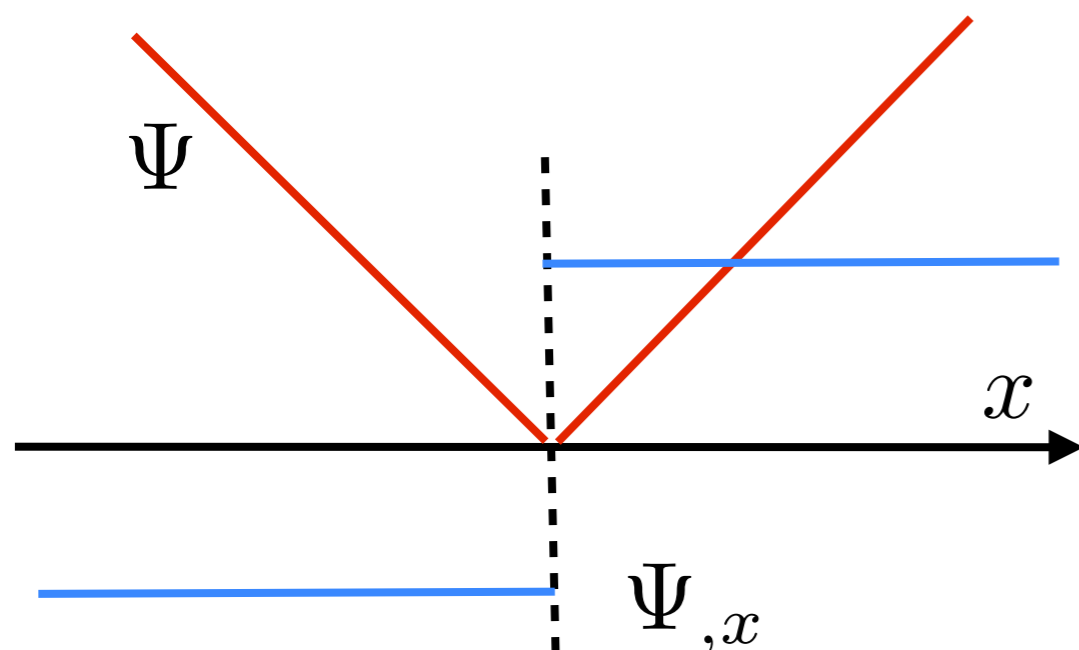
across interface, strain field is discontinuous level set representation

abs-enrichment function $\psi = |\phi(x)|$

signed distance function

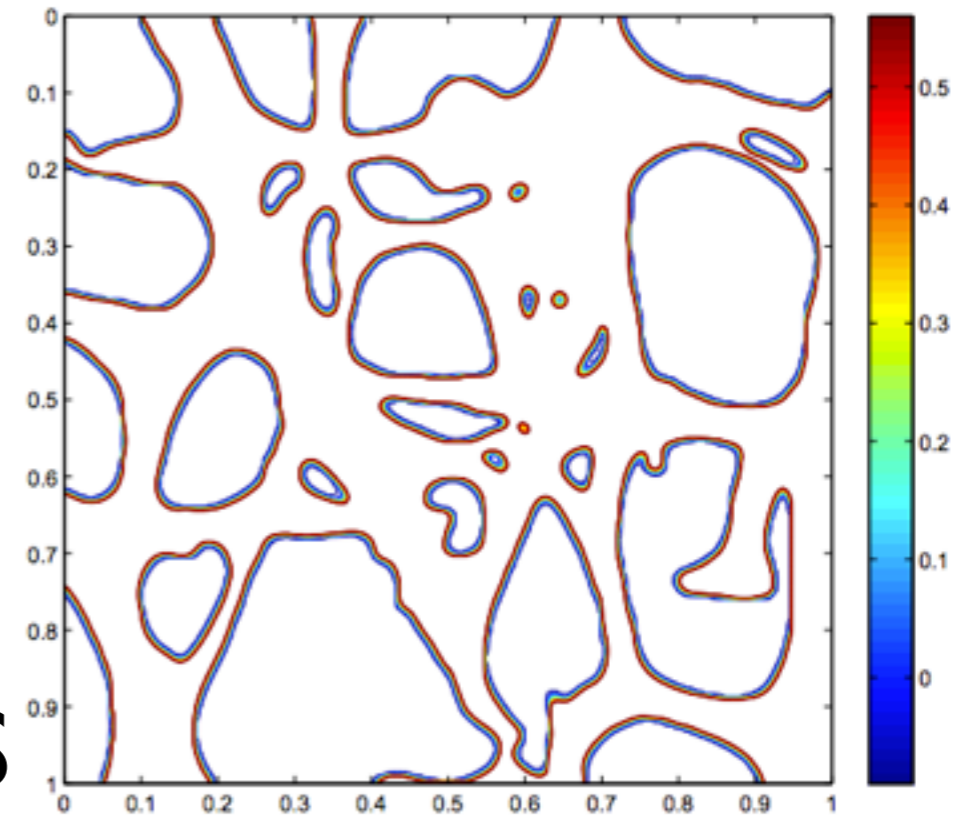
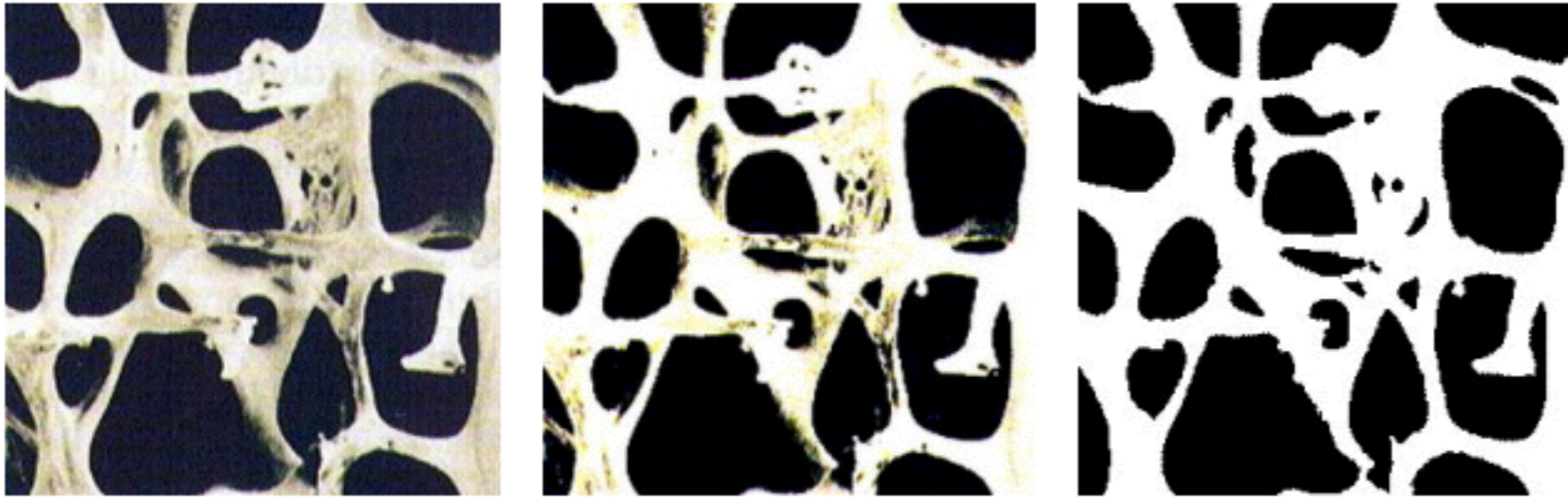
$$\phi = ||\mathbf{x} - \mathbf{x}_c|| - r_c$$

$$\phi(\mathbf{x}) = N_I(\mathbf{x})\phi_I$$

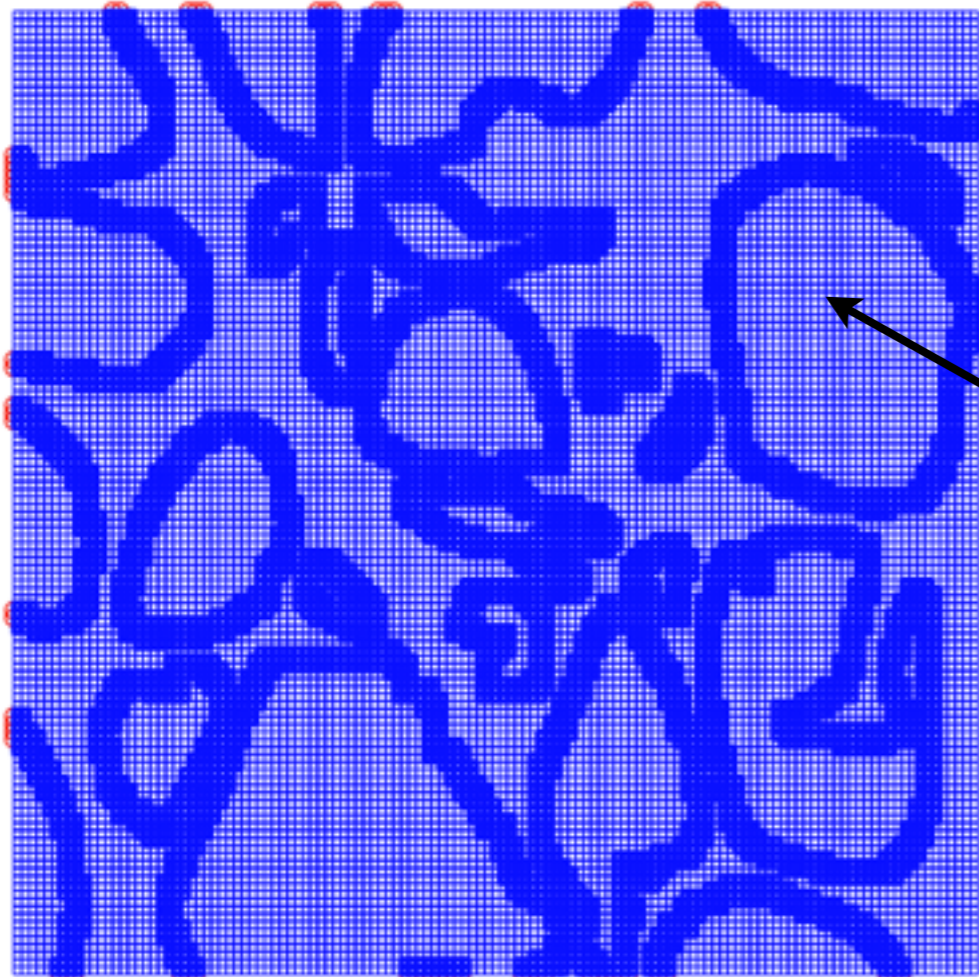


$$\mathbf{u}^h(\mathbf{x}) = \sum_{I \in \mathcal{S}} N_I(\mathbf{x})\mathbf{u}_I + \sum_{J \in \mathcal{S}^c} N_J(\mathbf{x})\Psi(\mathbf{x})\mathbf{a}_J$$

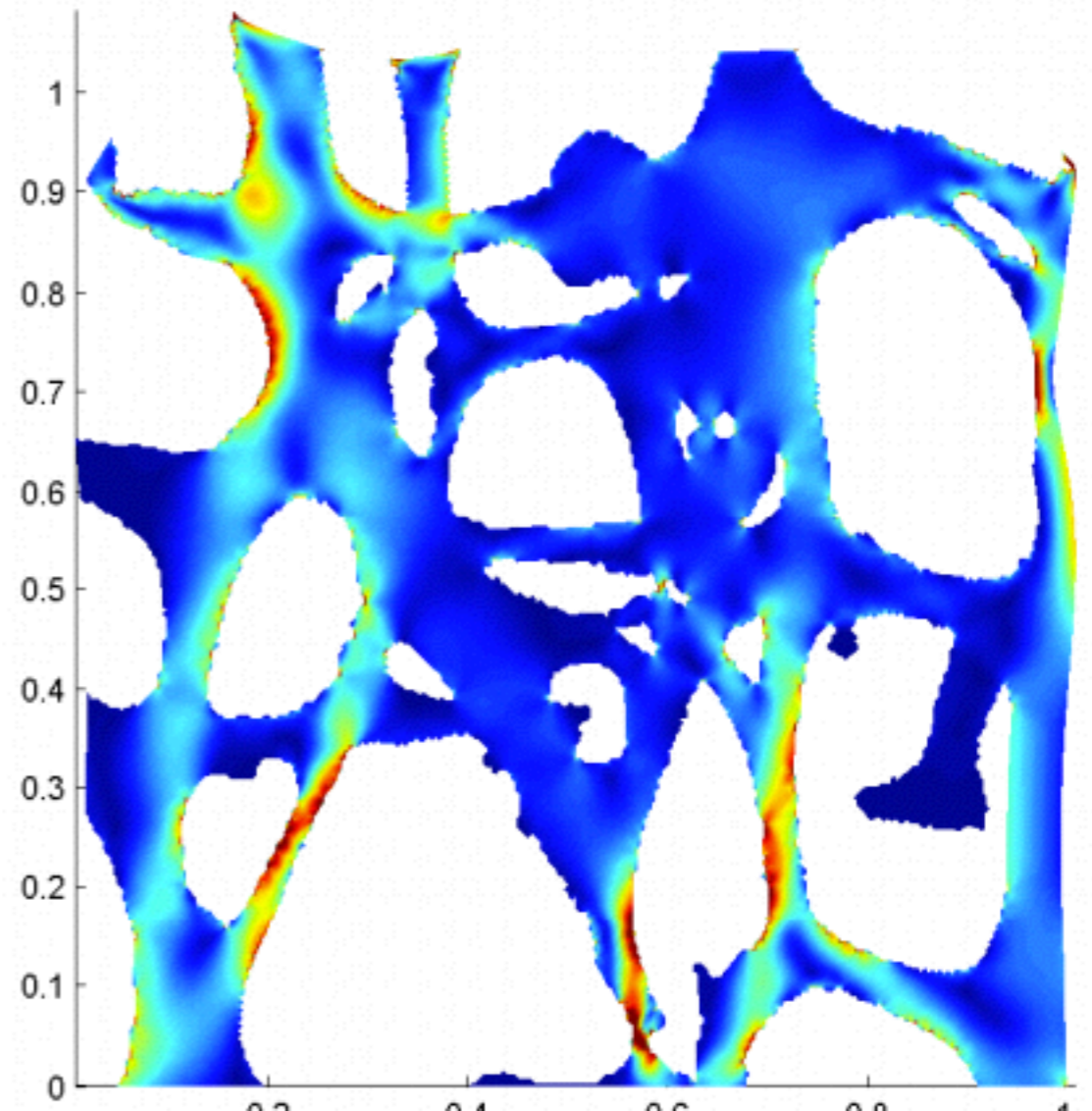
20 $\mathcal{S}^c \quad \phi_{\min} \phi_{\max} < 0$



trabecular bone, PhD thesis, Tran, NUS
holes=extremely soft inclusions



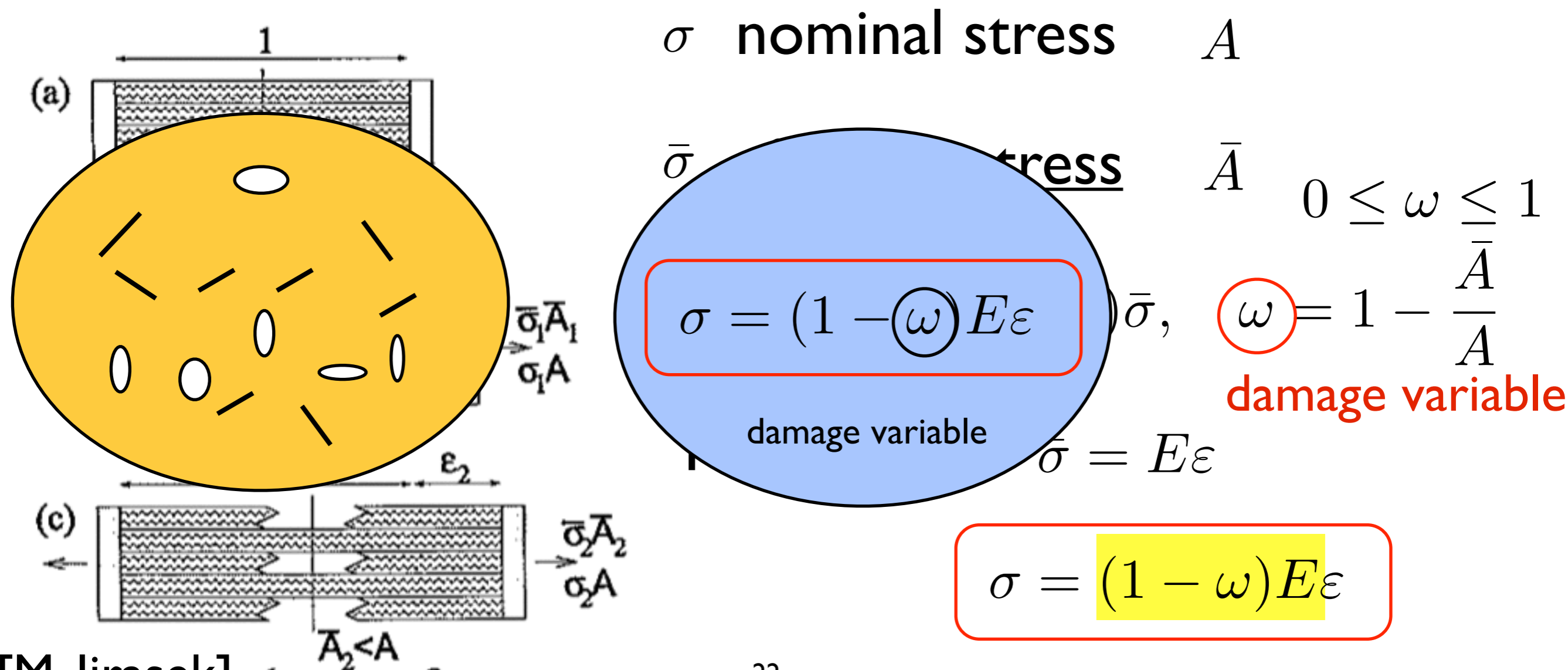
hole



Continuum damage mechanics

Kachanov, 1958, Rabotnov 1969, Hult 1979

CDM is a constitutive theory that describes the progressive loss of material integrity due to the initiation, coalescence and propagation of microcracks, microvoids etc. These changes in the microstructure lead to the degradation of the material stiffness at the macroscale.



Local damage model

Isotropic damage model

\mathbf{C} : elasticity tensor

ϵ_{eq} : equivalent strain [-]

$$\boldsymbol{\sigma} = (1 - \omega) \mathbf{C} \boldsymbol{\epsilon}$$

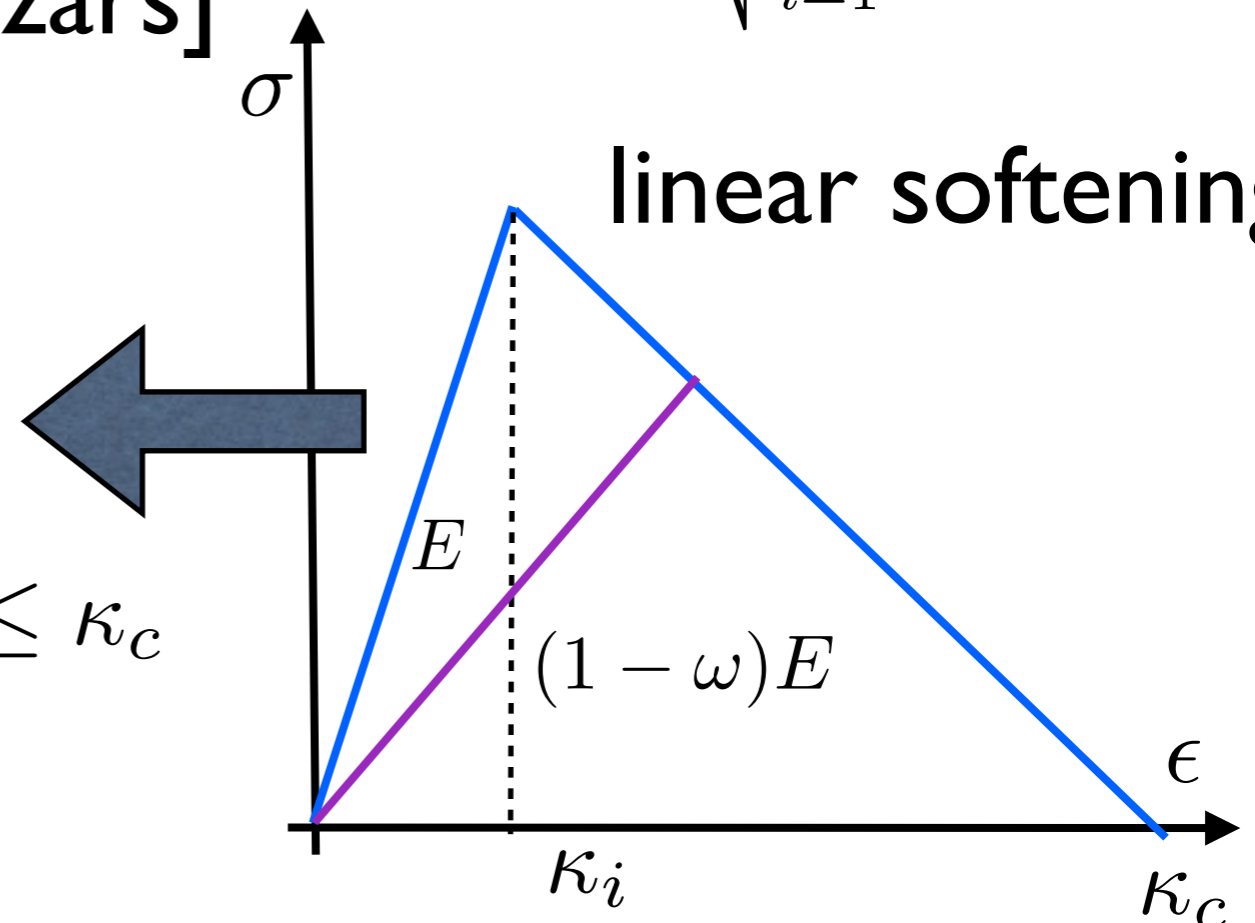
$$\omega = f(\epsilon_{\text{eq}})$$

$$\epsilon_{\text{eq}} = g(\boldsymbol{\epsilon})$$

Tensile failure
[Mazars]

$$\epsilon_{\text{eq}} = \sqrt{\sum_{i=1}^3 \langle \epsilon_i \rangle^2}$$

linear softening



Damage evolution law

$$\omega = \begin{cases} 0 & \text{if } \kappa < \kappa_i \\ 1 - \frac{\kappa_i}{\kappa} \frac{\kappa_c - \kappa}{\kappa_c - \kappa_i} & \text{if } \kappa_i \leq \kappa \leq \kappa_c \\ 1 & \text{if } \kappa > \kappa_c \end{cases}$$

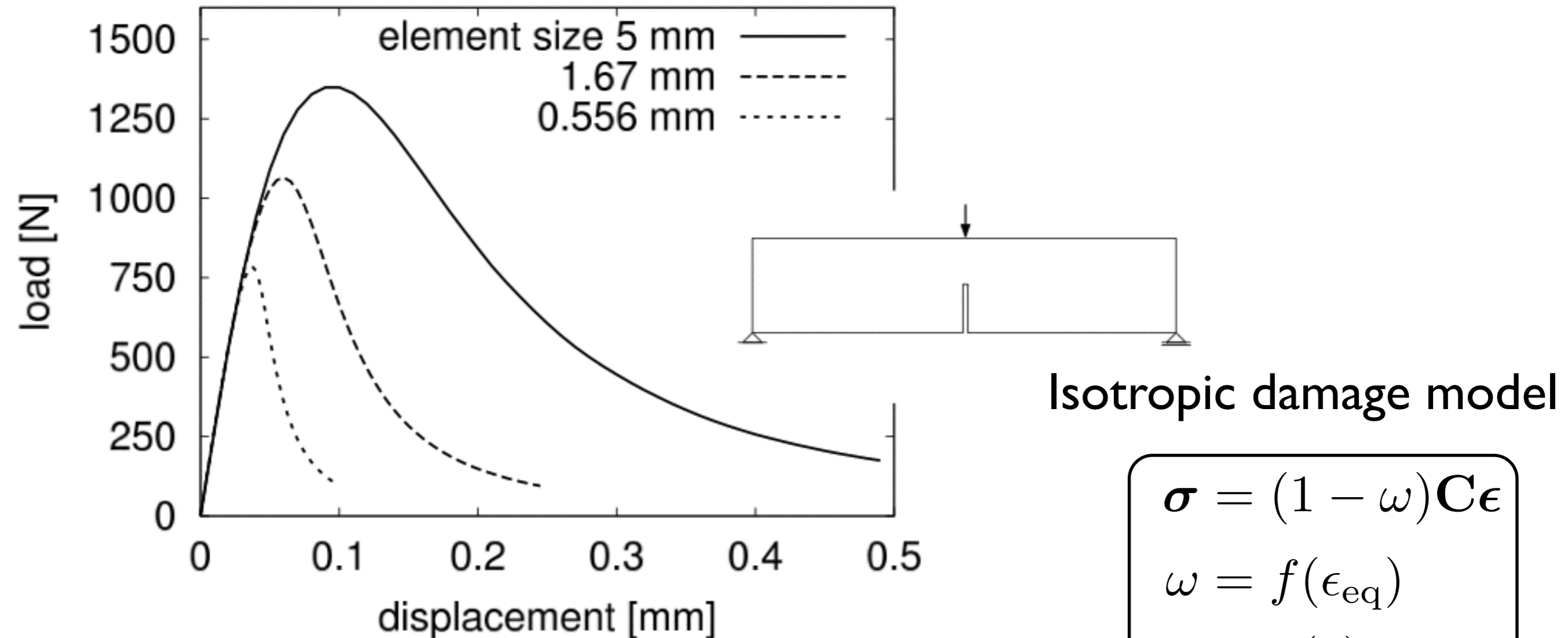
Irreversibility of failure

$$\kappa = \max \epsilon_{\text{eq}}$$

stress update: explicit and simple

Local damage model

In the early 1980s it was found that FE solutions of softening damage do not converge upon mesh refinement, Z. Bazant, 1984.



Isotropic damage model

$$\boldsymbol{\sigma} = (1 - \omega) \mathbf{C} \boldsymbol{\epsilon}$$

$$\omega = f(\epsilon_{\text{eq}})$$

$$\epsilon_{\text{eq}} = g(\boldsymbol{\epsilon})$$

No energy dissipation !!!

Softening plastic models: also suffer from mesh sensitivity.

Nonlocal damage model

Cabot and Bazant, 1987

$$\boldsymbol{\sigma} = (1 - \omega) \mathbf{C} \boldsymbol{\epsilon}$$

$$\omega = f(\bar{\epsilon}_{\text{eq}})$$

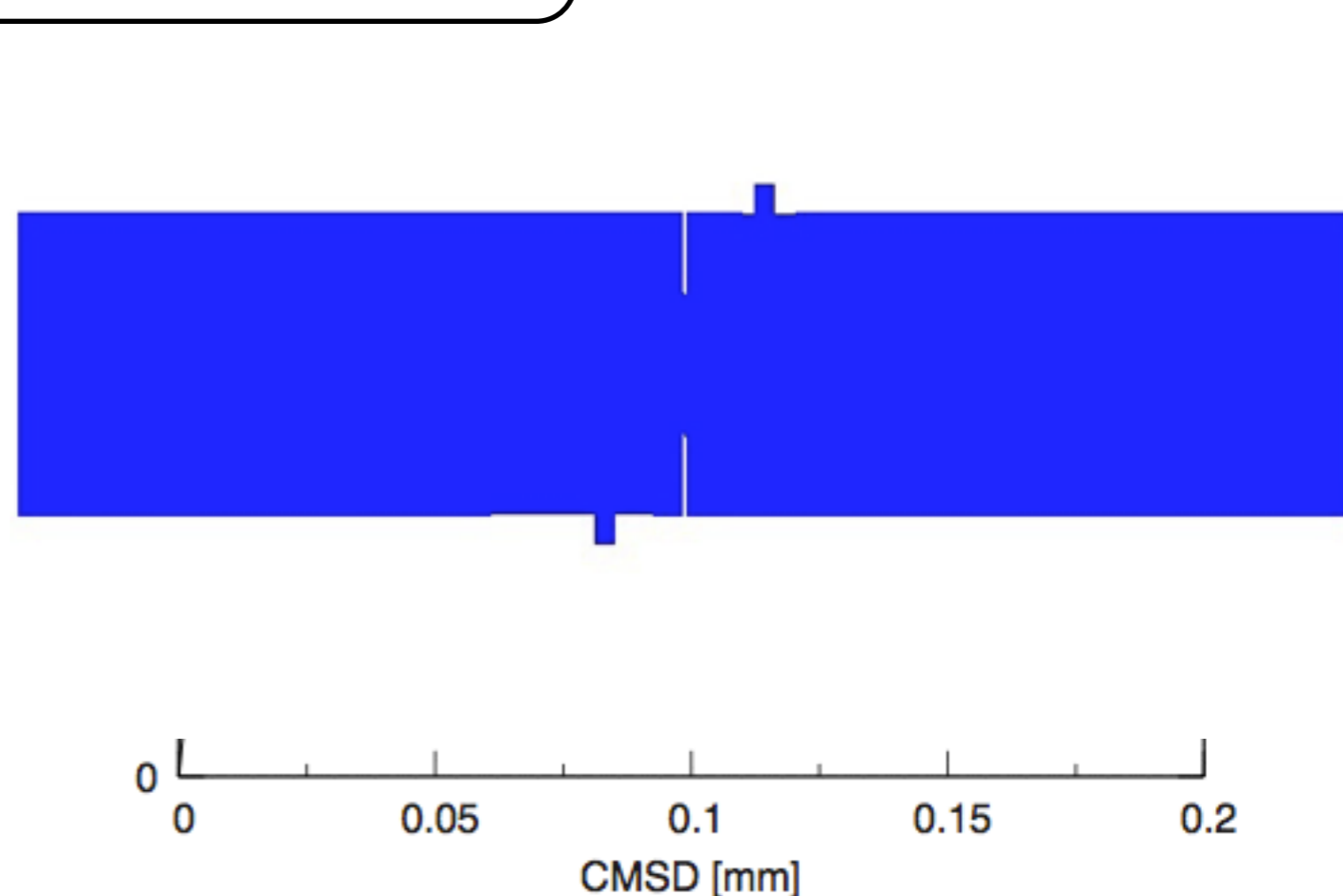
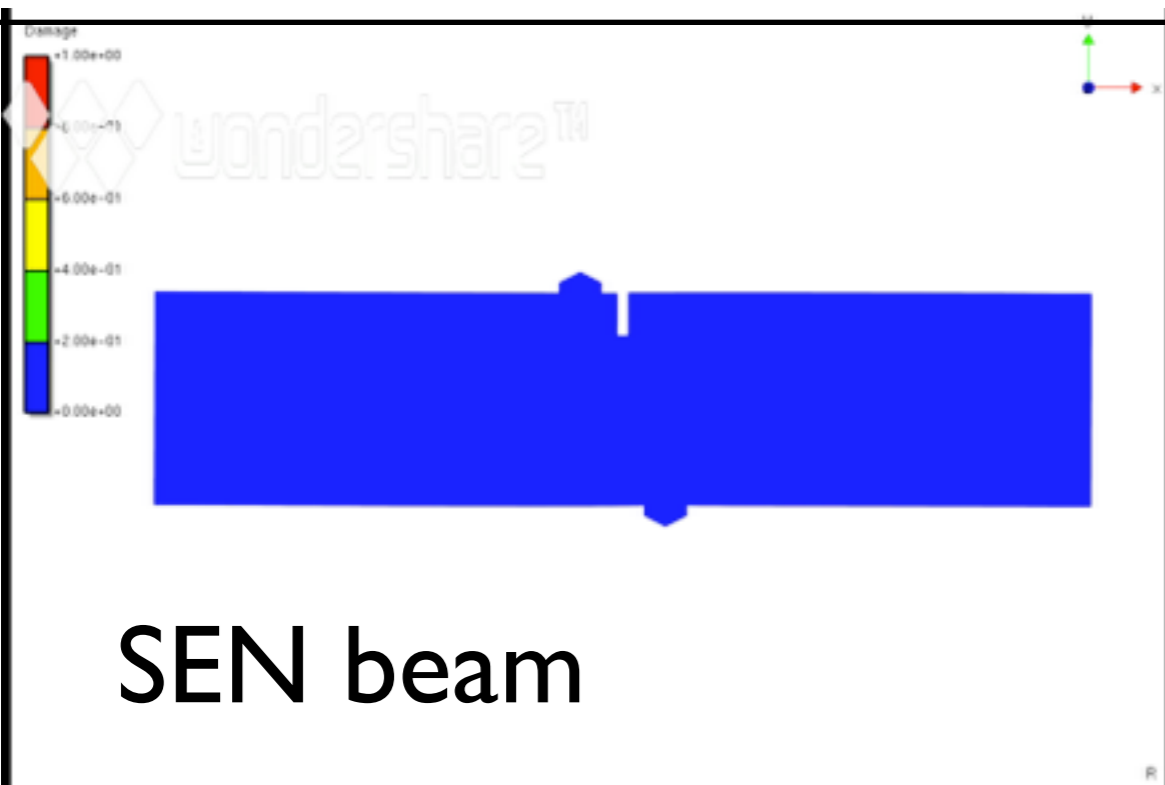
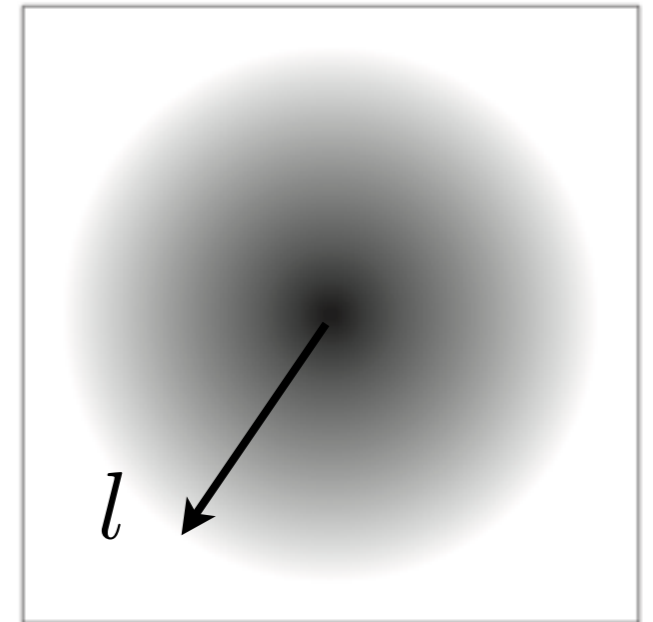
$$0 \leq \omega \leq 1$$

$$\bar{\epsilon}_{\text{eq}}(\mathbf{x}) = \int_{\Omega} \alpha(\mathbf{x} - \boldsymbol{\xi}) \boldsymbol{\epsilon}_{\text{eq}}(\boldsymbol{\xi}) d\boldsymbol{\xi}$$

nonlocal
eqv. strain

$$\alpha(r) = \exp\left(-\frac{r^2}{2l^2}\right)$$

Ⓛ is the length scale



Gradient damage model

$$\boldsymbol{\sigma} = (1 - \omega) \mathbf{C} \boldsymbol{\epsilon}$$

$$\omega = f(\bar{\boldsymbol{\epsilon}}_{\text{eq}}) \quad 0 \leq \omega \leq 1$$

$$\bar{\boldsymbol{\epsilon}}_{\text{eq}}(\mathbf{x}) = \int_{\Omega} \alpha(\mathbf{x} - \boldsymbol{\xi}) \boldsymbol{\epsilon}_{\text{eq}}(\boldsymbol{\xi}) d\boldsymbol{\xi}$$

$$\alpha(r) = \exp\left(-\frac{r^2}{2l^2}\right) \text{secant matrix}$$

$$\boldsymbol{\sigma} = (1 - \omega) \mathbf{C} \boldsymbol{\epsilon}$$

$$\omega = f(\bar{\boldsymbol{\epsilon}}_{\text{eq}})$$

$$\bar{\boldsymbol{\epsilon}}_{\text{eq}} - c \nabla^2 \bar{\boldsymbol{\epsilon}}_{\text{eq}} = \boldsymbol{\epsilon}_{\text{eq}}$$

$$c = \frac{l^2}{2}$$

$$\dot{\boldsymbol{\sigma}} = (1 - \omega) \mathbf{C} \dot{\boldsymbol{\epsilon}} - \mathbf{C} \boldsymbol{\epsilon} \dot{\omega}$$

*Microplane
Damage Models
(Z. Bazant)*

$$\dot{\omega} = \frac{\partial \omega}{\partial \kappa} \frac{\partial \kappa}{\partial \bar{\boldsymbol{\epsilon}}_{\text{eq}}} \dot{\bar{\boldsymbol{\epsilon}}}_{\text{eq}}$$

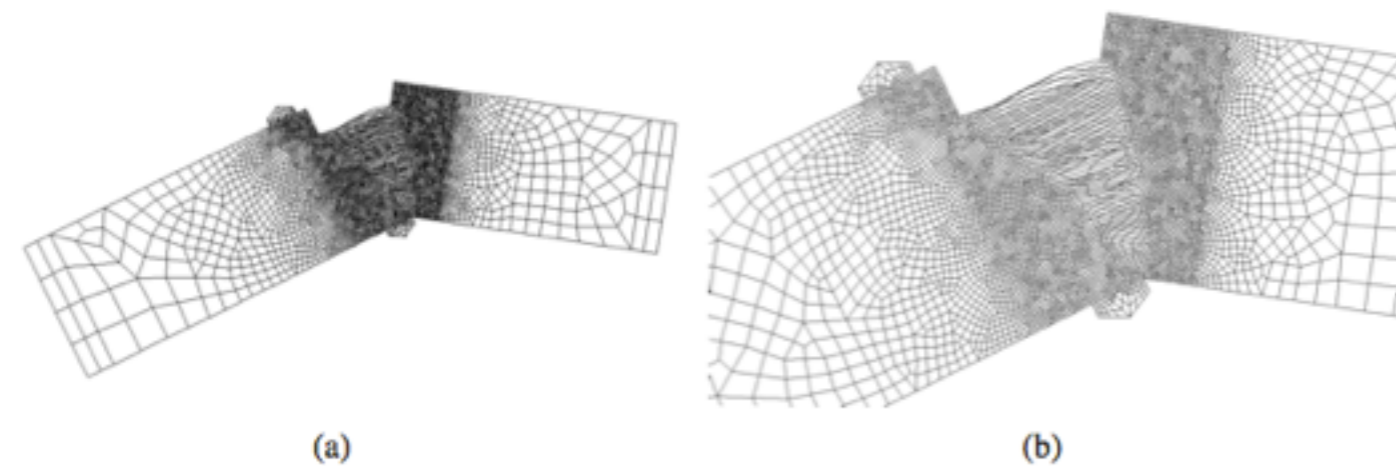
$$= \frac{\partial \omega}{\partial \kappa} \frac{\partial \kappa}{\partial \bar{\boldsymbol{\epsilon}}_{\text{eq}}} \mathbf{N}_{\boldsymbol{\epsilon}} \dot{\boldsymbol{\epsilon}}_{\text{eq}}$$



Implicit GD model
Peerlings et al., 1996

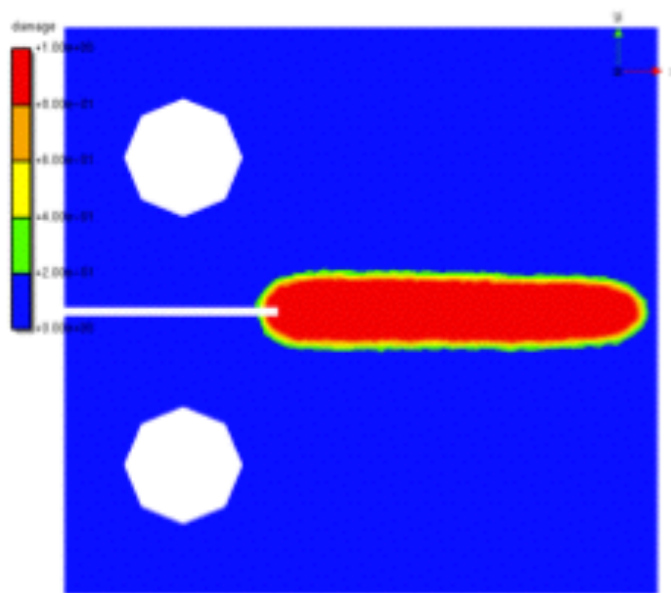
$$\begin{bmatrix} \mathbf{K}_{eu} & \mathbf{K}_{e\epsilon} \end{bmatrix} \begin{Bmatrix} \delta \mathbf{d}^{(i)} \\ \delta \bar{\boldsymbol{\epsilon}}_{\text{eq}}^{(i)} \end{Bmatrix} = \begin{Bmatrix} \mathbf{f}_{\text{int},u} \\ \mathbf{f}_{\text{int},\epsilon} \end{Bmatrix}^{(i-1)} - \begin{Bmatrix} \mathbf{f}_{\text{ext},u} \\ \mathbf{0} \end{Bmatrix}^{(i-1)} \quad \begin{array}{l} \text{displacements} \\ \text{ivalent strain} \end{array}$$

Continuous vs. discontinuous description



- + easy to implement (2D/3D)
- + one single constitutive law
- + standard elements
- incorrect final stage of failure
- evolving length scale

- hard to implement (3D)
- two separate constitutive laws
- enriched elements
- + correct final stage of failure

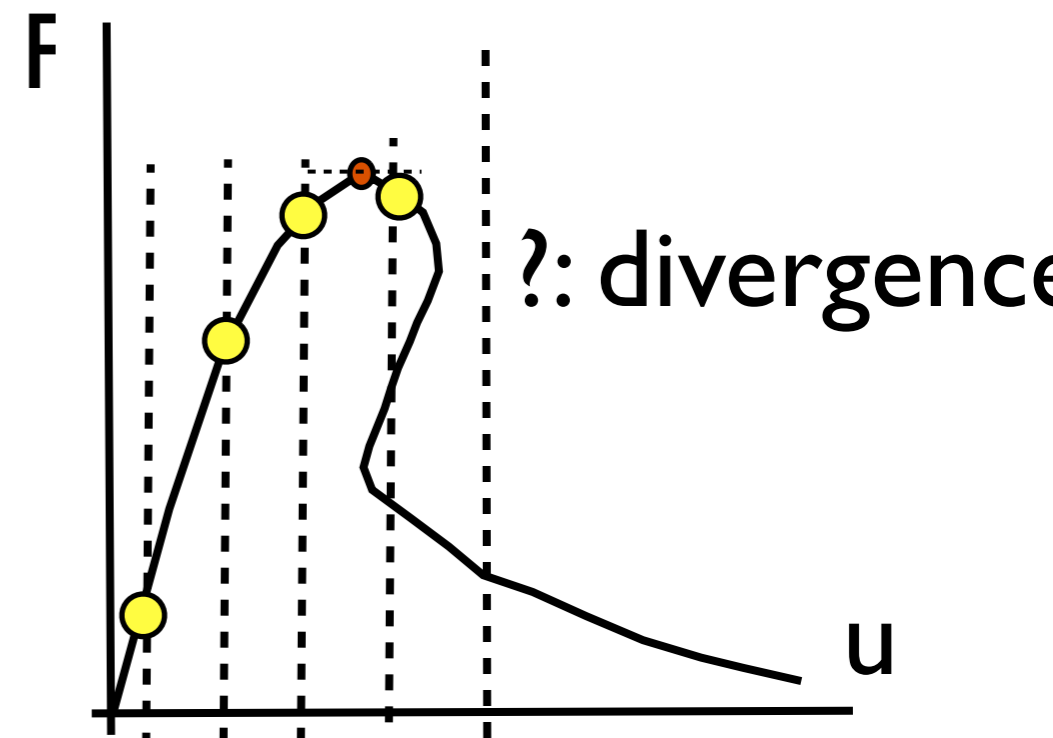
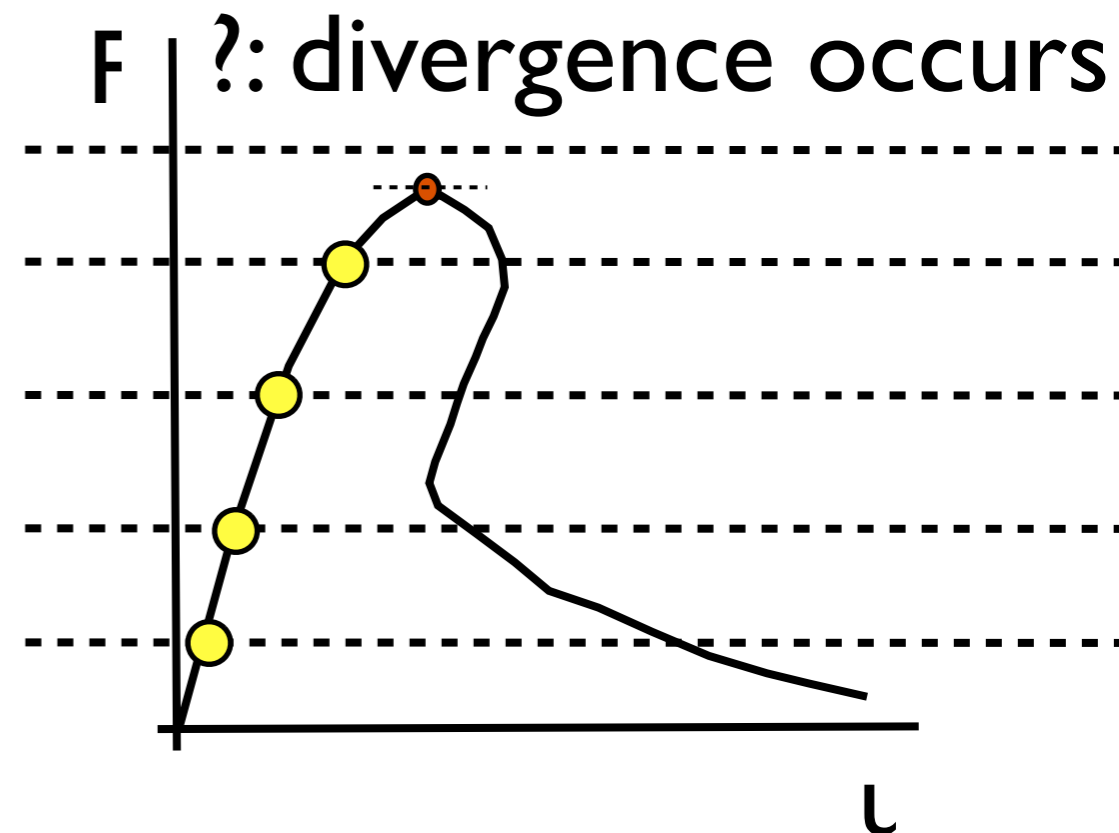
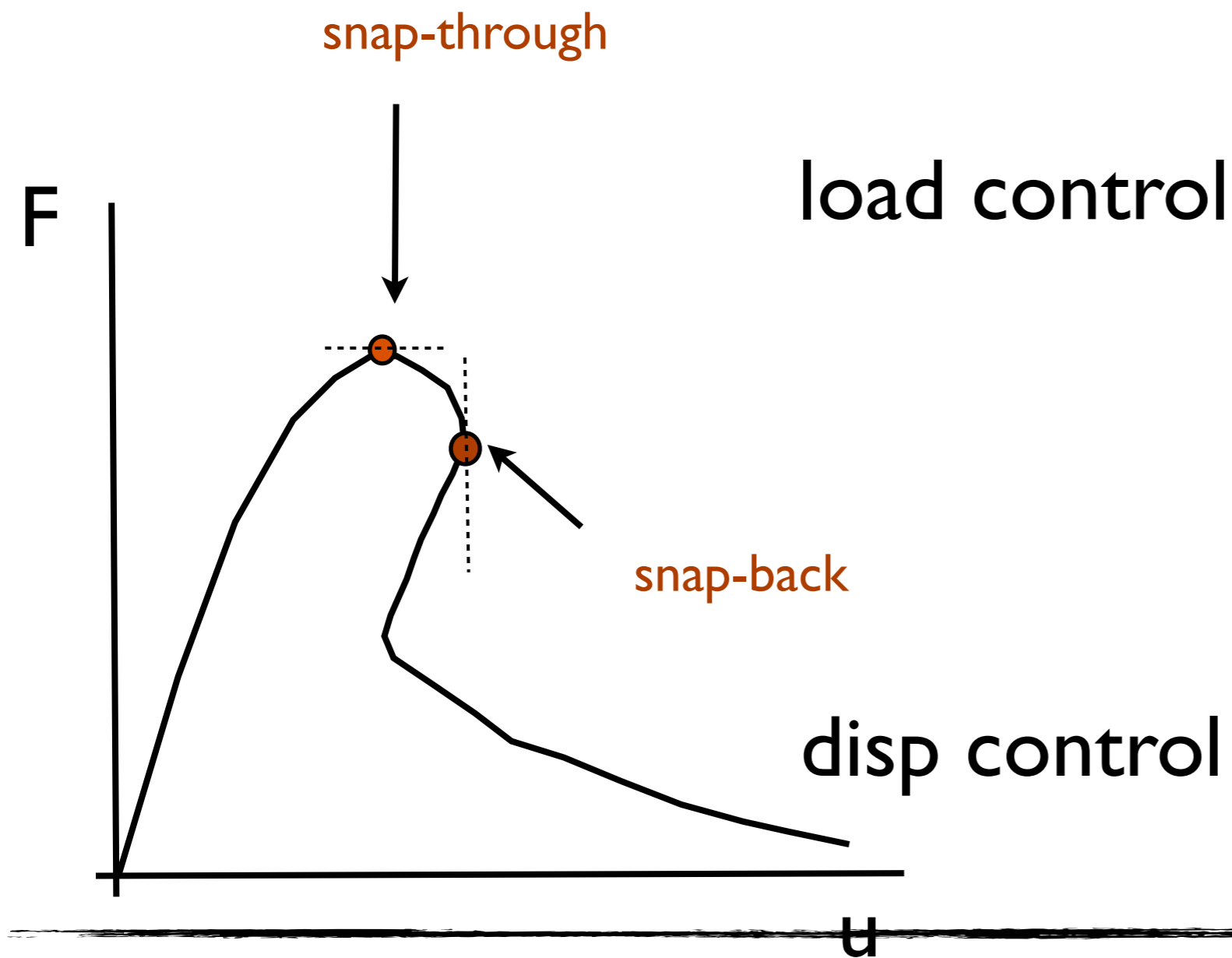


bests of both worlds: combined continuous-discontinuous approaches

(Dr. Nguyen Dinh Giang, Univ. Sydney)

Solution strategies

For a quasi-static analysis of softening solids, one encounters cases...



Incremental-iterative procedure

Path-following methods

Riks 1972

$$\begin{bmatrix} \mathbf{f}^{\text{int}}(\mathbf{u}) - \lambda \mathbf{g} \\ \phi(\mathbf{u}, \lambda) \end{bmatrix} = 0$$

$$\mathbf{f}^{\text{ext}} = \lambda \mathbf{g}$$

λ load factor

reference load vector

Newton-Raphson $\phi(\mathbf{u}, \lambda)$ arc-length/constraint function

$$\begin{bmatrix} \mathbf{f}^{\text{int}}(\mathbf{u}_{(k)}) - \lambda_{(k)} \mathbf{g} \\ \phi(\mathbf{u}_{(k)}, \lambda_{(k)}) \end{bmatrix} + \begin{bmatrix} \mathbf{K} & -\mathbf{g} \\ \mathbf{v}^{\text{T}} & w \end{bmatrix}^{(k)} \cdot \begin{bmatrix} \Delta \mathbf{u} \\ \Delta \lambda \end{bmatrix} = 0$$

where

$$\mathbf{K} = \frac{\partial \mathbf{f}^{\text{int}}}{\partial \mathbf{u}}, \quad \mathbf{v} = \frac{\partial \phi}{\partial \mathbf{u}}, \quad w = \frac{\partial \phi}{\partial \lambda}$$

$$\longrightarrow \begin{bmatrix} \Delta \mathbf{u} \\ \Delta \lambda \end{bmatrix} = \begin{bmatrix} \mathbf{u}_I \\ 0 \end{bmatrix} - \frac{\mathbf{v}^{\text{T}} \mathbf{u}_I + \phi}{\mathbf{v}^{\text{T}} \mathbf{u}_{II} + w} \begin{bmatrix} \mathbf{u}_{II} \\ 1 \end{bmatrix}$$

correction $\begin{bmatrix} \mathbf{u} \\ \lambda \end{bmatrix}^{(k+1)} = \begin{bmatrix} \mathbf{u} \\ \lambda \end{bmatrix}^{(k)} + \begin{bmatrix} \Delta \mathbf{u} \\ \Delta \lambda \end{bmatrix}$ $\mathbf{u}_I = \mathbf{K}^{-1} \mathbf{r}, \quad \mathbf{u}_{II} = \mathbf{K}^{-1} \mathbf{g}$

Energy control

Gutierrez 2004

$$\epsilon = \mathbf{B}\mathbf{a}$$

$$\mathbf{f}^{\text{int}} = \int_{\Omega} \mathbf{B}^T \boldsymbol{\sigma}$$

equilibrium

$$V = \frac{1}{2} \int_{\Omega} \boldsymbol{\epsilon}^T \boldsymbol{\sigma} = \frac{1}{2} \int_{\Omega} \mathbf{a}^T \mathbf{B}^T \boldsymbol{\sigma} = \frac{1}{2} \mathbf{a}^T \mathbf{f}^{\text{int}} = \frac{1}{2} \lambda \mathbf{a}^T \mathbf{g}$$

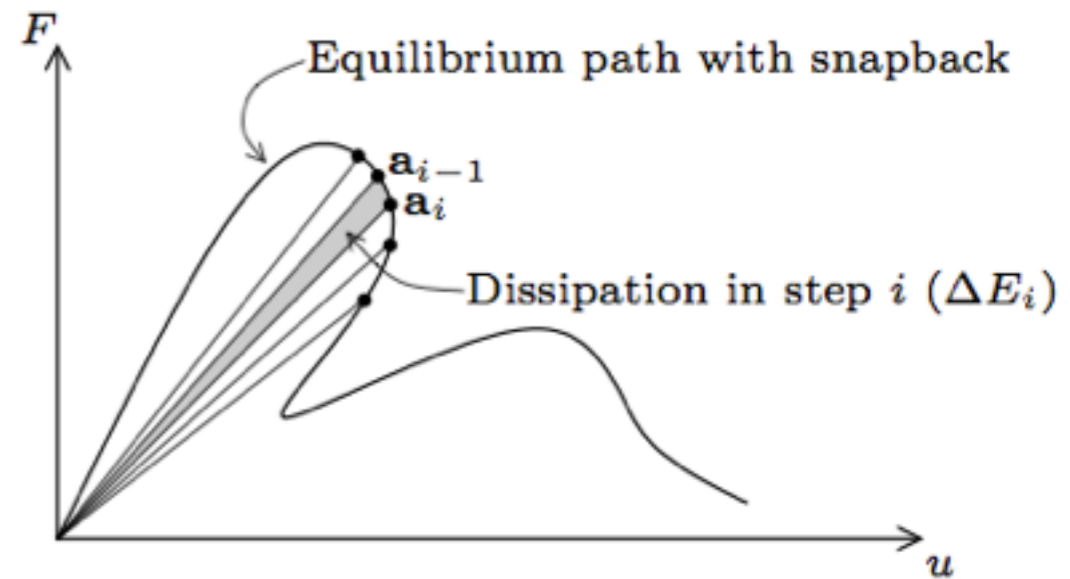
$$\dot{V} = \frac{1}{2} \lambda \dot{\mathbf{a}}^T \mathbf{g} + \frac{1}{2} \dot{\lambda} \mathbf{a}^T \mathbf{g}$$

Energy release rate

$$\dot{V} - \lambda \dot{\mathbf{a}}^T \mathbf{g}$$

$$G = \frac{1}{2} \lambda \dot{\mathbf{a}}^T \mathbf{g} - \frac{1}{2} \dot{\lambda} \mathbf{a}^T \mathbf{g}$$

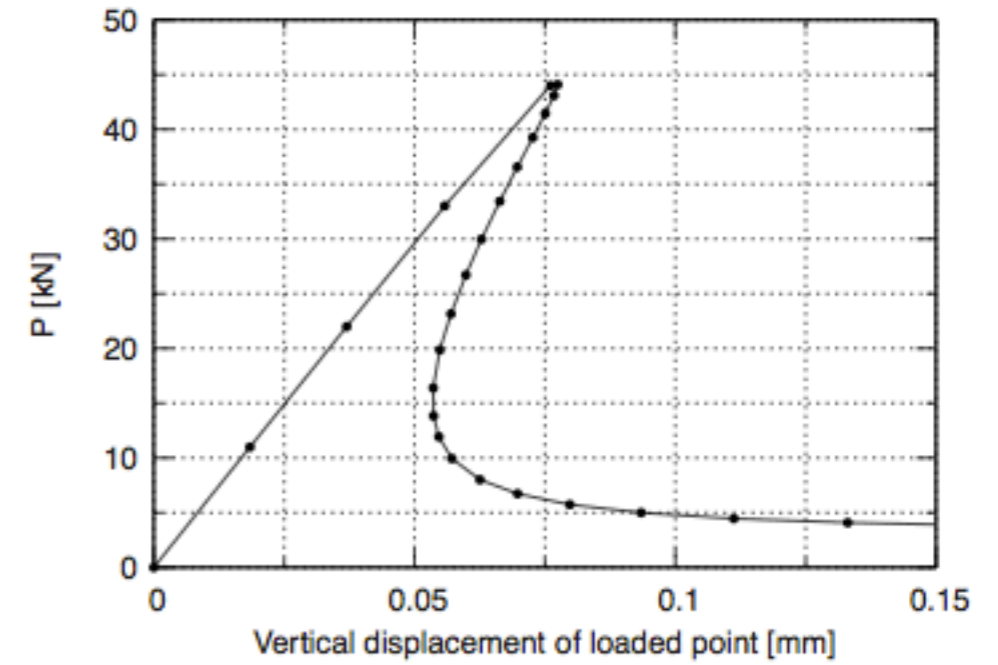
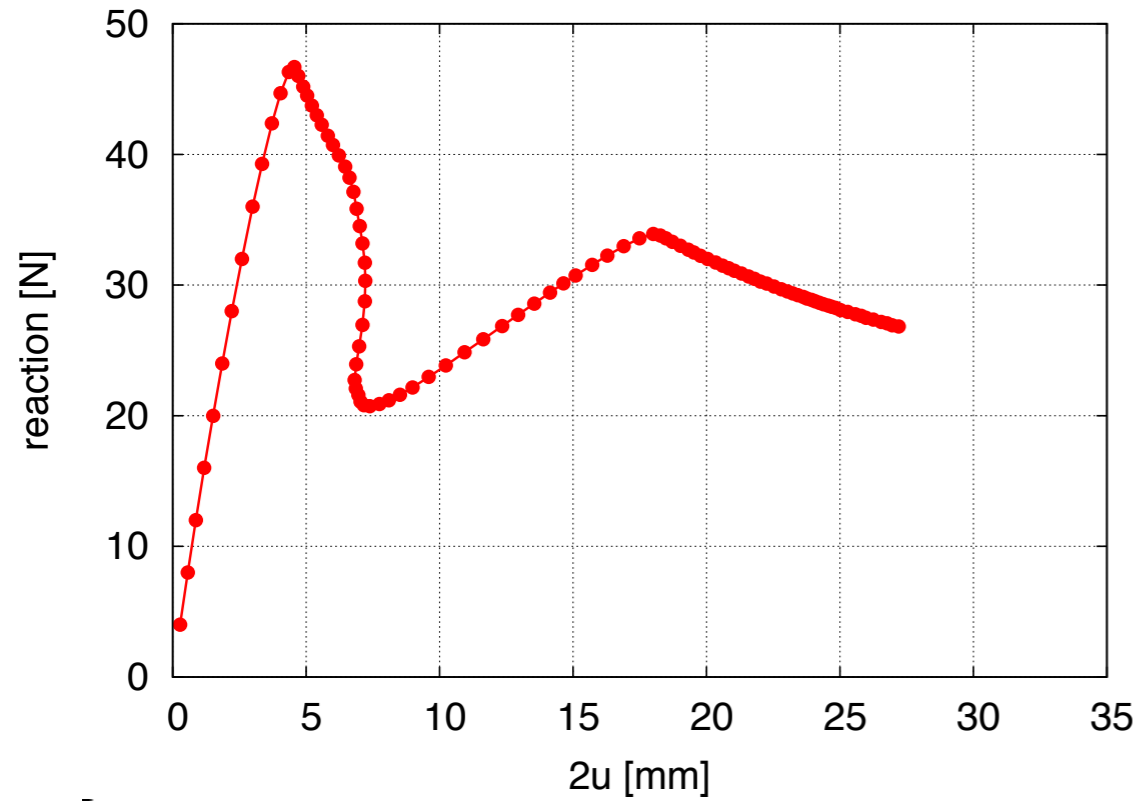
$$G > 0$$



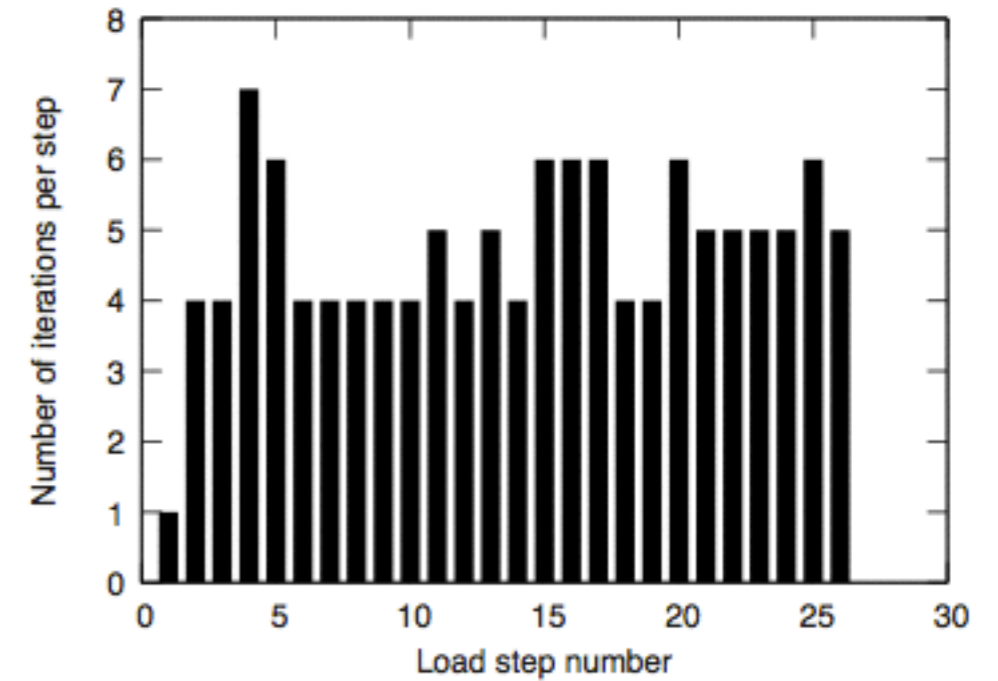
Arc-length function

$$\phi = \frac{1}{2} \left[\lambda^{(n)} (\mathbf{a}_{(n+1)}^T - \mathbf{a}_{(n)}^T) - \Delta \lambda^{(n)} \mathbf{a}_{(n)}^T \right] \mathbf{g} - \Delta \tau = 0$$

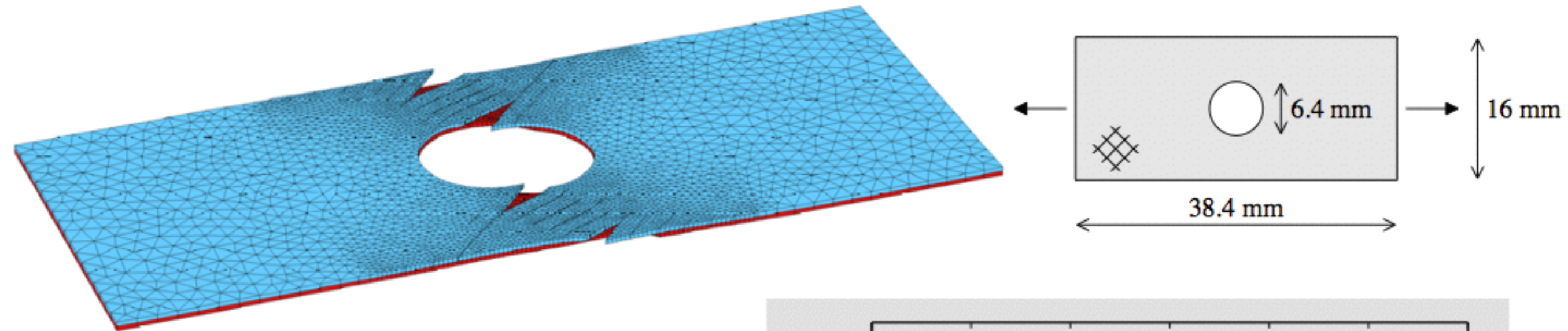
Energy based arc-length control



(a)

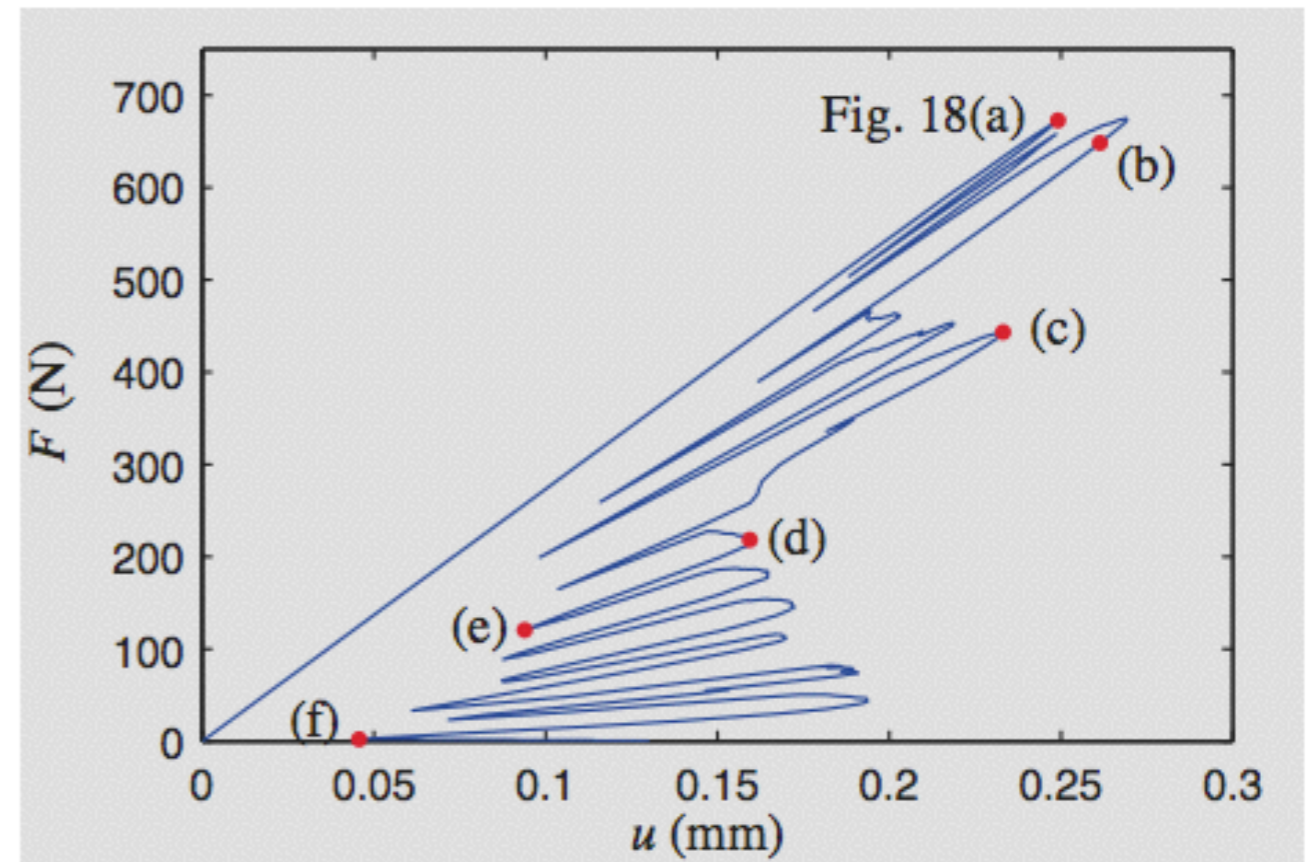


... and here we are
complex failure mechanism in composites $[\pm 45^\circ]$



- matrix cracking
- delamination of plies

F.P. van der Mer
EFM, 2008



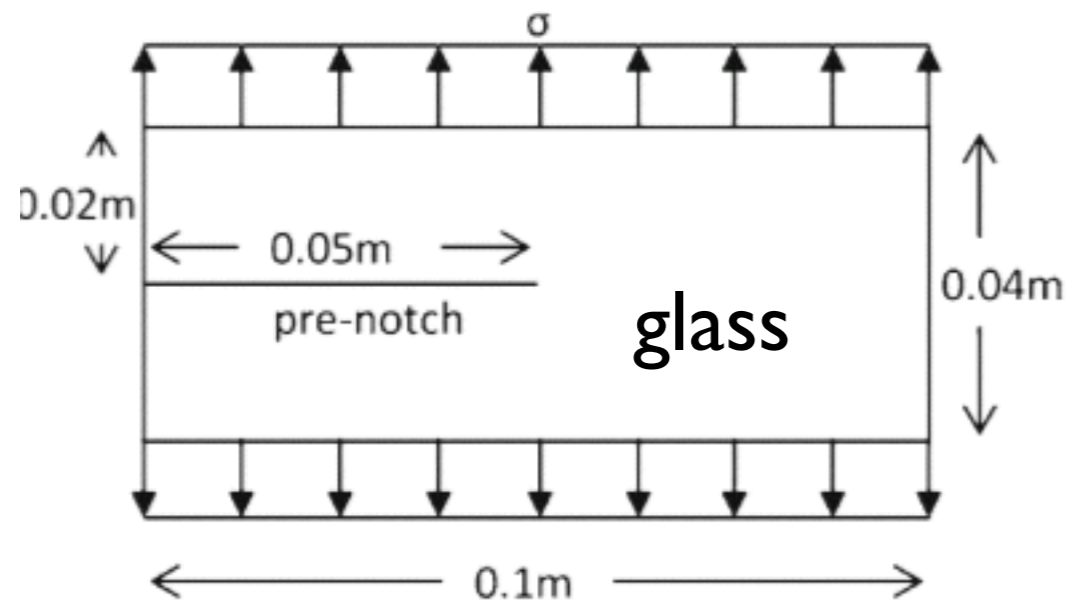
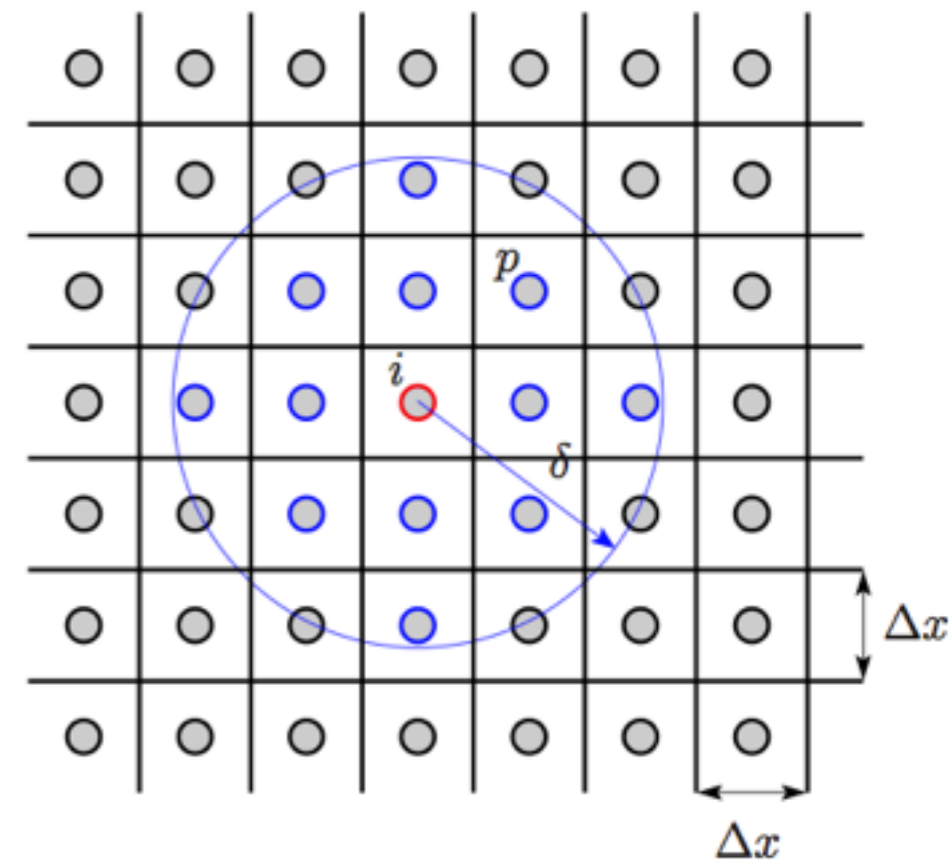
Peridynamics

S. Silling 2000

$$\rho_i \ddot{\mathbf{u}}_i^n = \sum_{p \in \mathcal{F}_i} \mathbf{f}(\mathbf{u}_p^n - \mathbf{u}_i^n, \mathbf{x}_p - \mathbf{x}_i) V_p - \mathbf{b}_i^n \equiv \tilde{\mathbf{f}}_i^n$$

time integration: Verlet integration

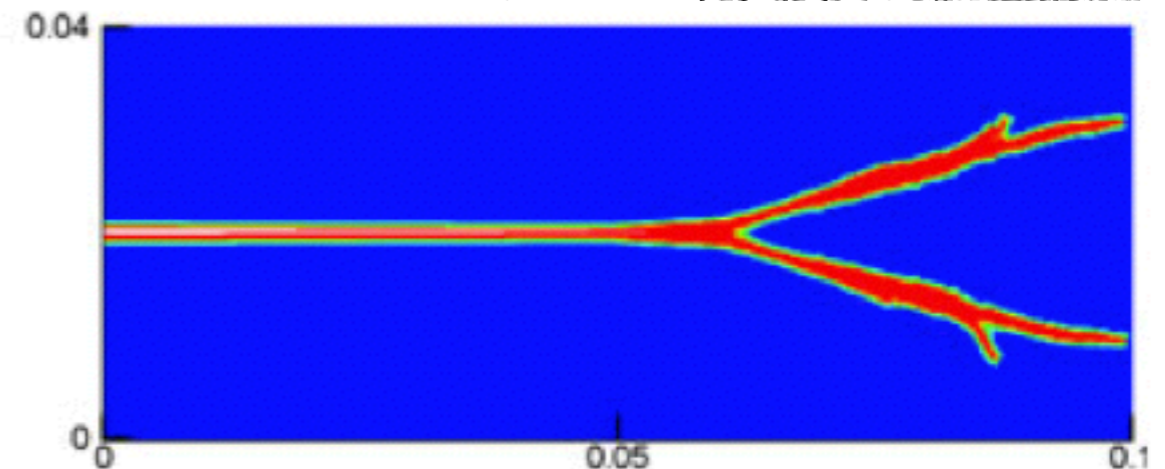
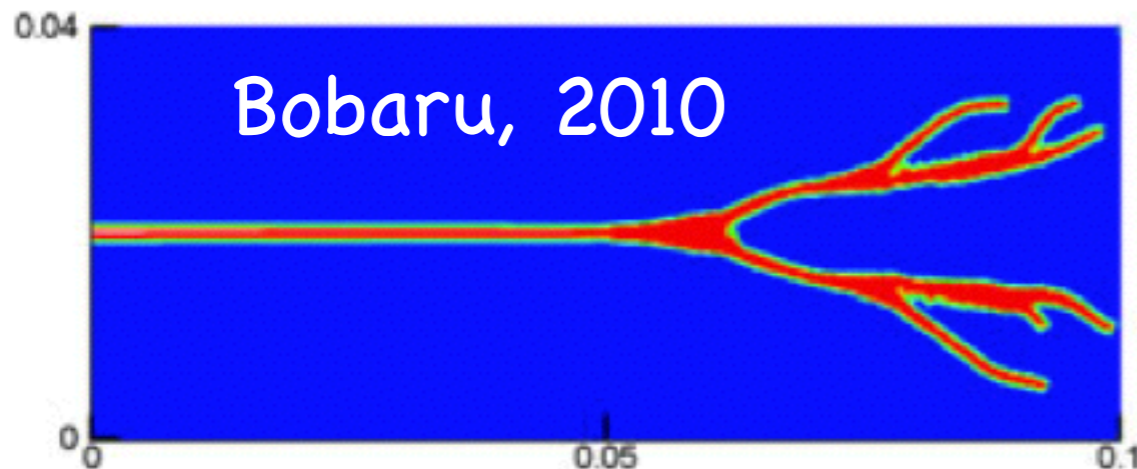
continuum version of MD (molecular dynamics)



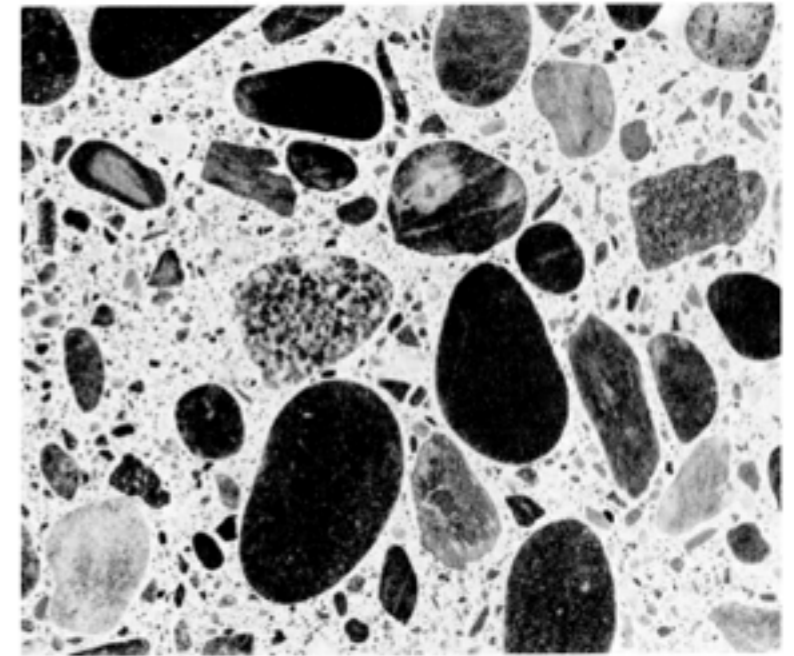
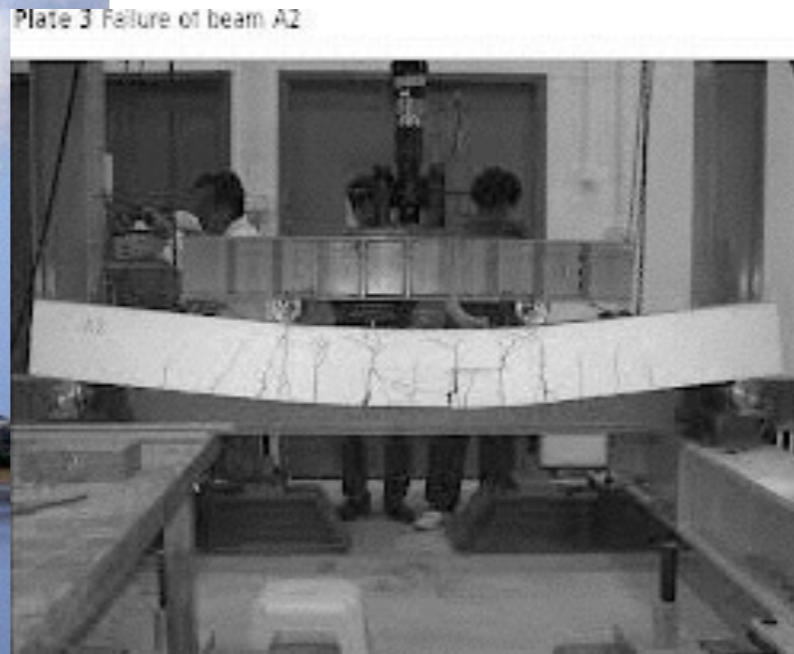
XFEM can do this as well but would require genius programmers (2D)

See phase field models for similar capacities

crack branching



Multiscale methods



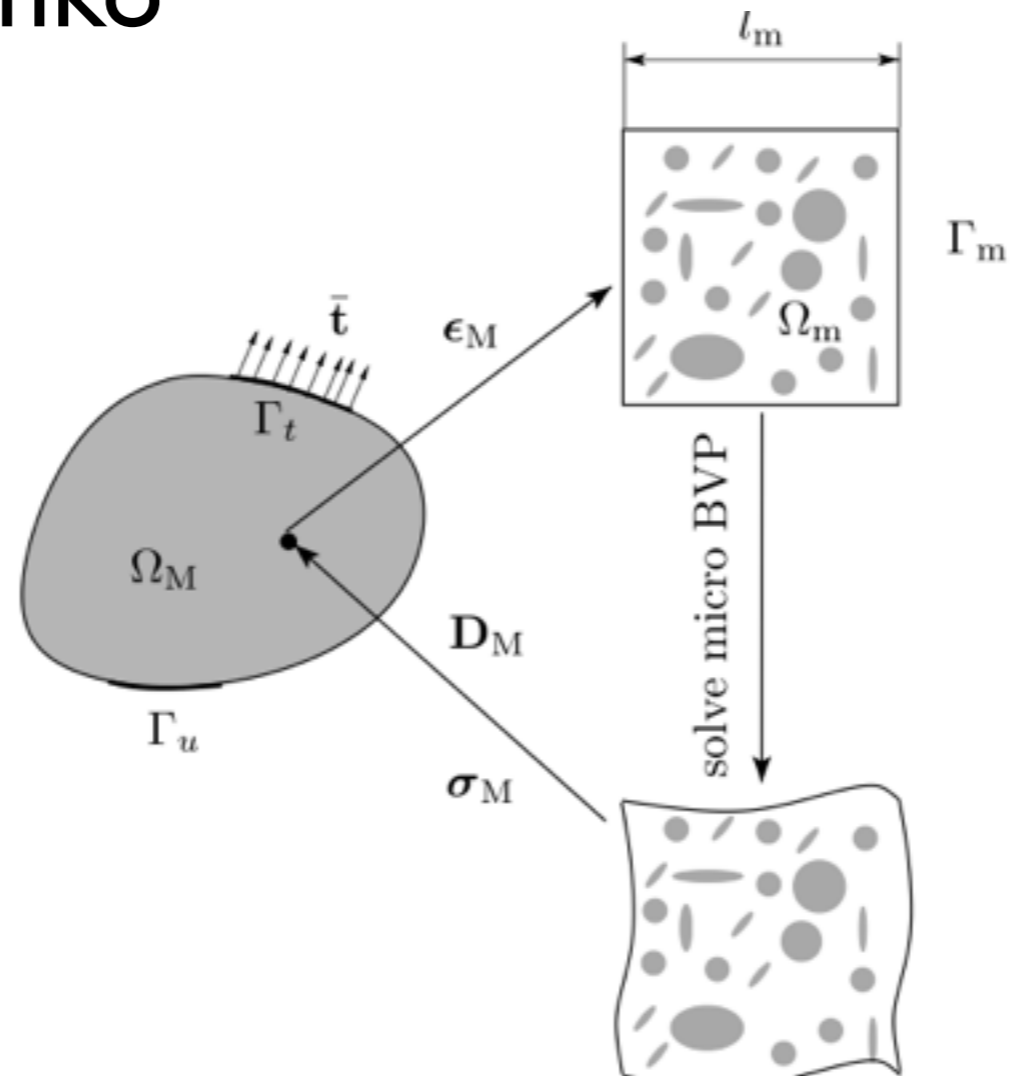
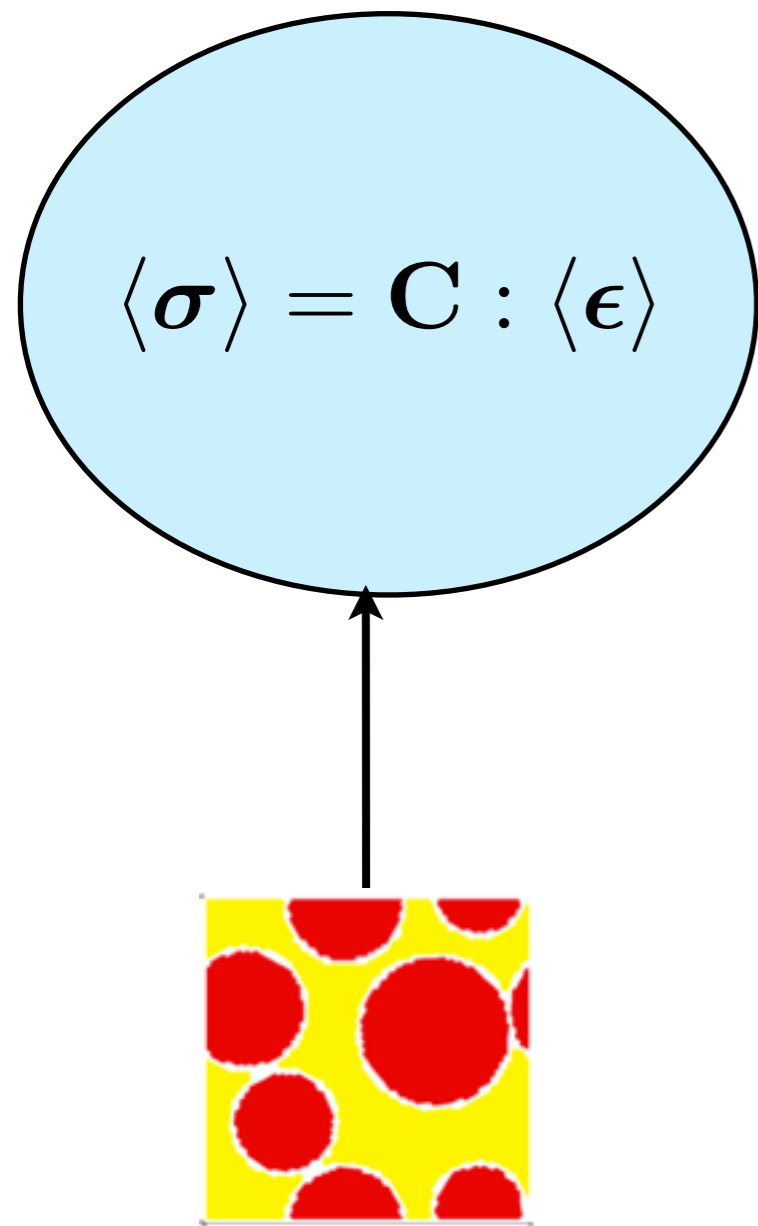
multiple length scales

Multiscale models:

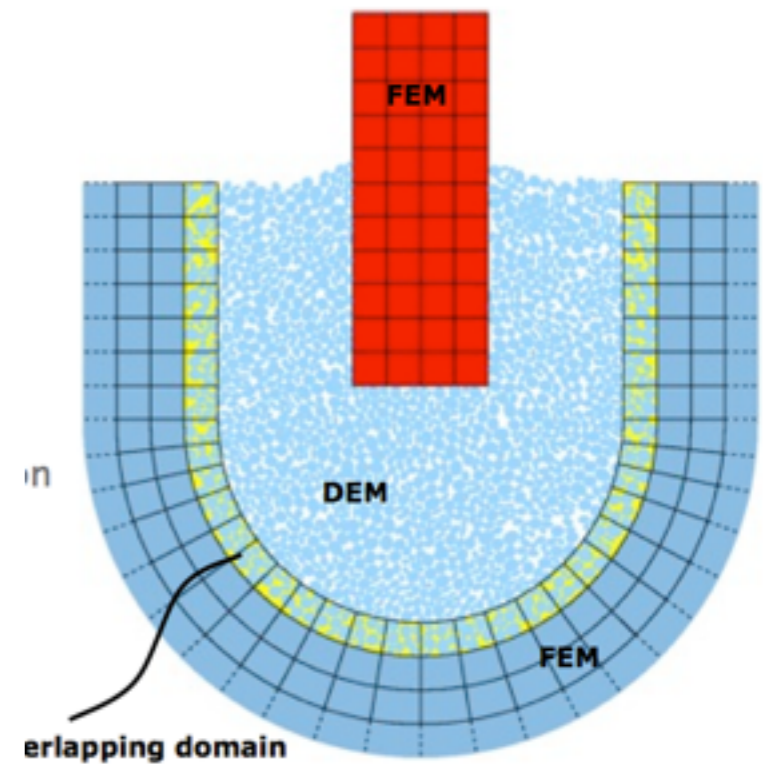
- better constitutive models
- design new materials

Classification

After Ted Belytschko



ARLEQUIN method



pile installation
Wriggers, 2011

hierarchical methods

semi-concurrent

concurrent method

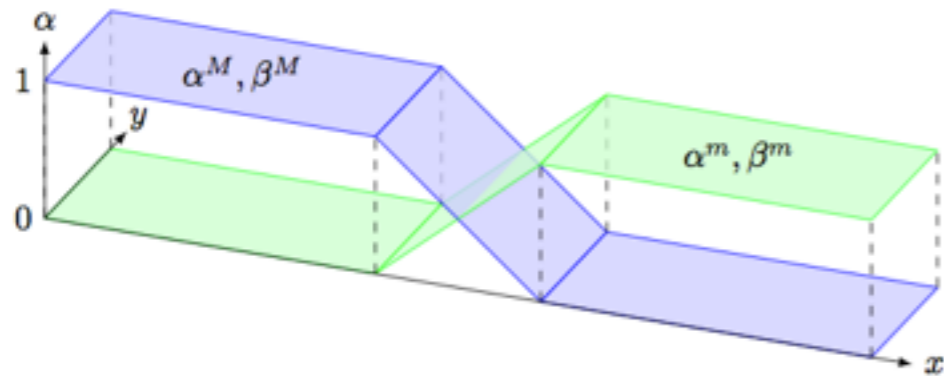
Arlequin method

[H. Ben Dhia, 1998]

- partition of unity for energy in gluing zone

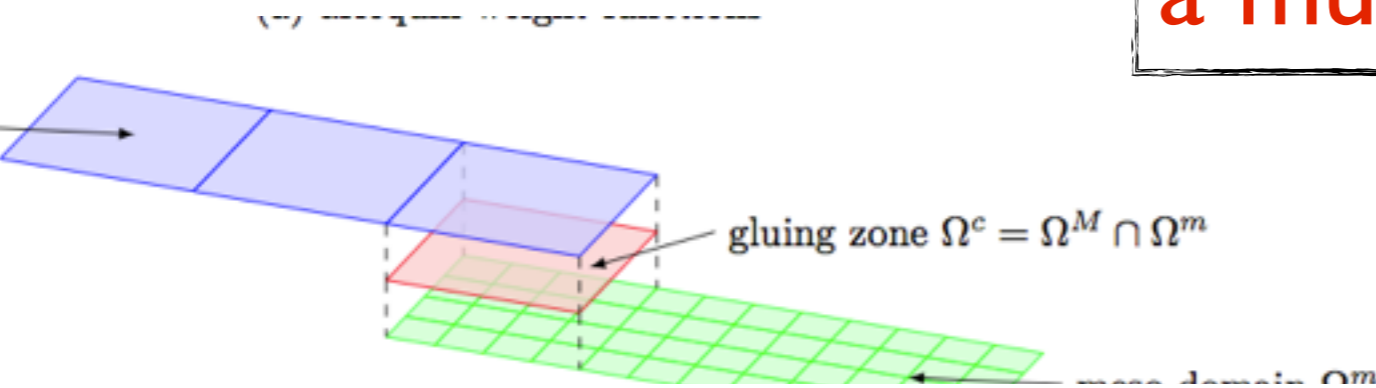
$$\alpha^M + \alpha^m = 1$$

- Lagrange multipliers to glue two models

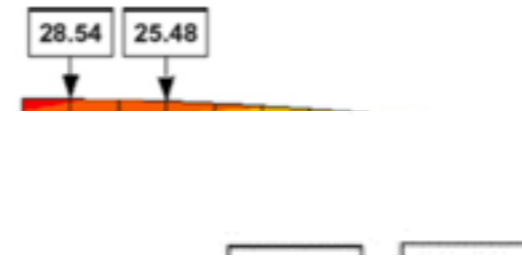


(a) arlequin weight functions

a multi-scale/multi-model method



(b) finite element models



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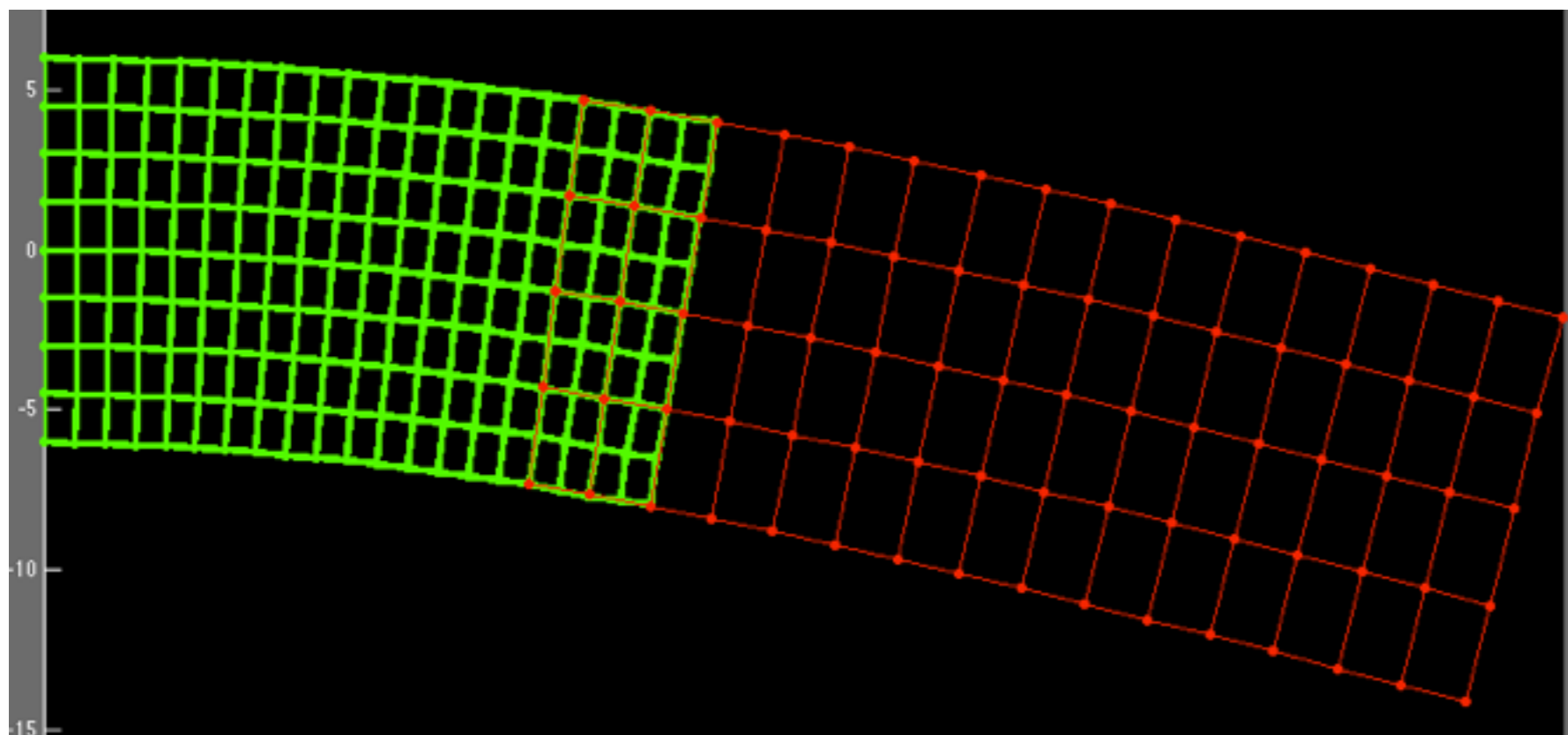
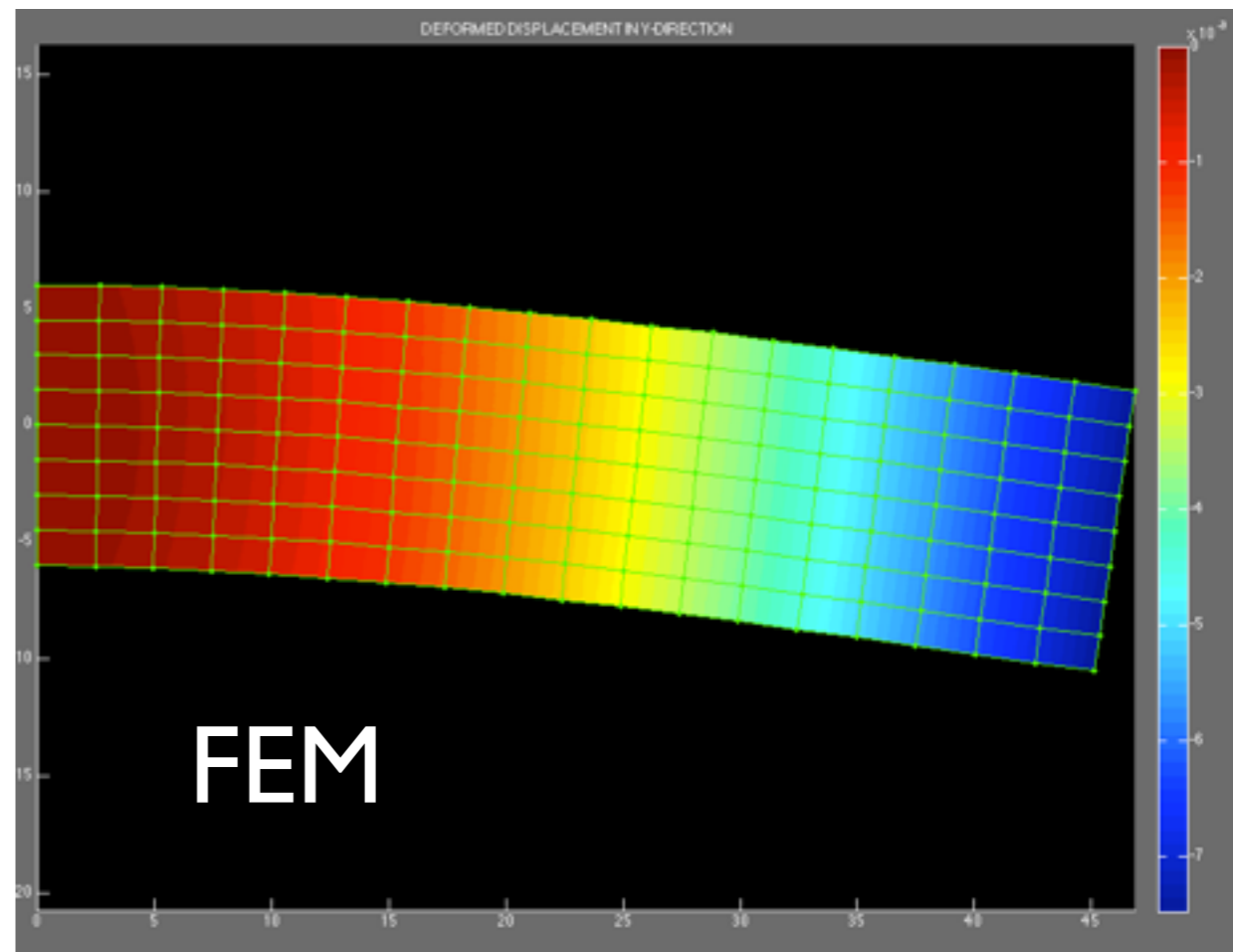
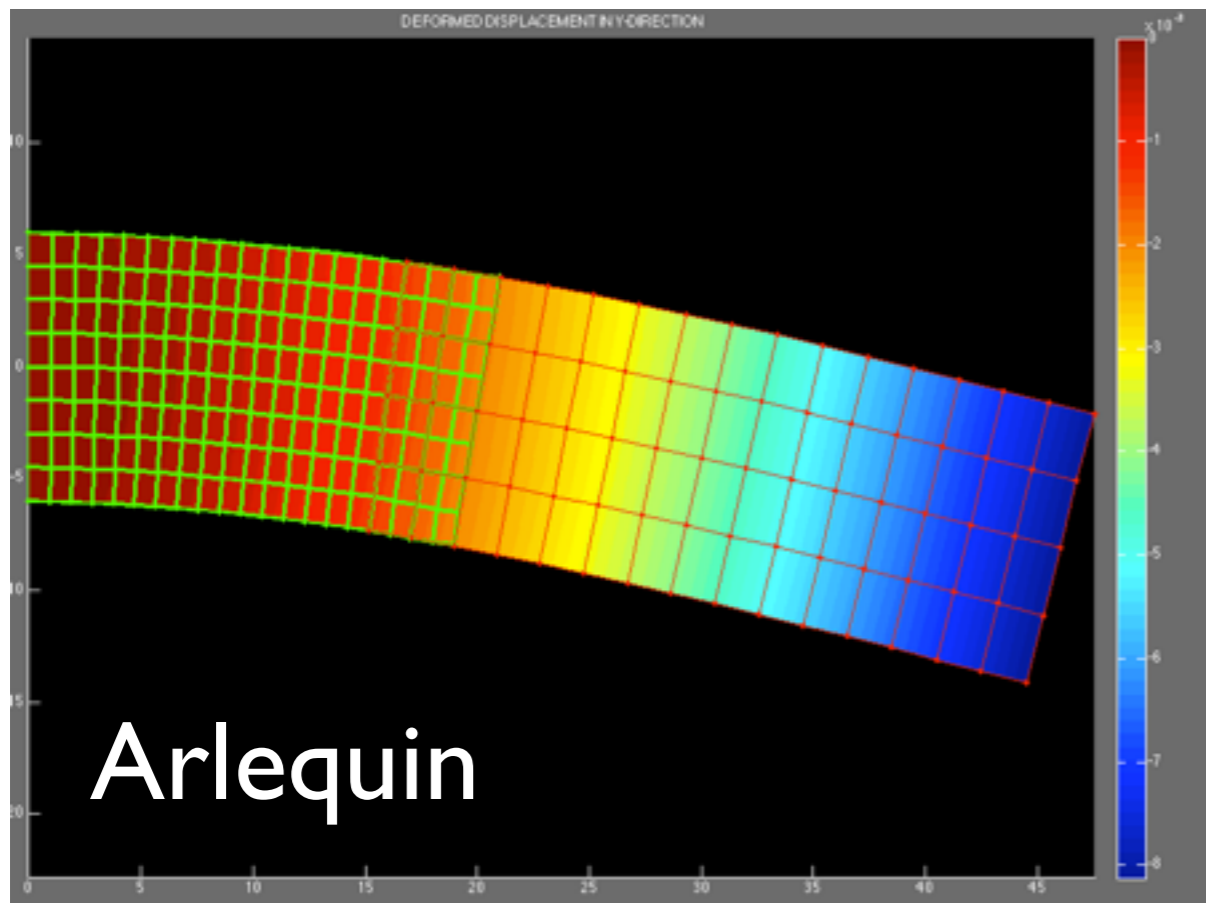
journal homepage: www.elsevier.com/locate/advengsoft



Implementation of the Arlequin method into **ABAQUS**:
Basic formulations and applications

H. Qiao^a, Q.D. Yang^b, W.Q. Chen^c, C.Z. Zhang^{d,*}

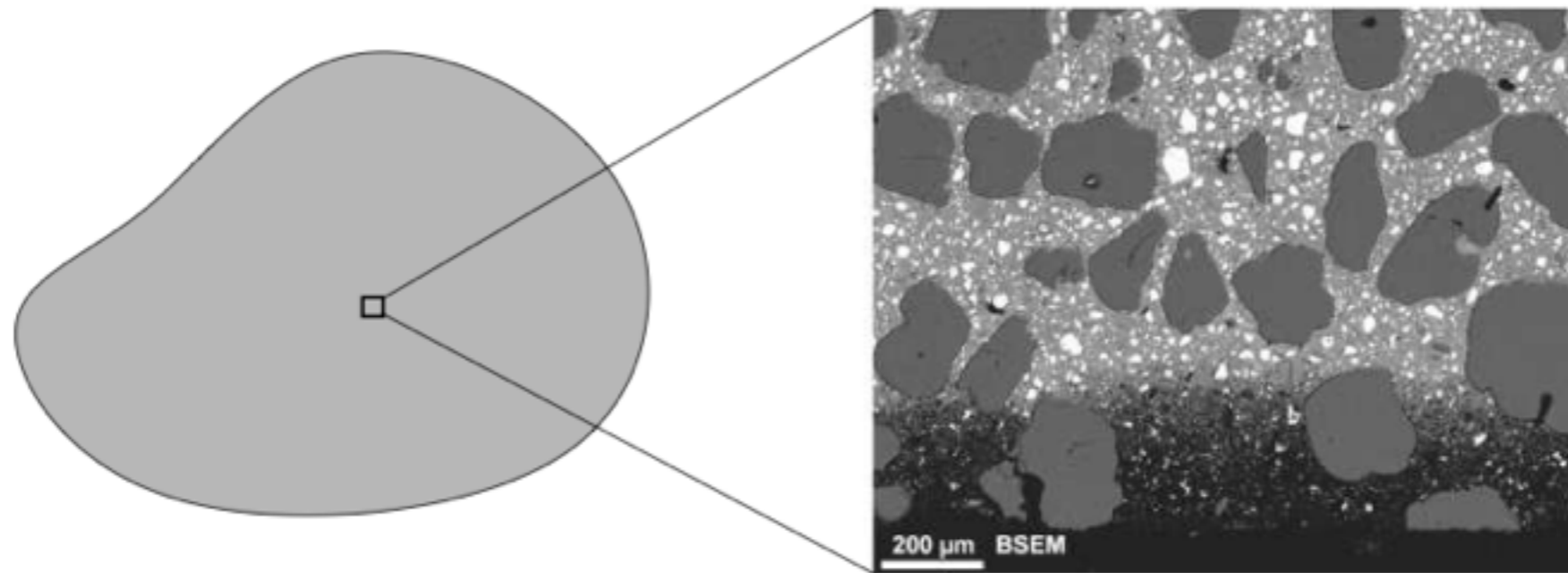
Mortar Method



work in progress

Heterogeneous materials

macroscopically homogeneous but microscopically heterogeneous



macroscopic length scale

microscopic length scale

macroscopic behavior depends on

phenomenological constitutive models $\sigma = f(\epsilon, \alpha)$

two many params

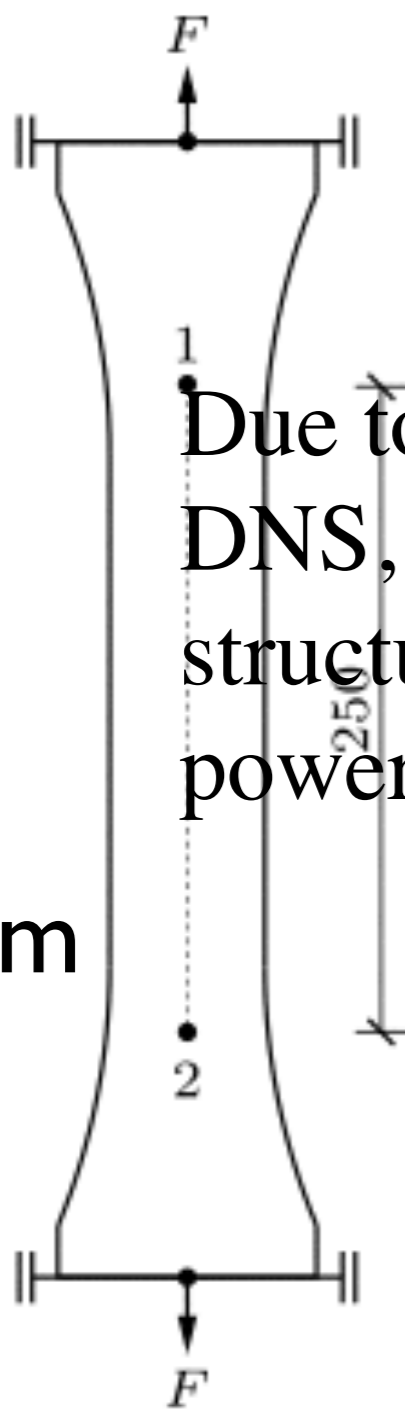
38

- size, shape
- spatial distribution
- volume fraction
- mechanical properties of the constituents.

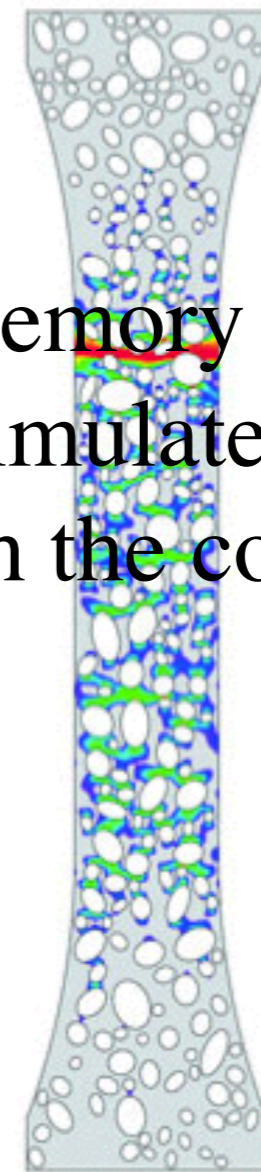
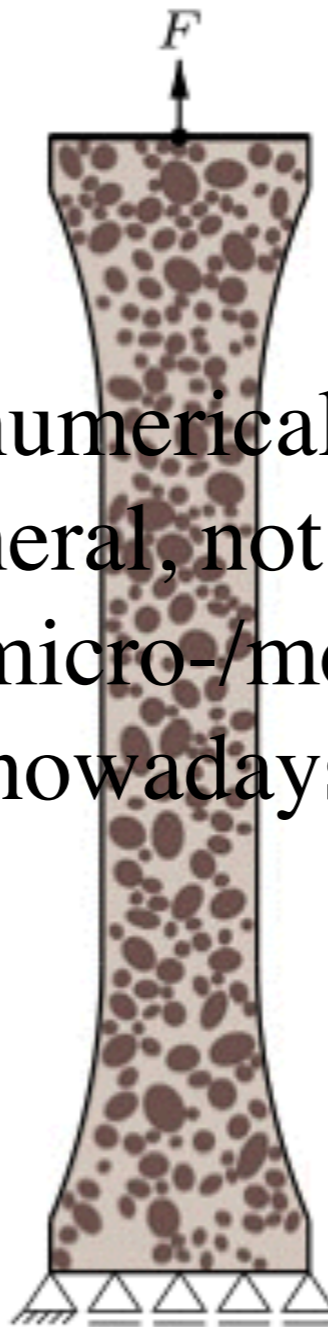
the identification of these parameters is generally difficult

Direct Numerical Simulation (DNS)

Unger, Eckard, 2011



Due to the high numerical effort and memory demand of DNS, it is, in general, not possible to simulate the full structure on the micro-/meso-scale with the computational power available nowadays

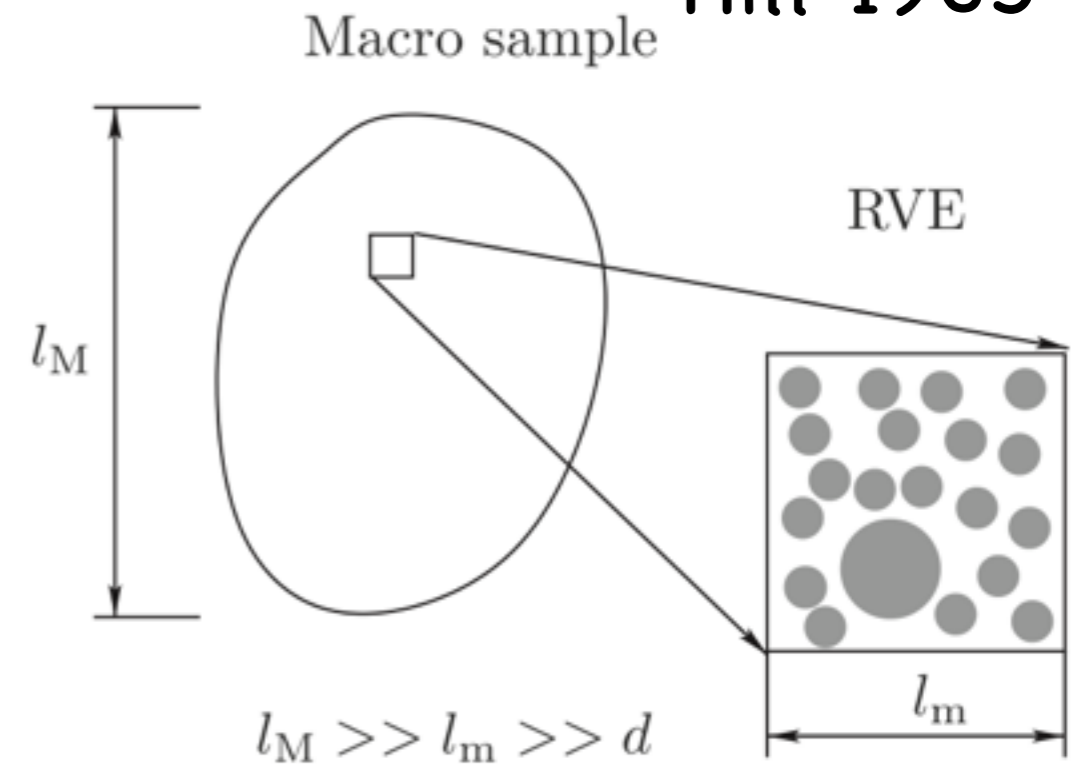
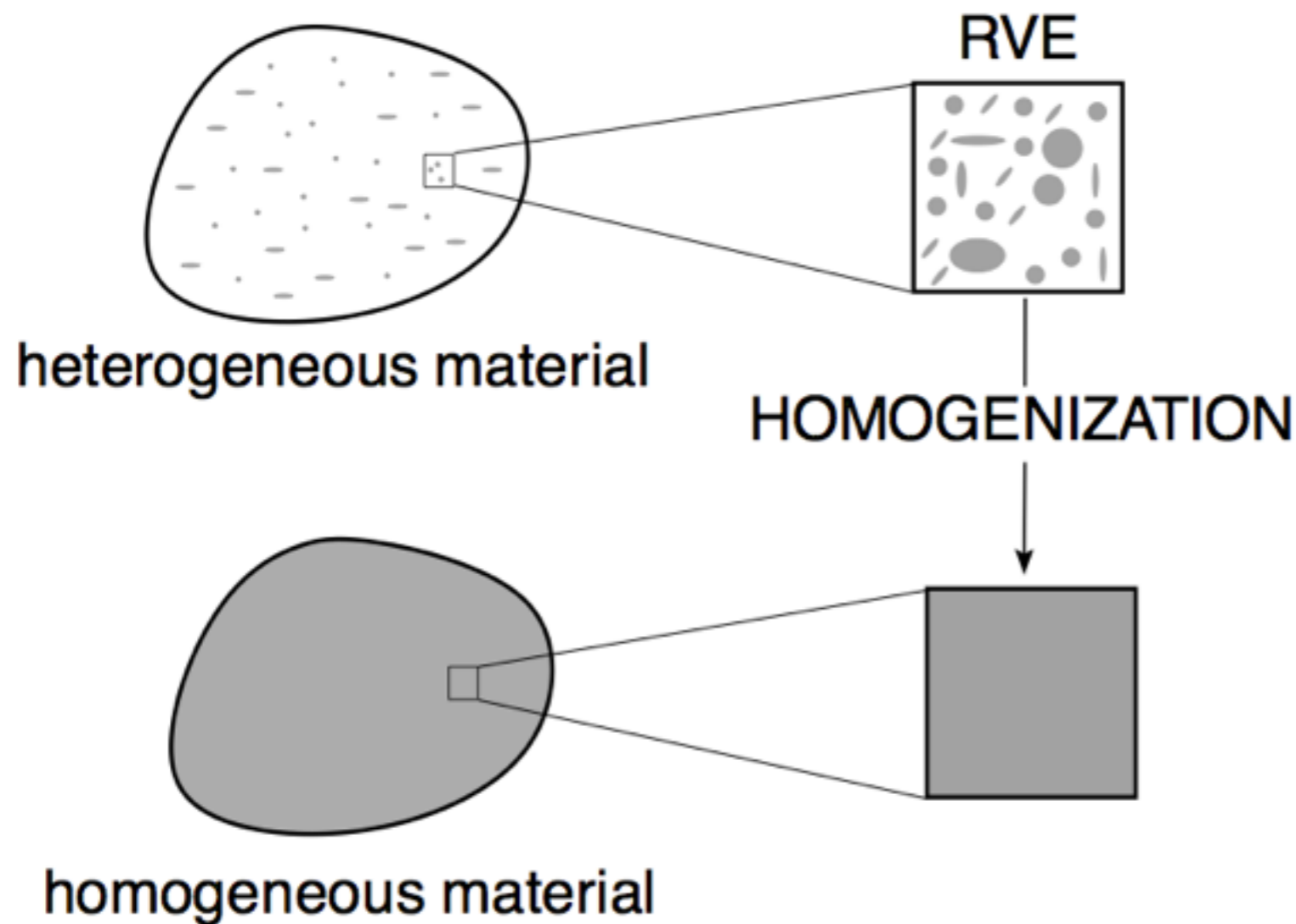


1 100 000 degrees of freedom

Homogenization

Homogenization = replace a heterogeneous material with an equivalent homogeneous material.

Voigt, 1910
Reuss 1929
Hill 1965

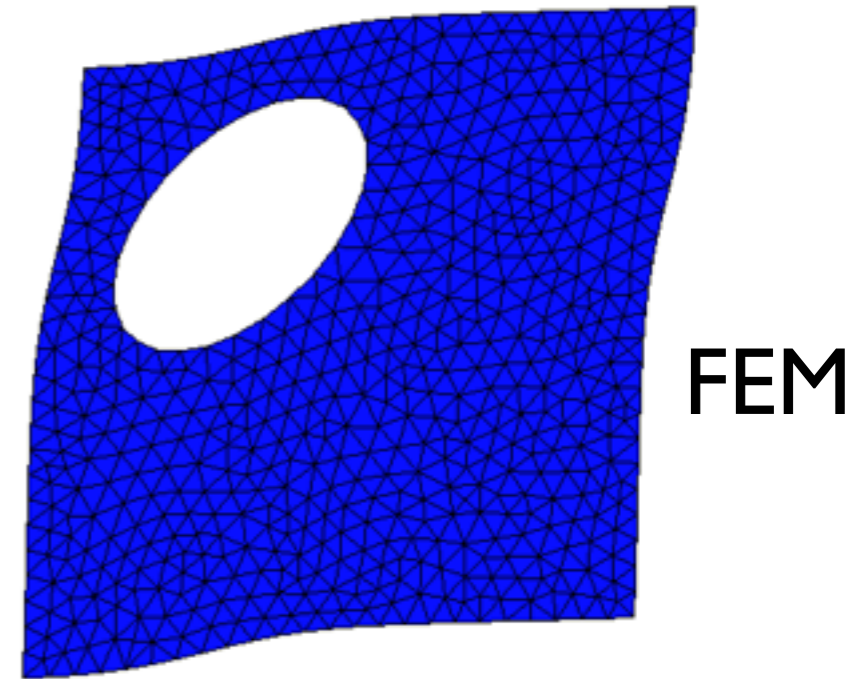


separation of scales

Homogenization

$$\langle \boldsymbol{\sigma} \rangle = \mathbf{C} : \langle \boldsymbol{\epsilon} \rangle$$

1



bottom-up approach

RVE



2

$$\langle \boldsymbol{\sigma} \rangle = \frac{1}{|\Omega_m|} \int_{\Omega_m} \boldsymbol{\sigma}_m d\Omega$$

$$\langle \boldsymbol{\epsilon} \rangle = \frac{1}{|\Omega_m|} \int_{\Omega_m} \boldsymbol{\epsilon}_m d\Omega$$

6 independent loads are needed to determine 36 constants

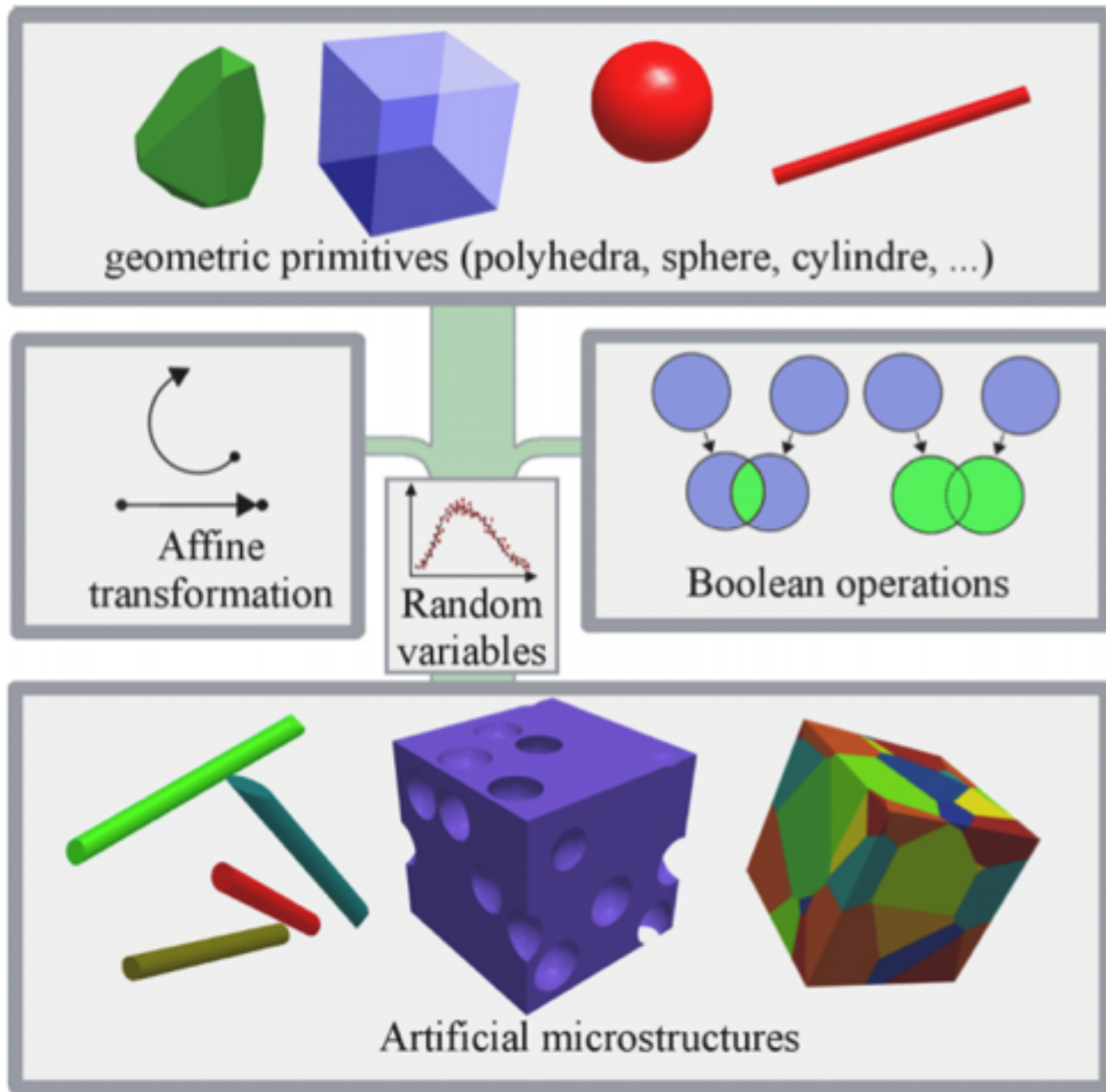
\mathbf{C} effective properties

$$\begin{Bmatrix} \langle \sigma_{11} \rangle_{\Omega} \\ \langle \sigma_{22} \rangle_{\Omega} \\ \langle \sigma_{33} \rangle_{\Omega} \\ \langle \sigma_{12} \rangle_{\Omega} \\ \langle \sigma_{23} \rangle_{\Omega} \\ \langle \sigma_{13} \rangle_{\Omega} \end{Bmatrix} = \begin{bmatrix} E_{1111}^* & E_{1122}^* & E_{1133}^* & E_{1112}^* & E_{1123}^* & E_{1113}^* \\ E_{2211}^* & E_{2222}^* & E_{2233}^* & E_{2212}^* & E_{2223}^* & E_{2213}^* \\ E_{3311}^* & E_{3322}^* & E_{3333}^* & E_{3312}^* & E_{3323}^* & E_{3313}^* \\ E_{1211}^* & E_{1222}^* & E_{1233}^* & E_{1212}^* & E_{1223}^* & E_{1213}^* \\ E_{2311}^* & E_{2322}^* & E_{2333}^* & E_{2312}^* & E_{2323}^* & E_{2313}^* \\ E_{1311}^* & E_{1322}^* & E_{1333}^* & E_{1312}^* & E_{1323}^* & E_{1313}^* \end{bmatrix} \begin{Bmatrix} \langle \epsilon_{11} \rangle_{\Omega} \\ \langle \epsilon_{22} \rangle_{\Omega} \\ \langle \epsilon_{33} \rangle_{\Omega} \\ 2\langle \epsilon_{12} \rangle_{\Omega} \\ 2\langle \epsilon_{23} \rangle_{\Omega} \\ 2\langle \epsilon_{13} \rangle_{\Omega} \end{Bmatrix}$$

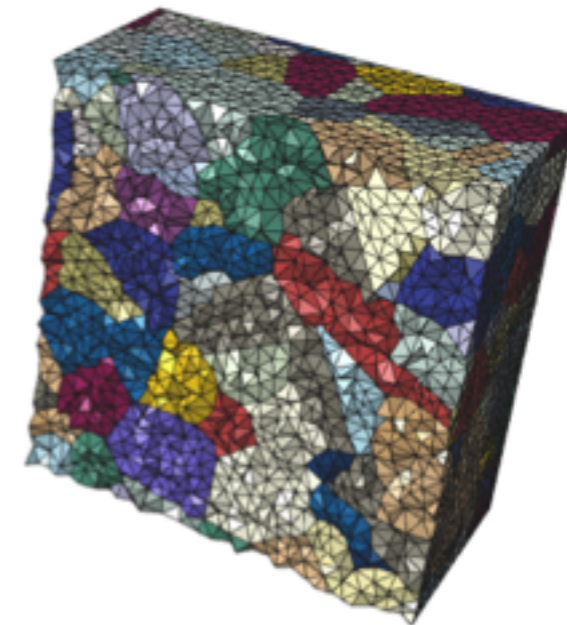
o constitutive model
simple plasticity

Artificial microstructures

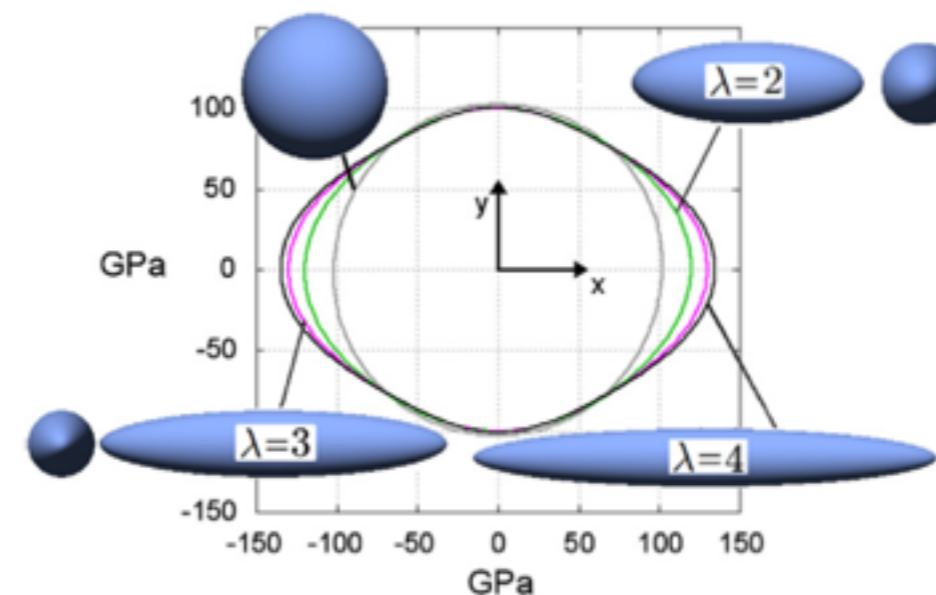
Real microstructures: hard to obtain and not meshable



Statistically equivalent to real microstructures
Easy to discretized into finite elements



Build tailor made materials

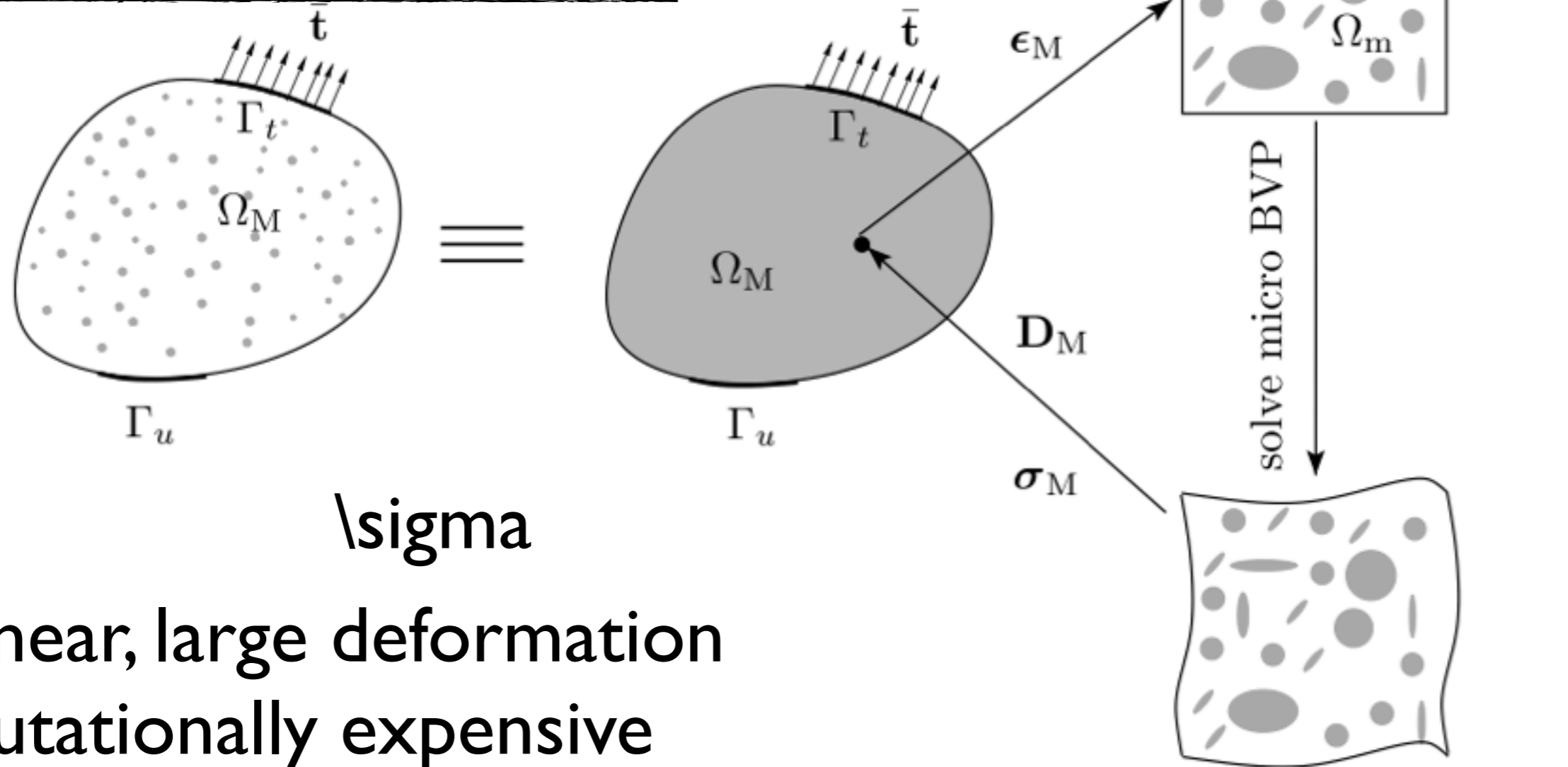


F. Fritzen 2010

Computational Homogenization

FE² method

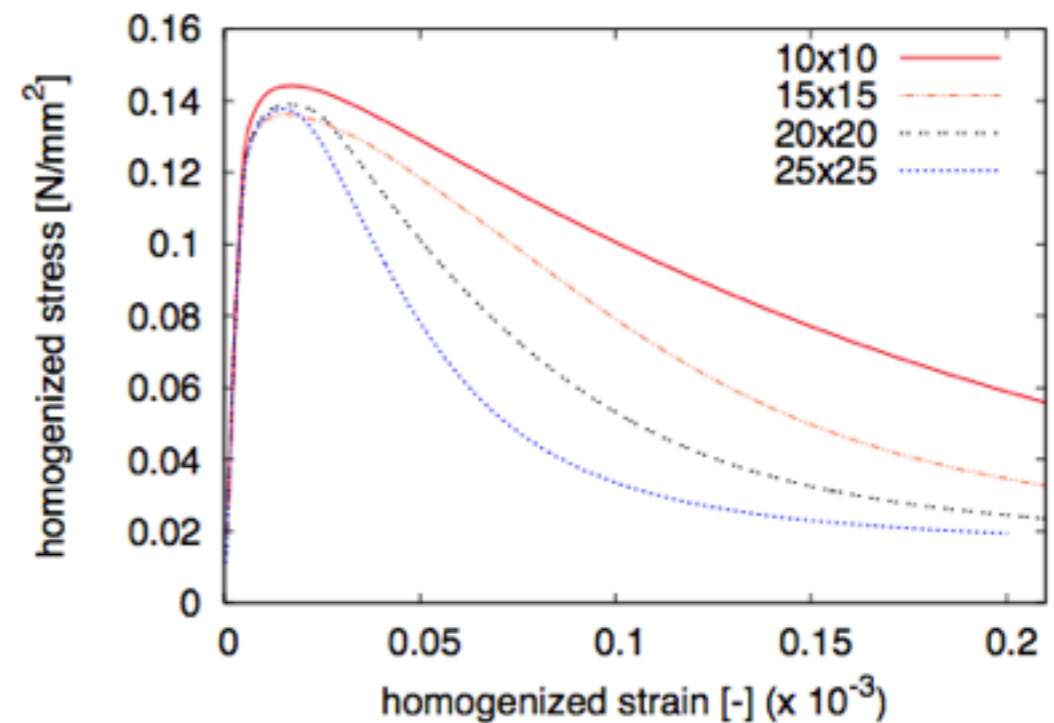
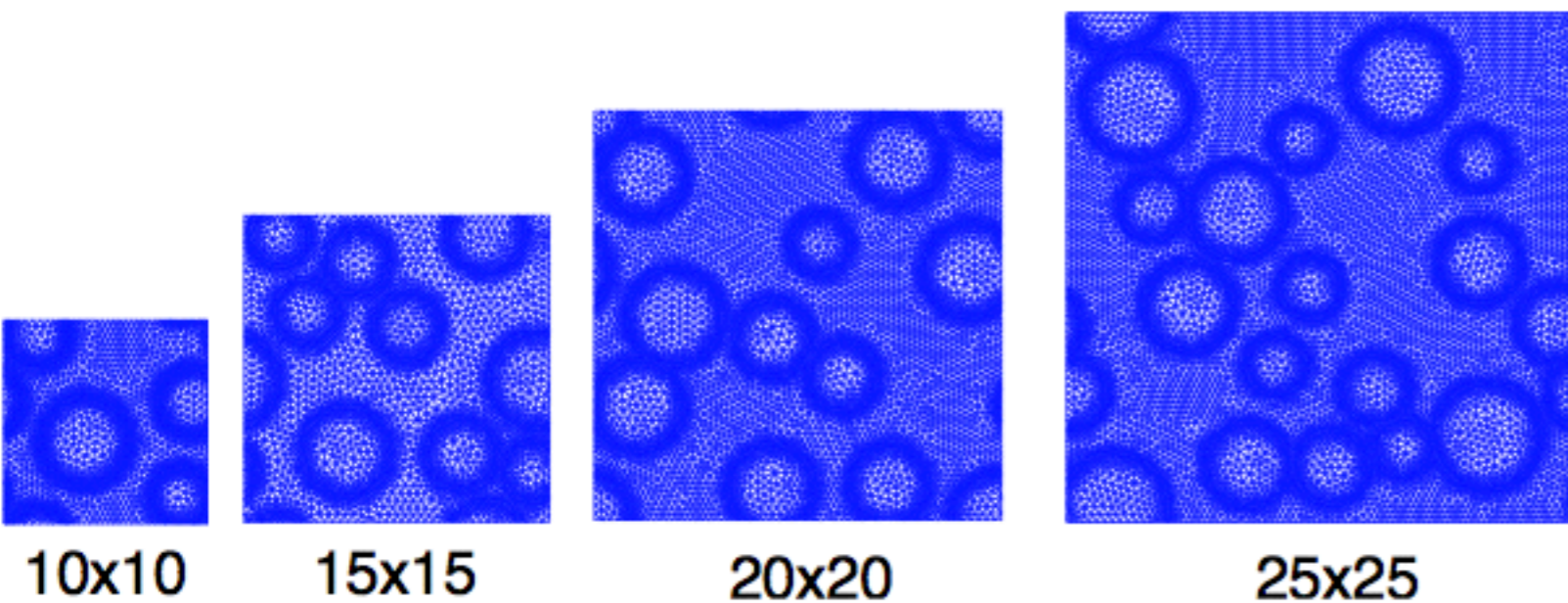
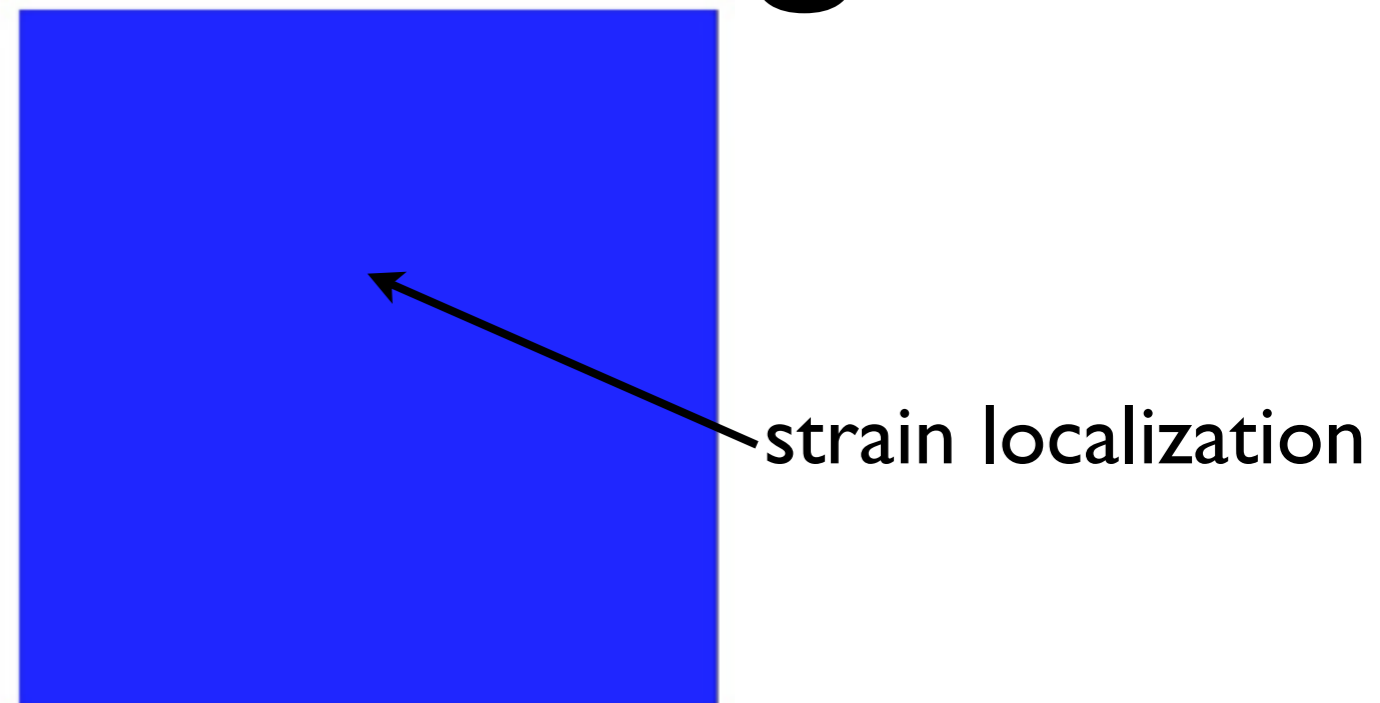
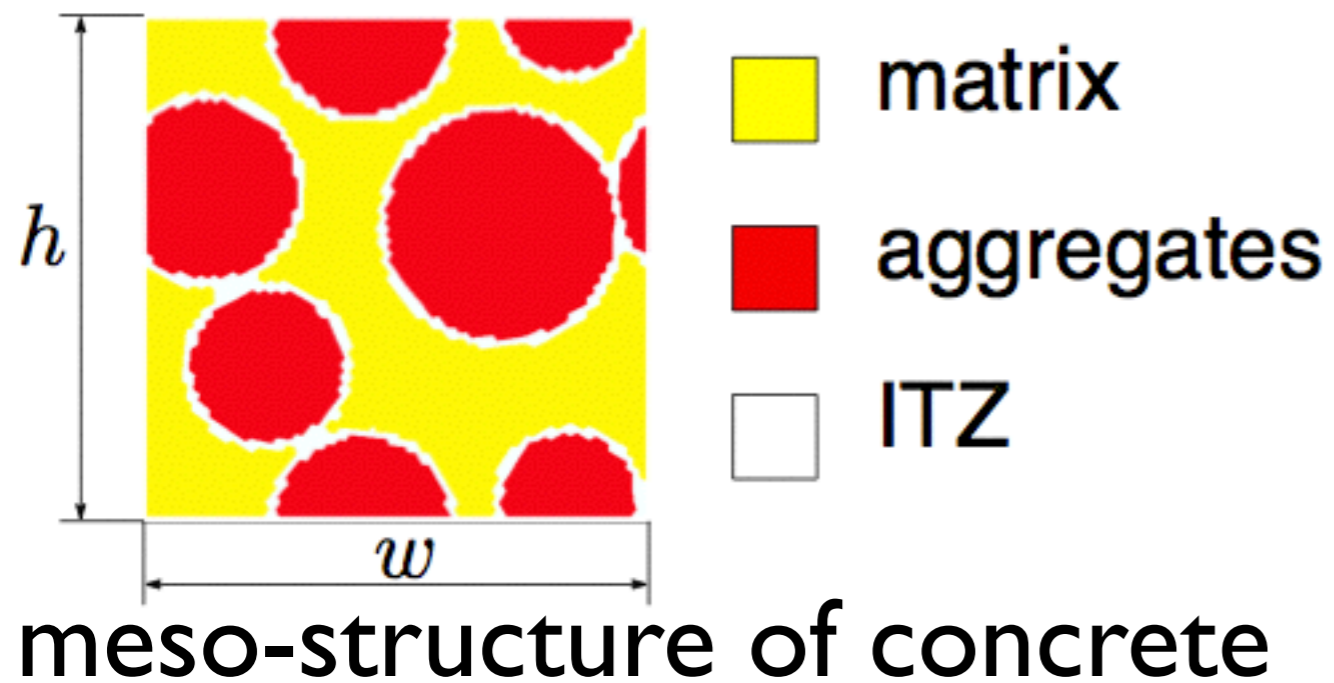
[Renard, 1987, Smit 1998,
F. Feyel, 2000]



- + nonlinear, large deformation
- computationally expensive
- 2D problems at laboratory scale
- not always robust!!!

Micro problems are solved in parallel

Troubles with softening RVEs



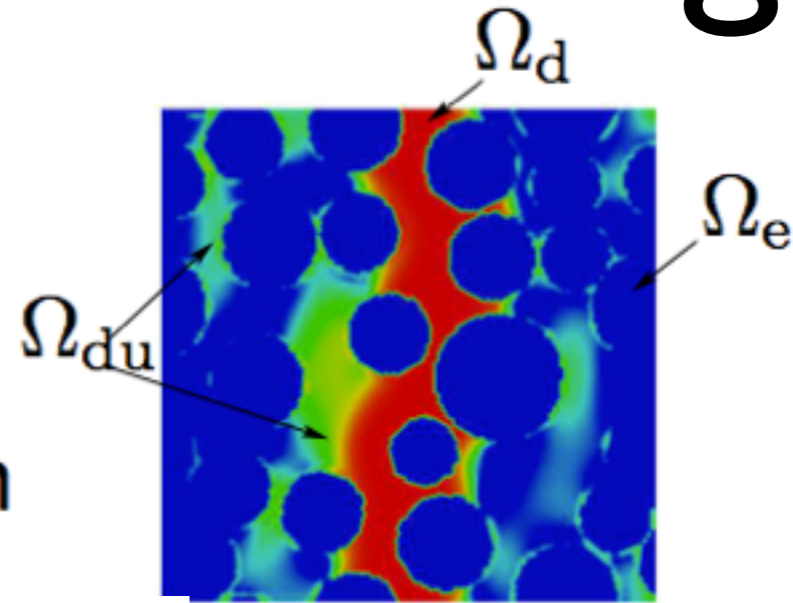
- ➔
- RVE does not exist for softening materials
 - CH cannot be applied for softening materials

Failure zone averaging

Ω_e : elastic domain

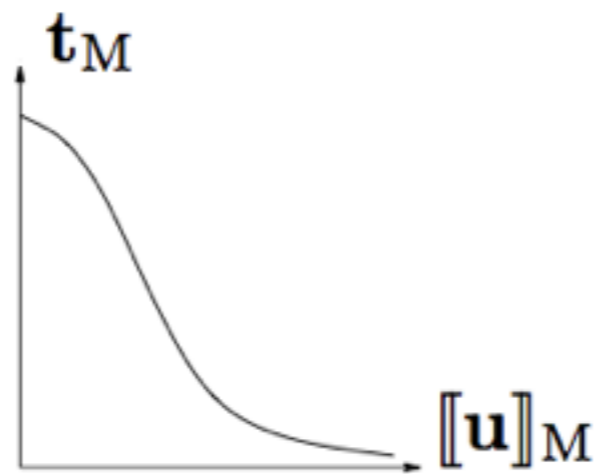
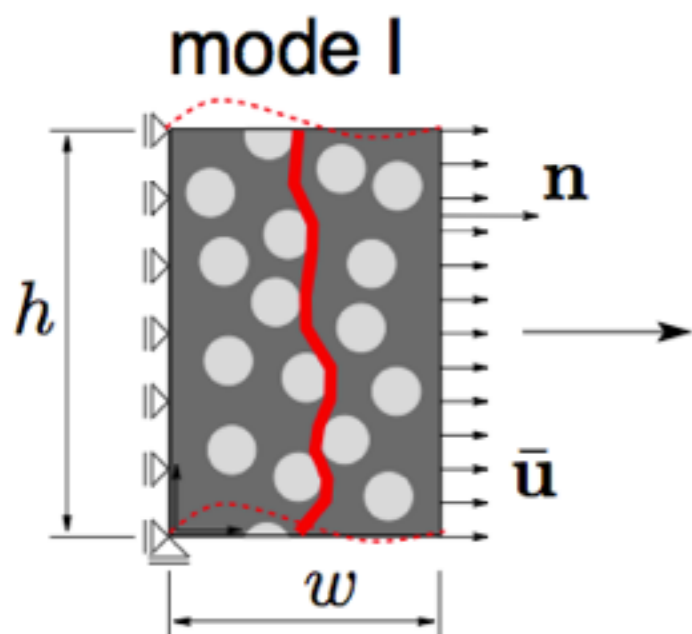
Ω_d : active damaged domain

Ω_{du} : inactive damaged domain



$$\Omega_d = \{ \mathbf{x} \in \Omega_m \mid \omega(\mathbf{x}) > 0, f(\mathbf{x}) = 0 \}$$

$$\langle \boldsymbol{\sigma} \rangle_{\text{dam}} = \frac{1}{|\Omega_d|} \int_{\Omega_d} \boldsymbol{\sigma}_m d\Omega_d, \quad \langle \boldsymbol{\epsilon} \rangle_{\text{dam}} = \frac{1}{|\Omega_d|} \int_{\Omega_d} \boldsymbol{\epsilon}_m d\Omega_d$$

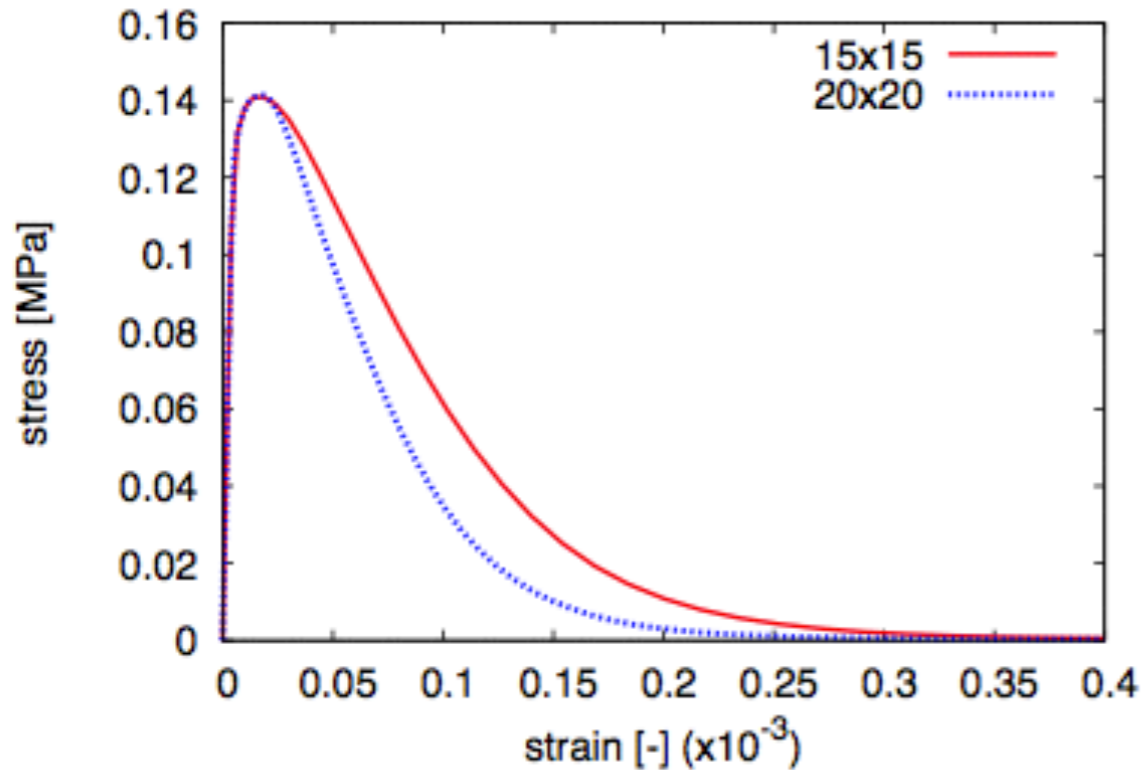


$$\mathbf{t}_M = \frac{1}{h} \boldsymbol{\sigma}_M \cdot \mathbf{n}$$

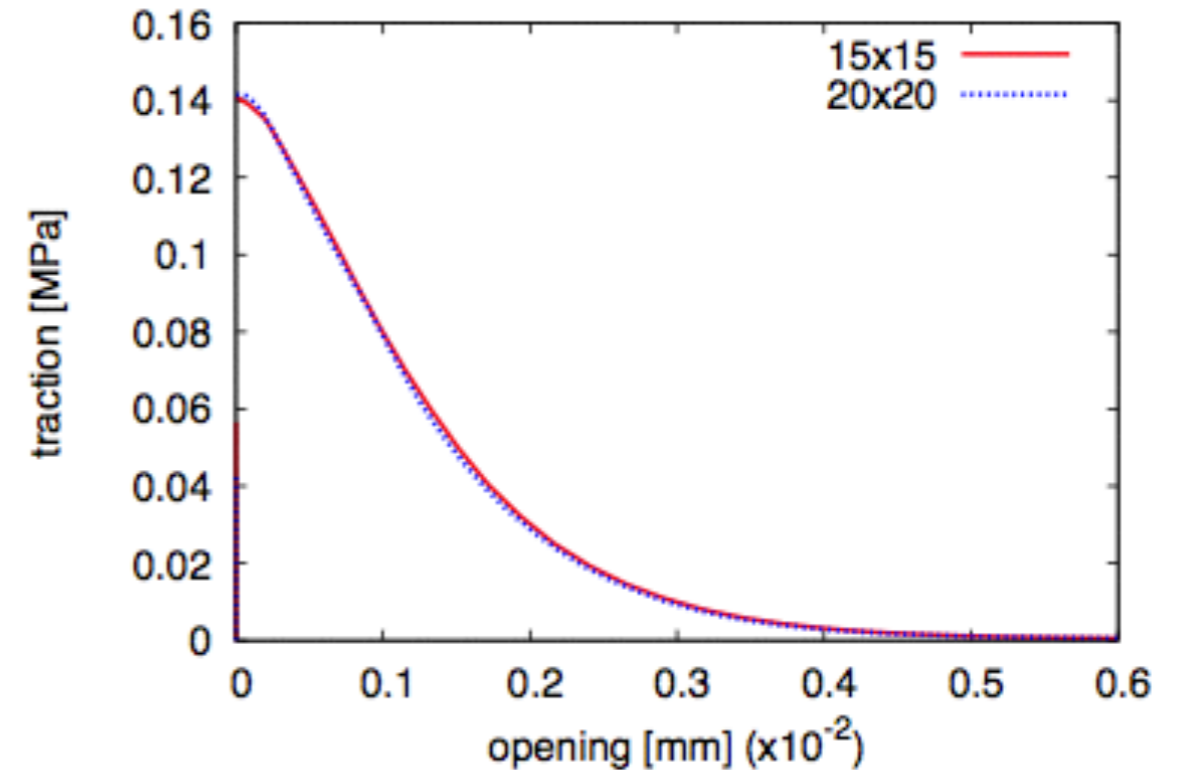
$$\mathbf{u}_{\text{dam}} = \langle \boldsymbol{\epsilon} \rangle_{\text{dam}} \cdot (l\mathbf{n}), \quad l = |\Omega_d| / h$$

$$[[\mathbf{u}]]_M = \mathbf{u}_{\text{dam}} - \dot{\mathbf{u}}_{\text{dam}}$$

standard averaging



new averaging



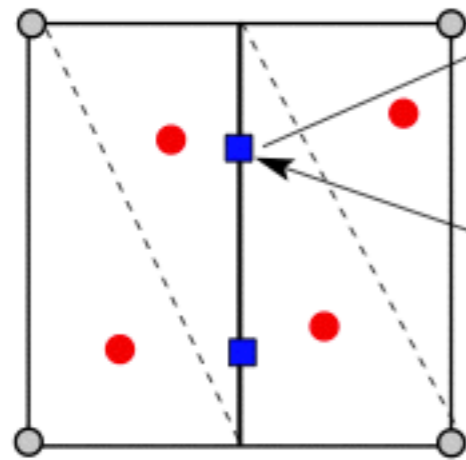
➔ RVE does exist for softening materials by using the failure zone averaging technique

Discontinuous CH model

MACRO

MICRO

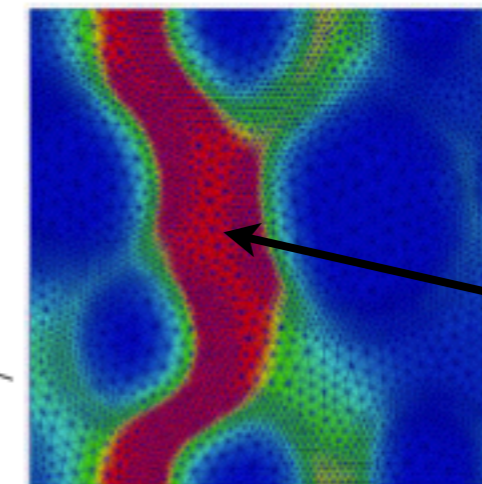
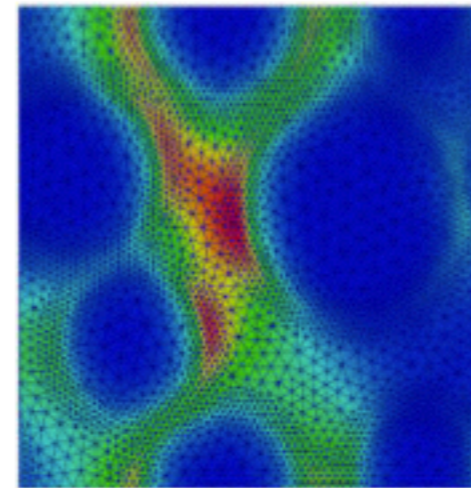
- $\dot{\boldsymbol{\sigma}}_M = \mathbf{D}_0 \dot{\boldsymbol{\epsilon}}$
- $\dot{\mathbf{t}}_M = \mathbf{T}_M [[\dot{\mathbf{u}}]]_M$



discrete crack

$[[\mathbf{u}]]_M$

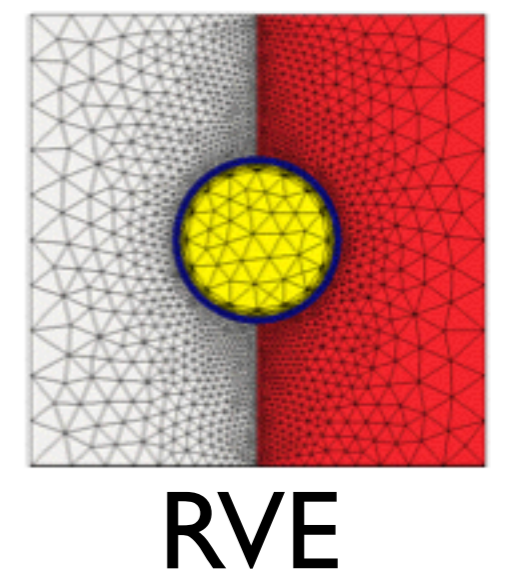
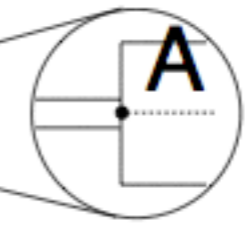
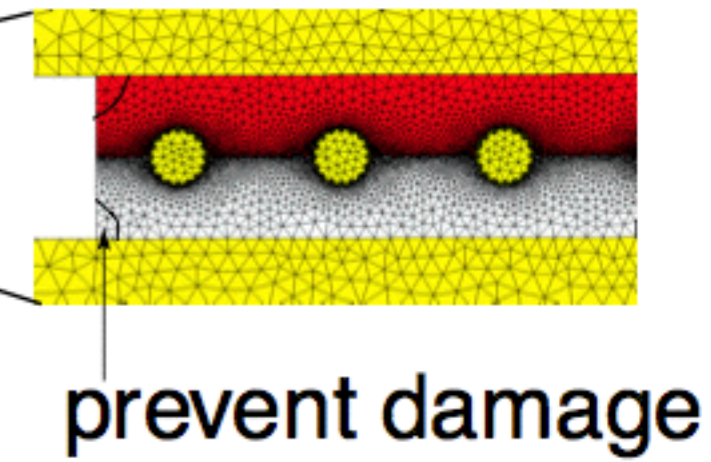
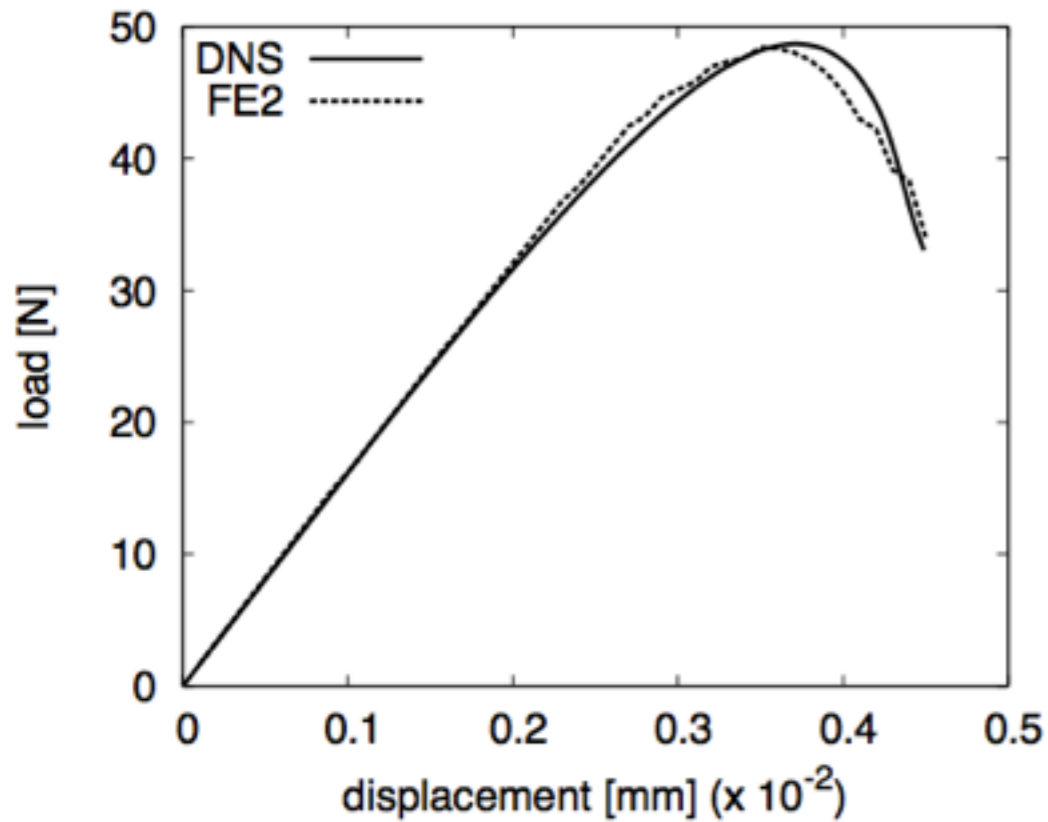
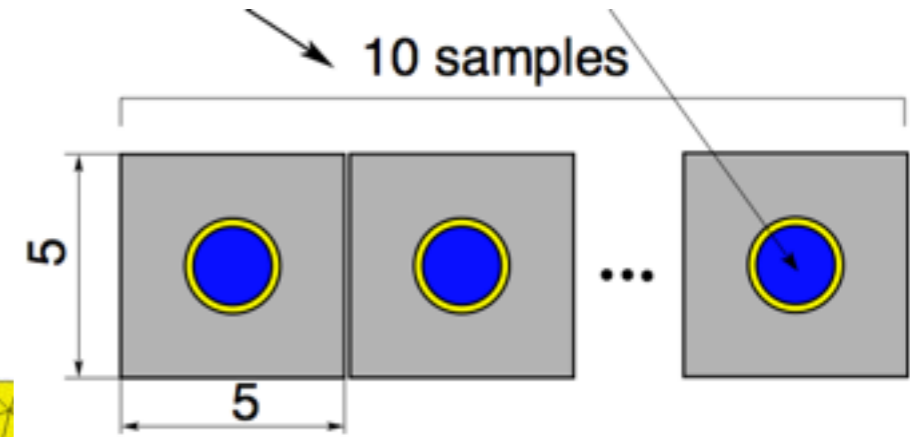
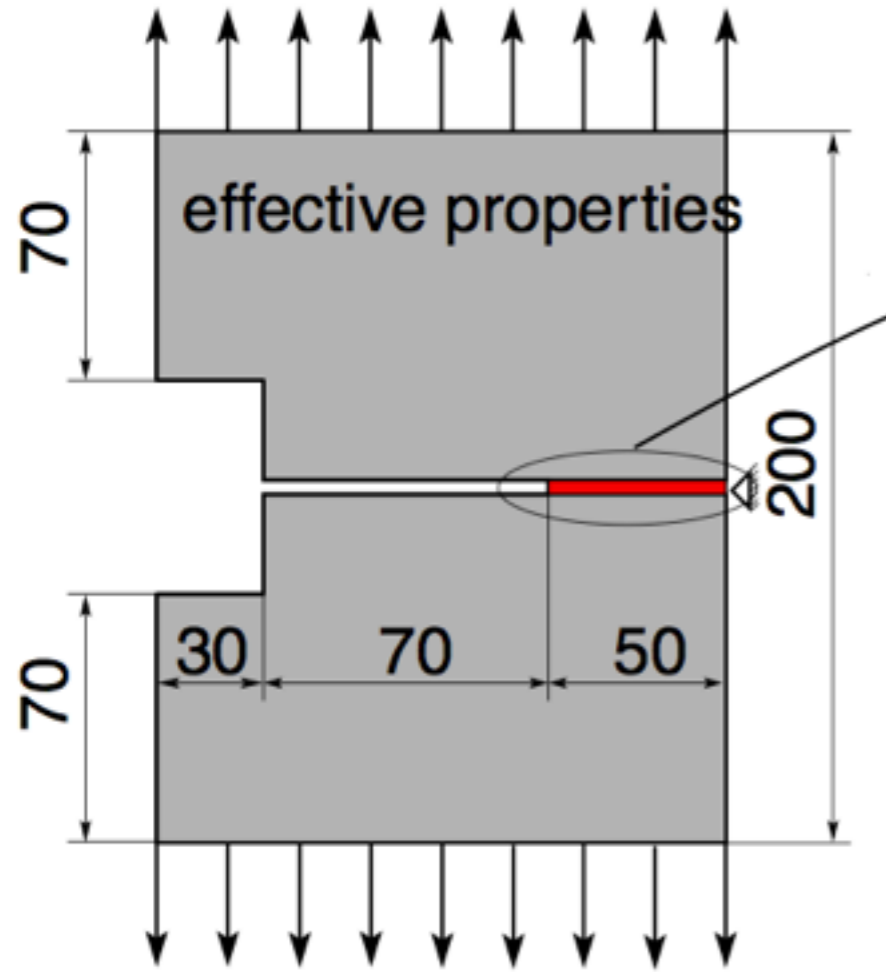
$\mathbf{t}_M, \mathbf{T}_M$



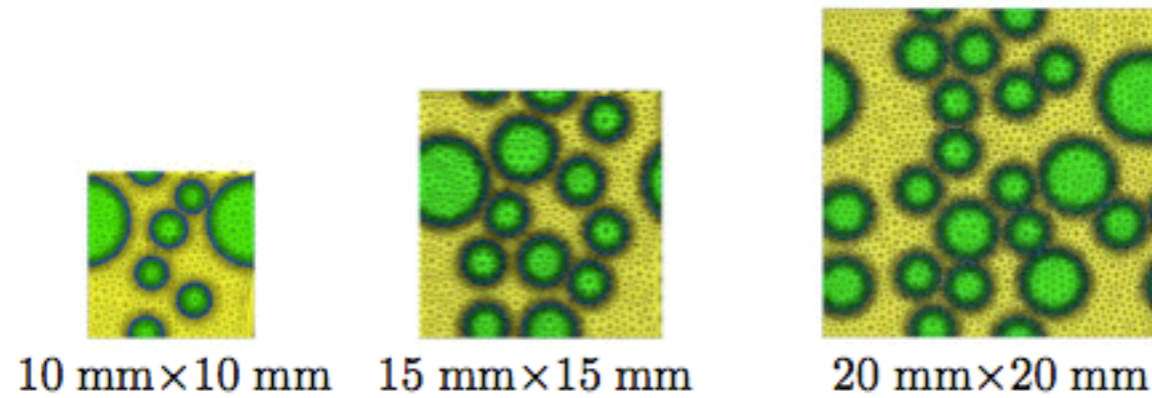
localization
band

Nguyen et al, 2011

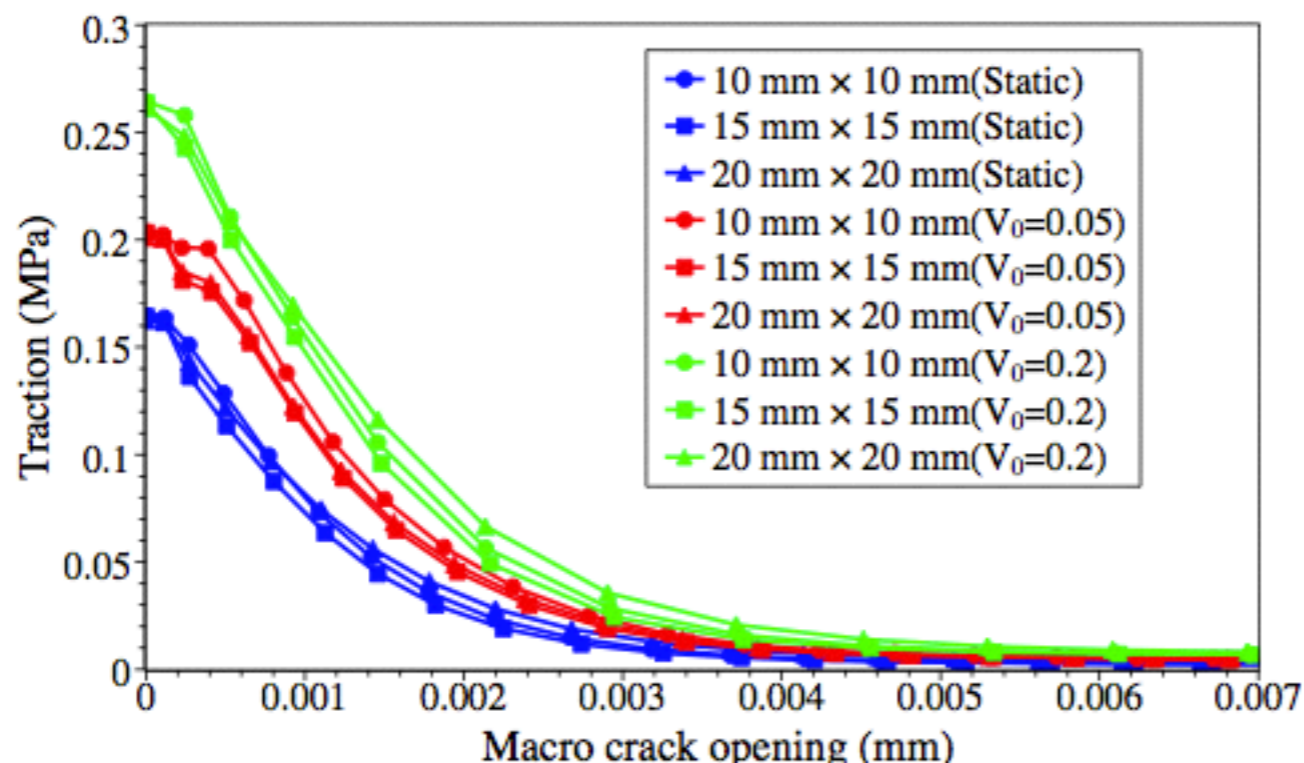
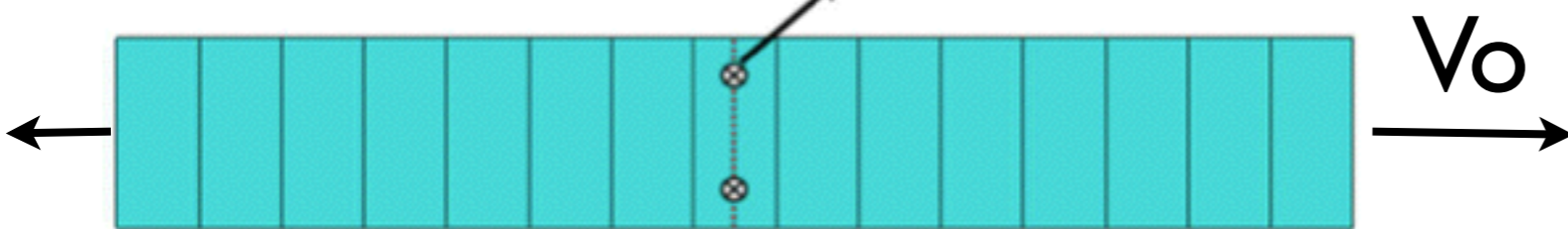
Example



Dynamic discontinuous CH model



- macro: implicit dynamics
- micro: quasi-static



A. Karamnejad,
Nguyen, Sluys, 2012

More information

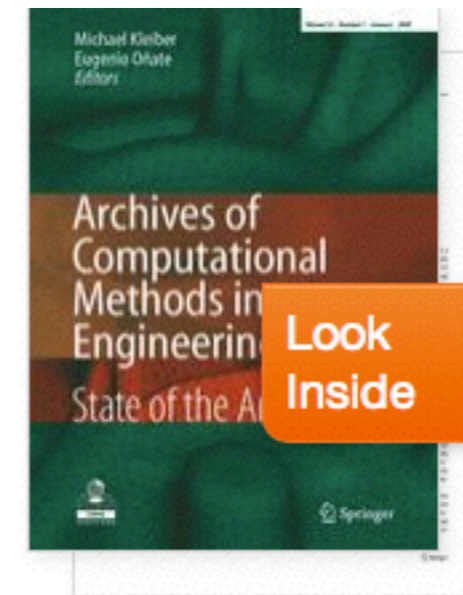
[Archives of Computational Methods in Engineering](#)

March 2009, Volume 16, Issue 1, pp 31-75

Multiscale Methods for Composites: A Review

[P. Kanouté](#), [D. P. Boso](#), [J. L. Chaboche](#), [B. A. Schrefler](#)

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MULTISCALE CONTINUOUS AND DISCONTINUOUS MODELING OF HETEROGENEOUS MATERIALS: A REVIEW ON RECENT DEVELOPMENTS

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Delft University of Technology, Faculty of Civil Engineering and Geosciences, P. O. Box 5048, 2600 GA Delft, The Netherlands

MARTIJN STROEVEN

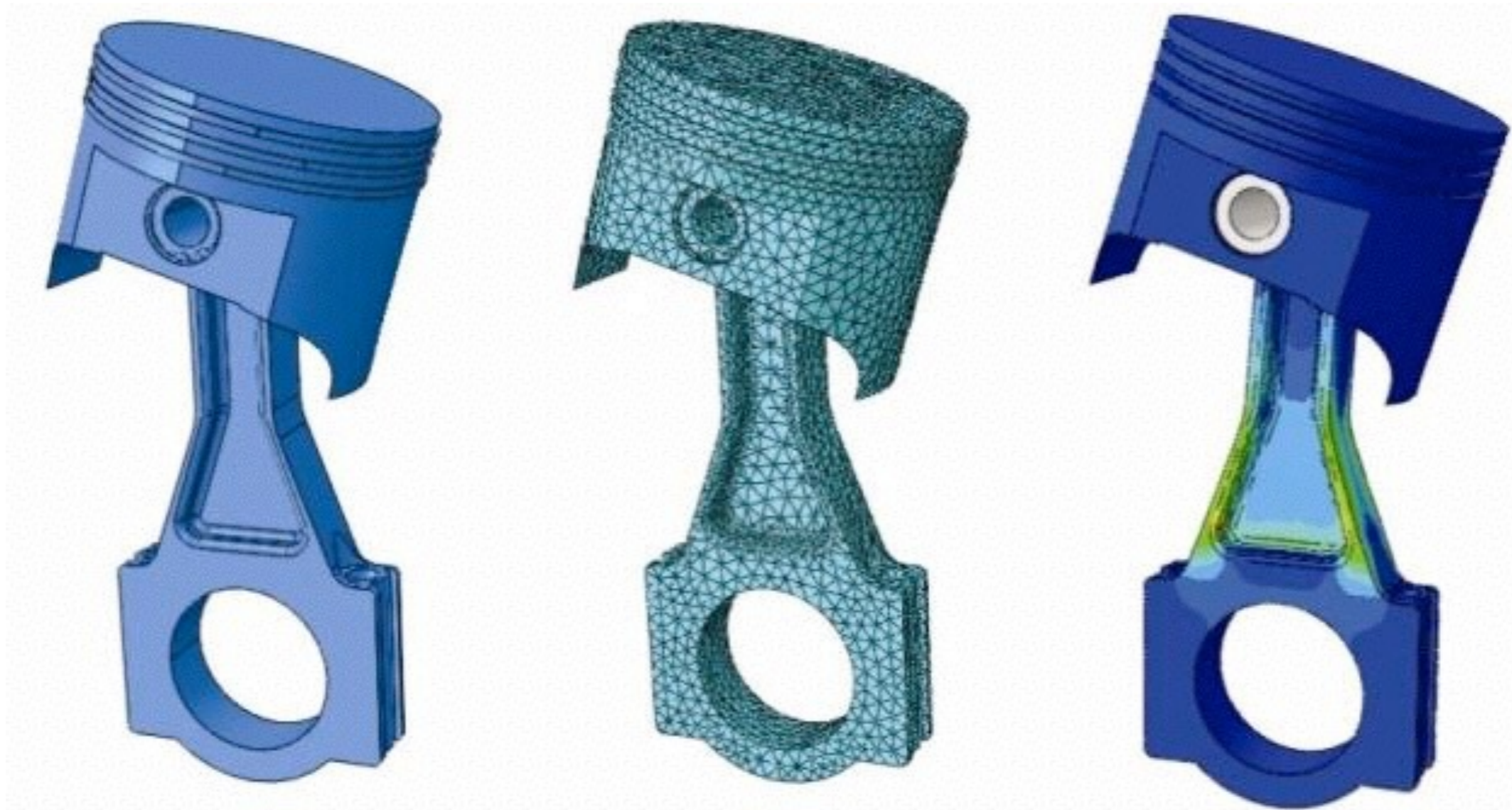
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LAMBERTUS JOHANNES SLUYS

Delft University of Technology, Faculty of Civil Engineering and Geosciences, P. O. Box 5048, 2600 GA Delft, The Netherlands

Image-based modeling

Traditional FE analysis



Geometry
(CAD)



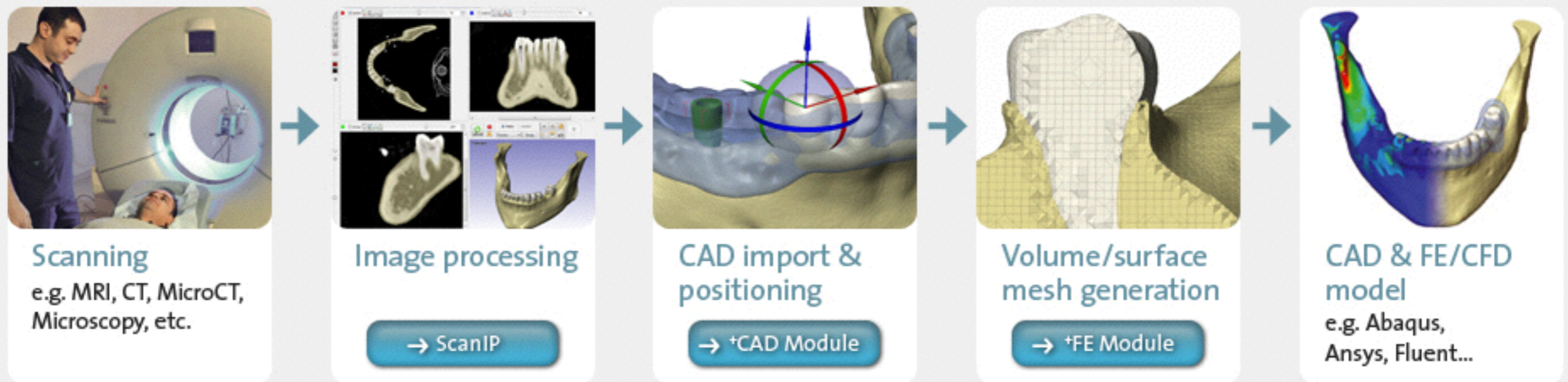
Mesh



FE solver

There are many cases in which such CAD geometries are not available. However, image data are so ready: medicine, material sciences...

(I) Industry



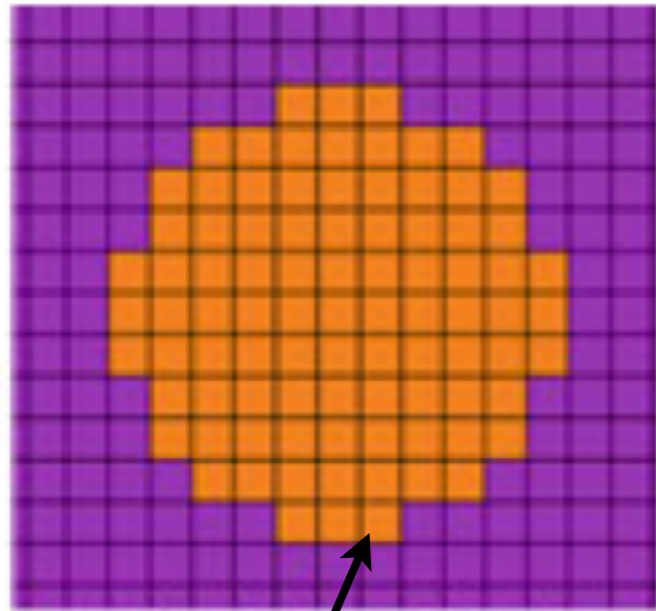
Simpleware

See also the FREE program OOF2, NIST, USA

(2) universities voxel based method

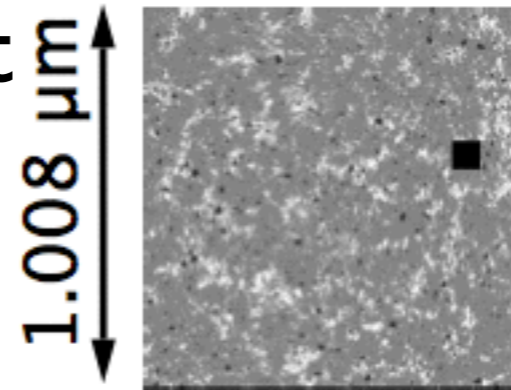
microstructure of cement paste

each voxel = one finite element

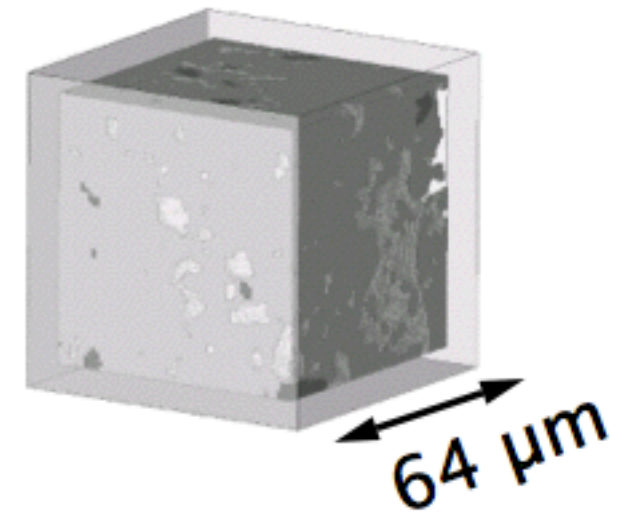


Pixel-Based Mesh

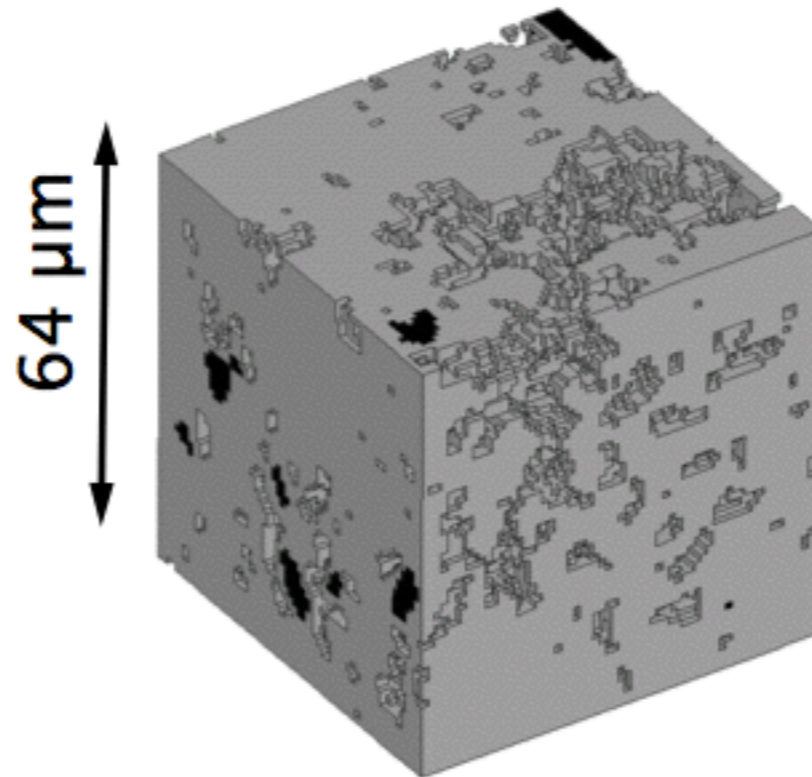
zig-zag boundary



1.008 μm

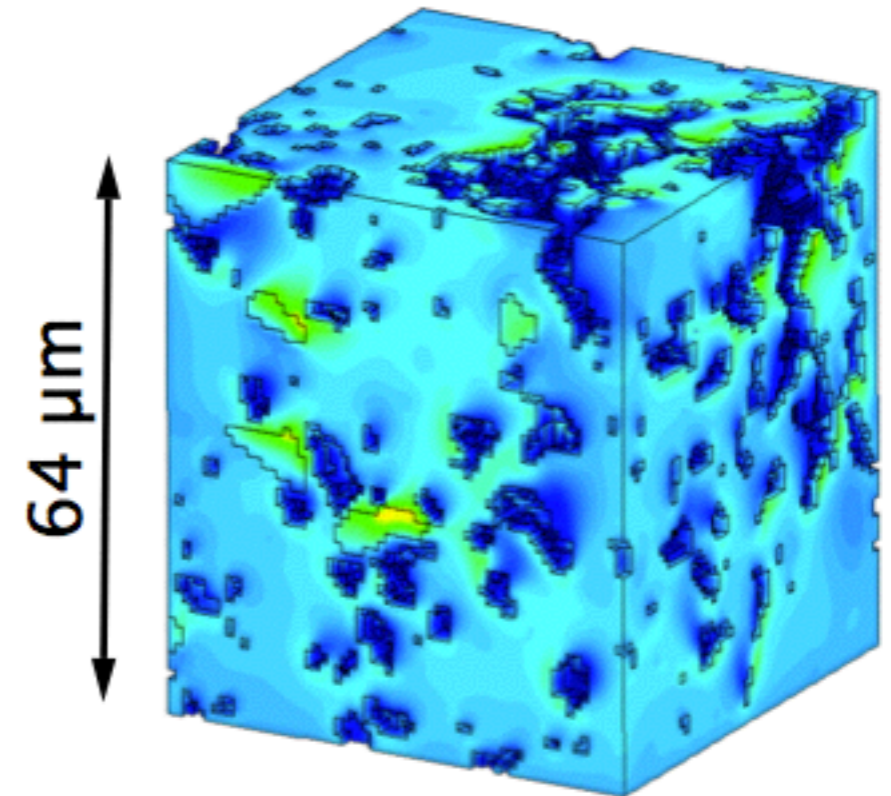


64 μm



64 μm

64^3 hexahedron elements



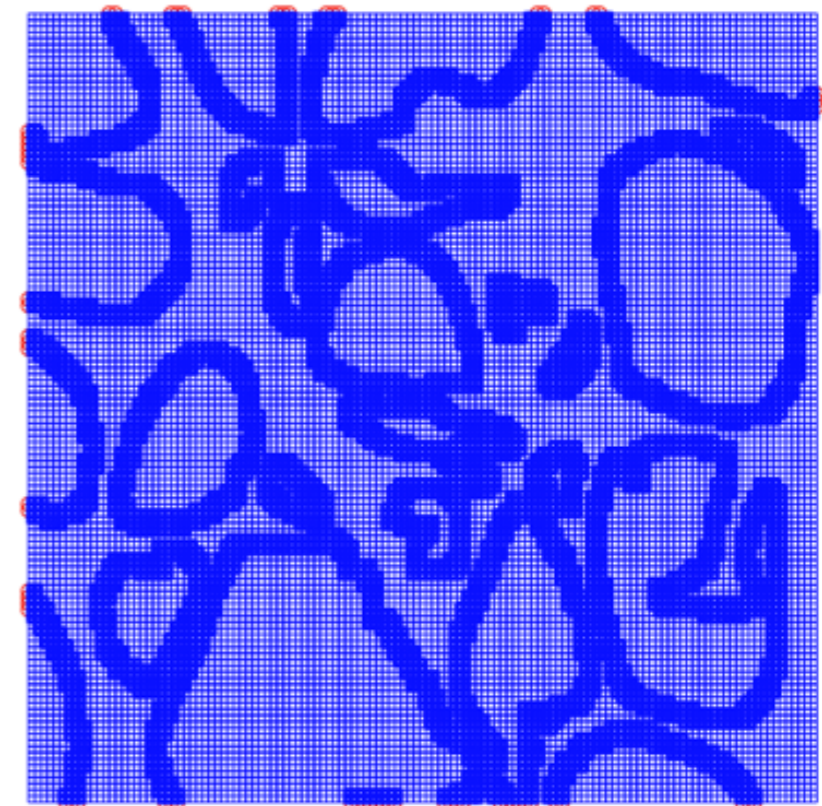
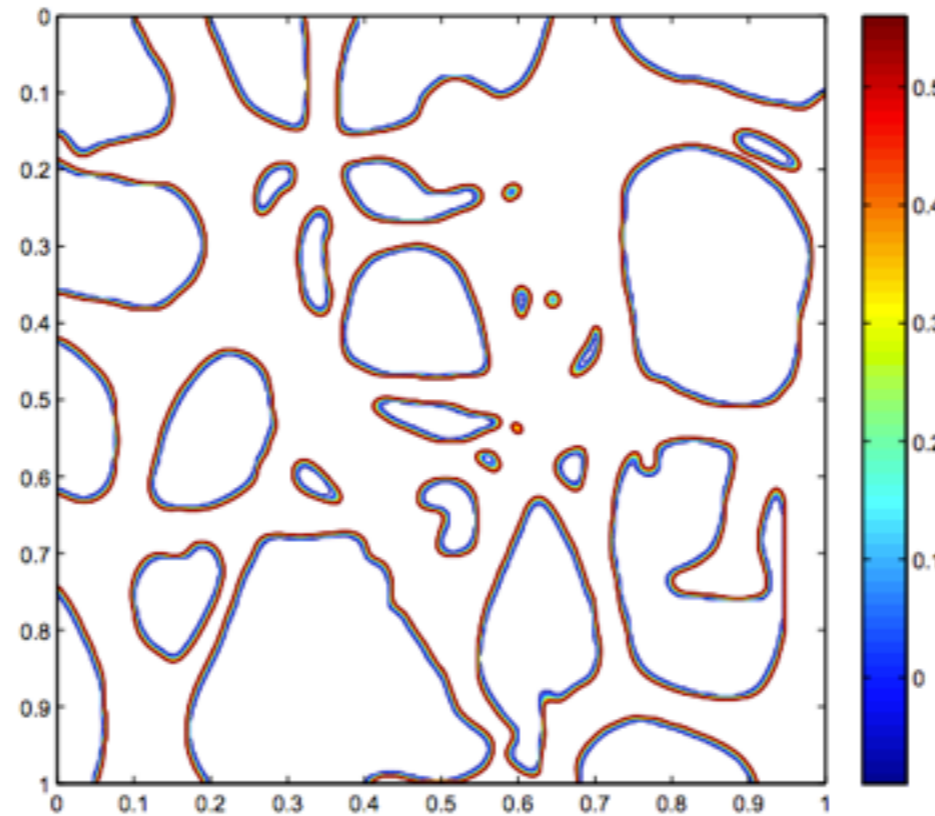
64 μm

820 000 DOFs

- incorrect volume fraction
- images with high resolution are required
- too large problem size!!!

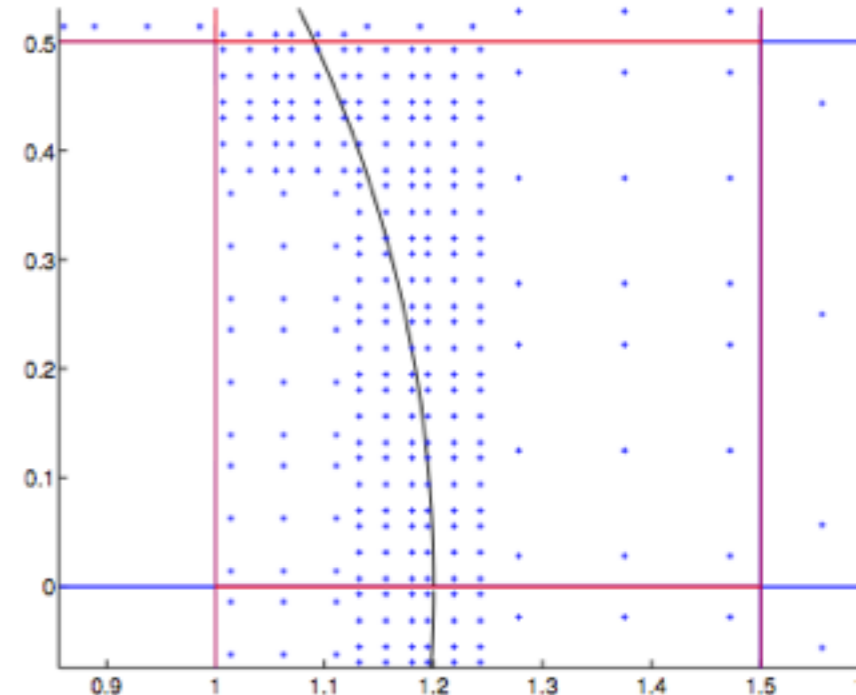
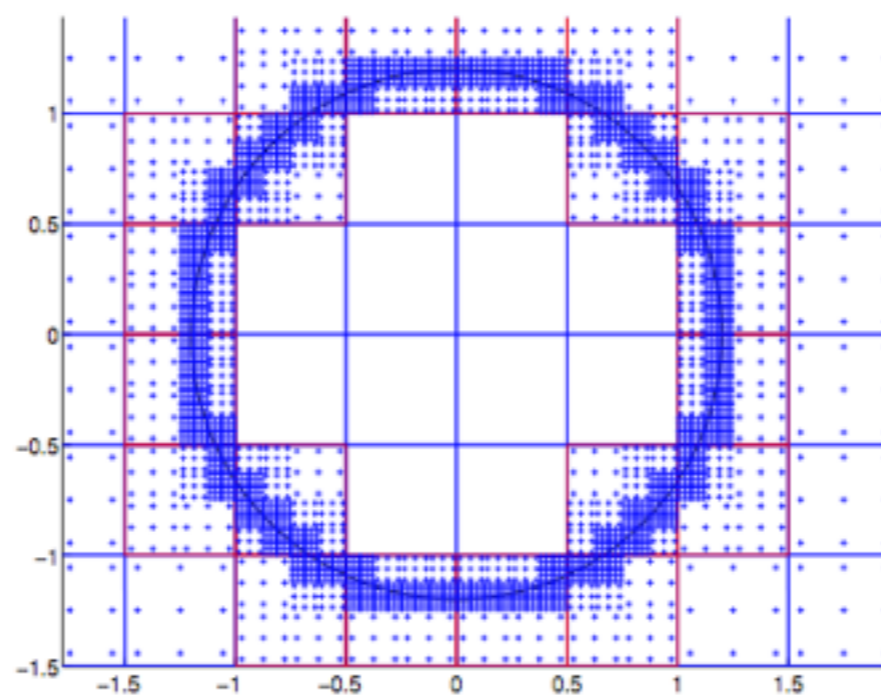
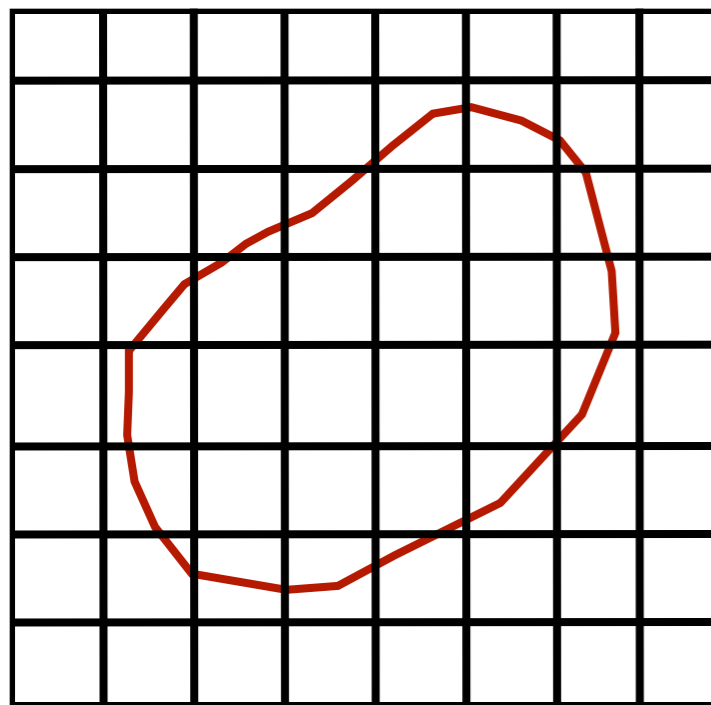
P. Wriggers

Level set/XFEM



Finite Cell Method (FCM)

(Fictitious Domain Methods)



Tools

Matlab is not enough. Consider Fortran, C++, Python.

Move to **Ubuntu Linux** to make your programming life much easier.

- **Preprocessing:** GMSH, GID, ANSYS, **ABAQUS**

- **Solvers:**

trilinos.sandia.gov

- FEM: FEAP, OOFEM, libMesh, KRATOS, Code Aster,
TRILINOS *PFRMIX*, OpenSees (earthquake, structures)

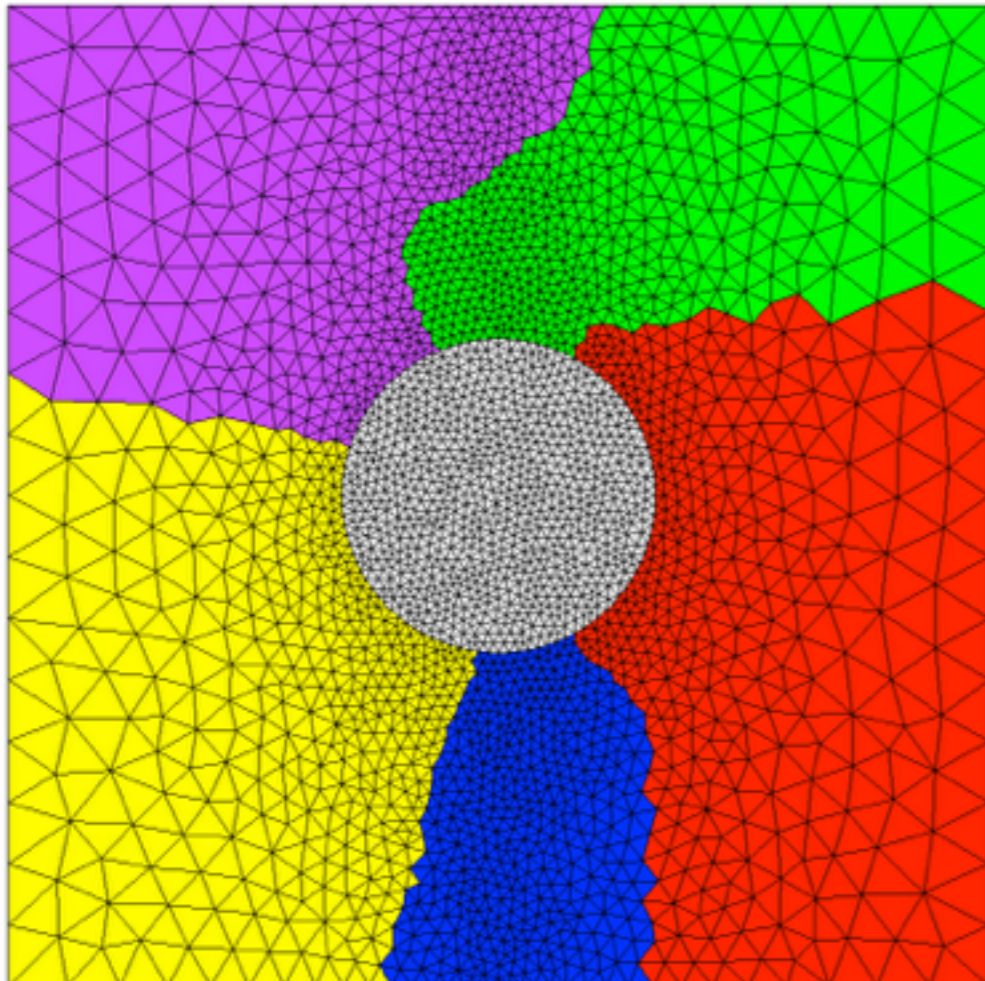
KRATOS, YADE...

, KRATOS...

OpenMPI
ParMETIS

MSH, PARAVIEW, MATLAB, *TECPLOT*

domain decomposition



Prof. S. Bordas



Dr. O. Lloberas Valls

Prof. L.J. Sluys



A. Karamnejad



Dr. E. Lingen



Dr. M. Stroeve

Habanera develops jem/jive C++ library

**Thank you for
your attention**